

SHUTTLE PAYLOAD INTERFACE VERIFICATION
EQUIPMENT STUDY
VOLUME II TECHNICAL DOCUMENT PART 2 APPENDICES

**APRIL 1976** 

NASA CONTRACT: NAS9-14000 CCA 140 REV. 1

PRÉPARED BY: SHUTTLE PAYLOAD INTERFACE PROJECTS GROUP APPROVED BY:

\*\*JEEncell\*\*

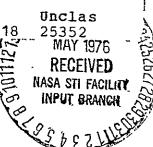
H. E. EMIGH

DIRECTOR

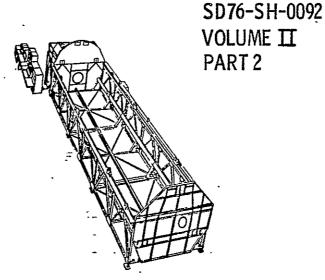
STS/PAYLOAD INTEGRATION

(NASA-CF-147667) SHUTTLE PAYLOAD INTERFACE
VERIFICATION EQUIPMENT STUDY. VOLUME 2:
TECHNICAL DOCUMENT. PART 2: APPENDICES
(Rockwell International Corp., Downey,
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N76-22263



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YOLUME TECHNICAL DOCUMENT PART 2 APPENDICES

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#### **FOREWORD**

This document is a contractual requirement of NAS9-14000, CCA 140 Revision 1 and is provided in response to the contract. The study was conducted by the Space Division of Rockwell International for the Johnson Space Center of the National Aeronautics and Space Administration. It is published in four volumes:

Vol.	I.	Executive Summary
··Vol.	II .	Technical Document - Part 1 Technical Appendices - Part 2
Wol.	III	Specification Data
Vol.	in	Project Plans

## TECHNICAL REPORT INDEX/ABSTRACT

	DOCUMENT	SECURITY CLASSIFICATION
ACCESSION NUMBER		ssified
TITLE OF DOCUMENT Shuttle Payload Interfa	ace Verification Equipme	ent(IVE) Study
CODE ORIGINATING AGENCY A	NO OTHER SOURCES	DOCUMENT NUMBER
Rockwell Intern Space Division	-	SD76_SH-0092
April 1, 1976	NAS9-14000 CCA 140 F	Rev. 1
•		,
DESCRIPTIVE TERMS		
· Shuttle	Preliminary Design	Data Management
Payloads	· Operators Console	Computer
Spacelab	Mission Station	Heat Exchanger
P/L Interface	On-Orbit Station	Development Plans
P/L Interface Verif.	Payload Station	Schedules
Avionics	Electrical Power	•
Payload Integration	Communications	•

STRACT

Orbiter within the 160 hour turnaround requirement for the Shuttle system. In order to accomplish this integration process some off-line integration capability is required. This report is a preliminary design analysis of a "stand alone" (no facility GSE support required) payload integration device (IVE) capable of verifying payload compatibility in form, fit and function with the Shuttle Orbiter prior to on-line payload/Orbiter operations. The IVE is a high fidelity replica of the Orbiter payload accommodations capable of supporting payload functional checkout and mission simulation. A top level payload integration analysis developed detailed functional flow block diagrams of the payload integration process for the broad spectrum of P/L's and identified degree of Orbiter data required by the payload user and potential applications of the IVE.

This work was performed for Johnson Space Center of the National Aeronautics and Space Administration under contract NAS9-14000 CCA 140 Rev. 1.

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·C·	Payload Integration Baseline Functional Flow Block Diagrams and Options	C-1

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### APPENDIX A

INTERFACE VERIFICATION EQUIPMENT HARDWARE UTILIZATION LIST (HUL)



## IVE HARDWARE UTILIZATION LIST INDEX

HUL I.D.	ITEM	WBS
1001-00-000	MID-BODY STRUCTURE	1.2.1.1.1.a
2101-00-000	MS/PS SUPPORT MODULE	1.2.1.1.1.b
2102-00-000	X <sub>O</sub> 576 BULKHEAD	.1.2.1.1.1.b
2103-00-000	PAYLOAD WIRE TRAY	1.2.1.1.1.g
2004-00-000	X <sub>0</sub> 679.5 POWER PANEL	1.2.1.1.1.f
2005-00-000	PREFLIGHT UMBILICAL	1.2.1.1.1.h
2106-00-000	X <sub>O</sub> 1307 CROSS SUPPORT	1.2.3.1.1.n
2007-00-000	X <sub>o</sub> 576 AVIONICS HARNESS SUPPORT	1.2.1.1.1.b
2108-00-000	MISSION SPECIALIST SECONDARY STRUCTURE	1.2.1.2.2.a.l
2109-00-000	PAYLOAD SPECIALIST SECONDARY STRUCTURE	1.2.1.2.2.a.3
21j0-00-000	ON-ORBIT STATION SECONDARY STRUCTURE	1.2.1.2.2.a.2
3001-00-000	OPERATOR'S CONSOLE	1.2.1.2.1
3002-00-000	DC POWER SET	1.2.1.2.3
3004-00-000	AFT FLIGHT DECK SET	1.2.1.2.2
3005-00-000	DELETED .	
3006-00-000	DELETED	-
3007-00-000	DELETED	
3108-00-000	CABLE SET	1.2.1.2.4
3109-00-000	DELETED - SEE 3108-00-000	
3110-00-000	DELETED - SEE 3108-00-000	•
4101-00-000	X <sub>O</sub> 576 AIRLOCK INTERFACE	1.2.3.1.1.g
4102-00-000	X <sub>O</sub> 660 TUNNEL INTERFACE	1.2.3.1.1.h
4003-00-000	P/L PRIMARY LONGERON FITTING, NON- DEPLOYABL	1.2.1.1.1.c 1.2.3.1.1.a
4104-00-000	AUXILIARY KEEL FITTING	1.2.1.1.1.e 1.2.3.1.1.c
4105-00-000	P/L UPPER CLEARANCE GAGE	1.2.3.1.1.i

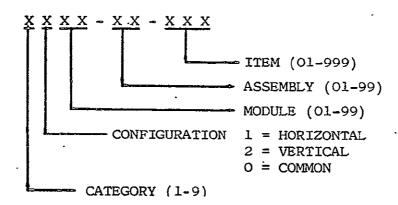


# IVE HARDWARE UTILIZATION LIST INDEX (CONT)

HUL I.D.	ITEM	WBS
4106-00-000	LOWER CLEARANCE GAGE	1.2.3.1.1.j
4107-00-000	OMS DELTA V ENVELOPE	1.2.3.1.1.k
4108-00-000	FLOODLIGHT X <sub>O</sub> 576 BULKHEAD	1.2.3.3.1.a
4009-00-000	X <sub>O</sub> 576 PAYLOAD SERVICE PANEL	1.2.1.2.2.a.5 1.2.3.3.1.f
4110-00-000	ENVIRONMENTAL CONTROL UNIT SET	1.2.3.2.1.a
4111-00-000	(DELETED)	• •
4012-00-000	HOISTING	1.5.1
4013-00-000	(DELETED)	
4014-00-000	TV, X <sub>0</sub> 576 BULKHEAD	1.2.3.3.1.c
4115-00-000	TV X <sub>O</sub> 1307 BULKHEAD	1.2.3.3.1.c
4116-00-000	PAYLOAD BAY LINER	1.2.3.1.1.1
4017-00-000	PREFLIGHT UMBILICAL ELECTRICAL PANEL	1.2.3.3.1.d
4018-00-000	PREFLIGHT UMBILICAL FLUID PANEL	1.2.3.2.1.d
4019-00-000	PAYLOAD BAY FLOODLIGHT	1.2.3.3.1.b
4120-00-000	X <sub>o</sub> l307 BULKHEAD	1.2.1.1.1.b -
4121-00-000	X <sub>o</sub> 1307 P/L OXIDIZER/FUEL PANELS	1.2.3.2.1.b
4122-00-000	PAYLOAD MASS SIMULATOR	1.5.1
4123-00-000	(DELETED)	
4124-00-000	X <sub>O</sub> 1307 ELECTRICAL SERVICE PANEL	1.2.3.3.1.e
4125-00-000	(DELETED)	
4126-00-000	STABILIZING LONGERON FITTING, NON- DEPLOYABLE PAYLOAD	1.2.1.1.1.d 1.2.3.1.1.b
4127-00-000	CABLE SETS	1.2.3.3.1.g
4128-00-000	T-O UMBILICAL PROVISIONS	1.2.3.1.1.m
4129-00-000	T-O UMBILICAL FLUID I/F ASSEMBLY	1.2.3.2.1.c

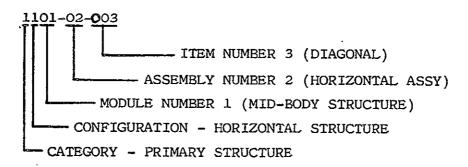


## IVE HARDWARE UTILIZATION LIST IDENTIFICATION CODE



CATEGORY	CODE
PRIMARY STRUCTURE	1XXX
SECONDARY STRUCTURE	2XXX
ELECTRONICS	зххх
OPTIONAL EQUIPMENT	4 <b>Y</b> YY

### SAMPLE:



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25			M c	ع.	1001-01-001		/			LONGERON-BOK	6x10 x 0.50 x 240 TUBE	1	CARBON STEEL, ASTHI-A 500			
ox			MC	z	1001-01-00Z	.				STIFFENER-AUGLE	0.50 x92 x 240 ANGLE"	439	CARBON STOCK	,		
05			nc	z	1001-01-003			, -		CLEVIS PAIL	ZKX3KXZAO, BAR***	310	1018 STEEL	i		
06			MC	2	1001-01-004					BRIDGE RAIL	Z/2 x 3/2 x 240 BAR45	1	1018 STEEL.	: •		
07			MC	3	1001-01-005					POST-END & CENTER	6x6x.3/25x95 TUBE	2/3	CALBON STORE	·		
08		·	Me	4	1001-01-006					DIAGONAL	4 x 6 x . 250 x 103 708	Ţ				
09			MC	z	1001-01-007	٠ _		:		LOWER CORD	6 x 10 x . 375 x 240 TUBE	1	OREBON STEEL	ORIGINAL' OF POOR		
10			MC	4	1001-01-008					GUSSET - ENO POST	13×21×.50 PLATE	38.4	ASTM-A-572B GRADE 50	GINAI POOR		
1/2			M C	Z	1001-01-009					GUSSET - CENTER (UP) GUSSET - CENTER (LAN)	9x//x. ZSO PLATE 9x/3x. ZSO PLATE	8.2		QUALITY		
13			MC	<b>4</b> 2	1001-01-011					GUSSET - DIAGONAL GUSSET - CORNER	9×11×.250 PLATE 11×13×.50 PLATE	20.2	45774.4-378.	AG		
15	·		MC BC	4	1001-01-013			VESETICAL PS	WH 455	GUSSET - HORIZONIAL BOLT & NUT	15 DA X 31/2 LONG	3.8	ASTH. A-STED GRADE 50	日田		111
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18			M C		1001-00-002	,				KEEL EGAM	10 x 5 2 WF x 115 (I)	ŧ	1			
. 70			MC		1001-00-003	•				TIE ROO	1/2 O/A. x 290 135 "	177				
20			M C	6	1001-00-004					KNEE BRACE	4 × 6 × . Z50 × 7/ TUBE	9/	ASTH-A 500 B			
2/			H C	6	1001-00-005	,		] .		FOOT-KNEE BEKE		. 4	ASTH-A-5728	٠.		
22 23	1 1		H C	3	1001-00-006 1001-00-007	1				HERO - KNEE BORCE KEEL LONGERON	5 x 1 x.375 x 10 TUBE 2 x 3 x 2 1/2 x .250 x 240	23 255 7384	ASTH-A500 B ASTH-A 36			
24 25	12/	1.10	H C	6	1001-00-008 1001-00-009		•	<u>'</u>		CLIP-C'BEAM CLIP-K'BEAM	7×7×.375 PLATE 3/2×7×.375 PLATE	5.2 2.4	ASTM- A36			
26	121	1.1.4	BH	4	1001-00-010					BOLT LEVELING	1 7 DIA X 14 (UNC -2 THE		ASTH-A325			
27 28		1.1.a	8 C	108 12	1001-00-011 1001-00-012			<b>!</b>	• •	BOLT - C'BEAM BOLT - KNEE BOKE	1/2 DIA X 1 1/2 * LONG	1.2		<b>'</b>		
29		1.1.14		4	1001-00-013	1	SECTION	l · · · ·		NUT -LEVELING	1 2 ZW. (CMC-Z THECHO)			<u> </u>		

2400-110-0100

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		14	-		2/02-0/-005			VERTICAL STIFFENCE		1.7			
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		77	- 1	Z	2102-01-011			NORIZ. STIFFENER	Itex I'E X. OGO X GO ANGER	1.1			
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		#	~		Z/0Z-p/-0/4	· · ·		344	7 2X 2X . 125 X 50 ANGLE.		AL EXTREMS TO		
<u>.</u>		."	H	2	2102-01-015	X. 576 BUCHENO.	UPPER BULKHEND ASSY.	ATTACH SEACKET	5819 X. 060		WEL SWEET		
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		i	m	1		2007-01-003		# 9**			1000	ORCHEST MESSON (TOUTS)			
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		Z,1.Z.t.e.Z	m /		,	z/10-00-000	ON-DRBIT STATION SECO	VDARY STRUCTURE		CONCEPTUAL DESARN ONLY.	124	<b>i</b>			111	۱
	1		11 1	4	/	z//0-01-000	•	CABINET ASSV.		18 K 56 K 60 RIMETED ASSY.	40					l
		•	14/1	1	/	2/10-01-00/1			FRONT PANEL	60 × 65 × . 050 SHEET	187	AL SMET		1.111		-
.	1	,	// /	/	2	2110-01-002			SIDE PHANEL	18 K 5 6 K . 050 SHOOT	5.	AL. SWEET		1111		-
		*	11/	4	4	2110-01-003	<u> </u>	,	FRONT SUPPORT AWALE	1x1 x . 050 x 45" ANGLE	.7	AL. EXHLUSTON				
١.	١		11/	/	4	2110-01-004			HORIZOUNIC SUMMO-FRONT	IXIK.050 K 60 ANGLE	:6.	-	<u>"</u> .			1
	1		11/	1	4	2110-01-005	•		VERTICAL SUMMER - RFT	1218.050 × 56 ANGUE	15		-			
	1		11 1	4	ž	2/10-01-006	. ↓	CADMET ASSY	Marizantik Differt -ENO	1×1×.050×18 ANGLE	,z	AL EXTENSION				1
			H h	1	4	2/10-00-001	ON-ORBIT STATION SECO	VORRY STRUCTURE	FLOOR ATTACH	1/2x 2 x . 060 X 2 FEE	7.4	AL ENEUSING				۱
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ITEM NUMBER	WBS HUMBER	SOURCE	Wall Control	QUANITY .	IDENȚIFICATIOM. NUNTBER		CONFIGURATION  H * HORIZONTAL  V * VERTICAL  C * COMMON  ECTRONICS  MENCLATURE  ASSEMBLY	X . TEST		WEIGHT	MATERIAL   SPEC.	DRAWING OR PROCURENONT SPEC. NUMBER	PENELLOMENT STATEM STAT
	1.2.1.2.1	888 H H B B B B B B B B B B B B B B B B		1 111.6 1 1 1 1 1 1 1 1 1 1 1 1 1	3001-00-000 3001-01-000 3001-01-000 3001-01-000 3001-01-004 3001-01-007 3001-01-007 3001-01-007 3001-01-007 3001-01-007 3001-01-007 3001-01-007 3001-01-007 3001-01-007 3001-01-007 3001-01-007 3001-01-007 3001-01-007	OPSERTARES COMESCE	THEST MESSLOSMENT UNIT	TAPE READER.  DISK DRIVE  HAGNETIC TAPE UNIT  CRT / KEVBOARD  CONTROL PANEL TEST STORY  CONTROL SAVE (GLOOU)  CLOPU - I/O UNIT  ELECTRONIC RACK  CARD DEPARTER	28 K PRIMARY CONSISTS OF THE CRETARION LINE PROPERTY LINE PROPERTY LINE VIENTALY	1800			X

VUMBER	NUMBER	JEFTION	7	IFICATION	SOURCE M • MAKE B • BUY O • OFF THE SHELF		X · TEST	OMTS. REQUIRED TY FOR TEST	. 44	EIAL   SPEC.	us or sea.	TEST REQUITS
ITEM ,	wes A	SOURCE	OUANI	TUENT 110013		TECTROVICS MENCLATURE ASSEMBLY	ITEM	DESCRIPTION	WEIG	MATERIA	DRAW!! PROCUR NUMBE	EVELOPA VECENT
	12.12.6 	S H H HAMA H MANARA M MANARAMANA B B B H	1 1 1111 1 111111 1 1111111 1 1 1 1 1 1	3004.0/-000 3004.0/-001 3004.0/-002 3004.0/-002 3004.0/-002 3004.02-000 3004.02-005 3004.02-006 3004.03-006 3007.00-000	NAT PUINT DECK SET	PRYLOAD STATION  FRYLOAD STATION  ON-ORBIT STATION  MISSION STATION  VISSION STATION  4C INVESTER	ENCLOSURE THEEMAR CONTROL AS AUGO ASSEMBLY PATCH PAMEL WIRING COTY MONITOR MISSION THER PATCH PAMEL WIRING  C & W AND SAFING PIL POWER CONTROL COTT KE POWER WIRING  C & W AND SAFING PIL POWER CONTROL COMMUNICATION CO COMMUNICATION CO COMMUNICATION CO COMMUNICATION CO COMMUNICATION CO COMMUNICATION CO COMMUNICATION CO COMMUNICATION CO COMMUNICATION CO COMMUNICATION CO COMMUNICATION CO COMMUNICATION PROTECT ANNOL WIRING  MICROPHONE LOUD SPEAKER  CABLE SET	FAMEL FAMEL FAMEL FAMEL FAMEL FAMEL FAMEL FAMEL FAMEL FAMEL FAMEL	,	cereo.	OF POOR QUALITY	
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	LICHI MUITEEK	WBS NUMBER	SOURCE	3	IDENTIFICATION NUMBER		CONFIGURATION H : HORIZONTAL V : VERTICAL C : COMMON  ELECTRONICS ALENCLATURE	X · TEST		NEIGHT	MATERIAL   SPEC.	AUING OR XURENGNT SPER. MSER	LOONENT BESTON B
'	1	3	88	Ŏ	45	MODÙLE'	ASSEMBLY	ITEM	DESCRIPTION	3	£	485	25 S S S S S S S S S S S S S S S S S S S
			M N M H		3/08-00-000 3/08-0/-000	l P	AFT FLIGHT DECK TO OPERATORS CONSOLE CADLE ASSEMBLY				·	;	
			14 14	.	\$08-02-000		E-GROUND ASSEMBLY SUBSYSTEM TO SINGLE POINT GROUND.)					•	
		· · · · · · · · · · · · · · · · · · ·	א נמ א נמ		3/08-03-400 3/08-04-000		OPERATOR'S CONSOLE TO D.C. PRIMER SET CABLE ASSEMBLY OPERATOR'S CONSOLE TO PRYLOAD ODDIANT			.			
			74 PA		3/08-05-000		UNIT CABLE ASSY. FACILITY FOLLORE TO SUB-SYSTEMS					•	
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	3	13	ÿ	11.00			· · · · · · · · · · · · · · · · · · ·	15		300	
	<b>У</b>	21:	120	100	OPTIONAL EQUIPMENT. NOMENCLATURE		<u> </u>	2		1 200	188378
	8	33	300	17.3		INEM	DESCRIPTION	3	Ž	823	120002
_	<del></del> .		-		MODULE   ASSENTALY			<del> </del>	<u> </u>		1965
$\cdot  $	23/1.9	11 11	,	4/0/-00-000	HO STE ARCOCK INTERFACE			108	-		
1	٠,	HH	,	4/0/-00-00/		RING-RILLOCK THINKS.	50" DVA X Z X . 625	48	,	1.	
- ∤∙	'	" "	ź	4/0/-00 00Z		ATTACH SEACKET	ZX 6 K ZOO X 18 CHANNE	9	014850N 37550 ASTAT-A-36	COXEZ CHIME.	
		HN	. 2	4/0/-00-003		ATTHCH BOYCKET	ZXG X. ZOO X/4 CHANNEL	9	CARBON STOR	C6 FBZ CHAMEC	
	!	B N	8	4/01-00-004		BOLT	KEDIA X 13 TPI X 1 LONG	z			
		11.		1		30LT	1/2 DIA K / LANG-FLAT MEAD	.e			
	i	8 #	12	4/01-00-005	X. 576 AMELOGE AMERICAN		12 WAY 12 CONG-PUN NEND	'			
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	123111	ر ا بدا	,	4102-00-000	Ho 660 mare, INTER CACE		1 '	183			
ľ		1					4m*			'	
		M H		4/02-01-001		RNG- TUNNE WHERMOR	47" DIA X 3.5 X, 625 PLATE	ļ	CHECON SHEEL ASTM-A-84		
		HH	1	4/02-01-002		SUPPORT - LEFT	1/2× 4 x 247 x 98 CHANNEL 1/2× 4 x 247 x 98 CHANNEL	- 51	1.574-11-34	CF K6.25 DHAWEZ	]
	·	M M	1	4102-01-004		SUPPORT - FWO	1/2 x 4 x . ZET x ZE CHANNEL	1	}		
		H H	,	4102-01-005	- X0660 TENNAS INTOCHOS	SUPPORT-AFF	1/2 x & x. ZE7 x 69 CHAMMER	.36	15 11-1-136		
١	•							1.	• •		
	2.3.1.1.a 2.1.61.C	9/H C	,	4005-00-000	PROMEY LONGERON FITTING.	1	1 ;	1			
ľ		% c	,	1003-01-000	NEW-DESCONDE POPLINO.  JOURNAL ASSEMBLY	l :	·		] ; ,	의 유	
	•	<i>2</i> ∕√ c	, ,	4003-01-001		VARE JOHNNE.	5X5 X /3 FORGING		⇒75€€Z	,	
						•	SK6K13 FOCGING		उरहेटर.	GINAI	
		of C	′	€003-01-00Z		LOWER JOVENAL	The state of the s		MOONER 7/8		
		B C	/ ~	4003-01-003		SHEARING - BALL *	E.O. O. K. F. COO SPHERKAL MADE	9	AVS 5463 STEEL A-284 STEEL	PAGH IS	
		8 C	1	4003-01-005		BEARING-RACE	1.750 x 5.375 EVANSTER. _2.0 EVA K 3.125 LONG	1	Arts 5737	AGI	
		MC	<i>',</i>	4003-01-007		JOURNAL LOCK BOLT	1.0 DIA. 8 3.0 CONG	1	SPECL		
1		6 c	,		PRIMARY LONGOROM FTG. JOURNAL ASSEYBLY	BOLT	10 an. x 4.50 Lava	-	STEEL	36	1111

ITEM NUMBER !	SOURCE	CONFIGURATION	TUENTTICOTION NUNTBER	SOURCE CONFIGURATION  M · MAKE H · HORIZONTAL  B · BUY V · VERTICAL  O · OFF THE SHELF C · COMMON  TOPTIONAL EQUIPMENT  NOMENCLATURE  MODULE ASSEMBLY	X· TEST	OESCRIPTION	WEIGHT	MATERIAL   SAEC.	DRPUING OR PCCUREMENT SPET. NUMBER	EVELOAMENT  WHISTORIOM  WELLINGS  WE
123.11	c H	+	4/04-00-000	- AUKILIARY KEEL FITTING		7×0/2×2	21	5755C		
- 123.22	A A A A	N Z N Z N Z H I	4/05-01-000 4/05-01-001 4/05-01-002 4/05-01-003	PIC UPPER CURRENCE GAR.  LEG SUPPERT ASSY.	STIFFENER  STIFFENER  GUSSAT  STIFFENER	1/2 x 2 x . 060 x 97 : AWGLE  1/2 x 2 x . 060 x 58 AMALE  1/3 38 x 97 x . 060 SHORT  4x 20 x 1/ x . 060 CHANNEL  1/2 x 2 x . 040 x 7 ANGLE	;73 25 2 12 11 .6	AL. ETTENSION AL. ETTENSION AL. SHEET AL. SHEET AL. GOTHUSTON		
	H H H H B	H 1 H 2 H 2 H 2 H 1 H 1	1	LEG SUMMORT ASSY.  THEROWORK SUMMET ASSY.  HOCKDONIAL SUPPORT ASSY.	ROLLER SWAME, L.S. ROLLER SPACER SPACER SOLT  PANEL STIFFENER STIFFENER	4 X 10 X . Z 50 SHEET  A X 10 X . Z 50 SHEET  Z . 5 D/A X 1 . 5 LONG  1 5 D/A X 1 LONG  . 50 D/A . X 3 LONG  . 50 D/A . X 3 LONG  . 15 X 9 Z X . 060 SHEET  1/2 X 2 X . 060 X 9 Z ANGLE  1/2 X 2 X . 060 X 15 ANGE	1 1 7 .2 .8 13.6 8 2 0.4	AL SMET.  AL. SMET.  AL. ROO  AL. ROO  AL. ROO  AL. ROOT  AL SMEET  AL STRUGGEN  AL STRUGGEN  AL STRUGGEN		

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AND THE CAN	WALE Z	15.	1201164	\\\\\\\\\\		PONROE PO	, I	) + x557		,,,		'G OR STER.	TEST REGNITS
1	۲ ع	50.6.4.0.6	COREL	0000	Z5 - 17. No. 108	. MODULE	LVCLATURE ASSEMBLY	ITEM	DESCRIPTION	-W.	ME	DEA !	SPAILE RECELL SVST. SPST.
	12,31	ندب			4105-00-001	UPPER CLEARANCE	dise (conto)	SOLT	9 K 15 K . 375 SWEET - 94%	\$.5°	AL SHOOT		
-	-	<i>A A A A</i>	, <sub>1</sub> , <sub>1</sub> , <sub>1</sub> , <sub>1</sub> , <sub>1</sub> , <sub>1</sub> , <sub>1</sub> , <sub>1</sub>	Z	4/05-03-002 4/05-03-000 4/05-03-001 4/05-03-002		SLOS GARC ASSY	GARAGE - SORPHENT HETTOCH CLIP	12 x .060 x 87 shear 50%	4.2	AL SMITT		
	1	<i>n</i>	, ,,	,	4/05-04-000 4/05-04-00]		CENTUR GREE ASSY	GAGE SEGNENT	IZK.OGOX 90 SHEET . THE	6.5	AL. ENGET		
	123.11	ľ	ע ע	2	4/05-03-002 4/05-00-003 4/05-00-004 4/06-00-000	- LOWER CLEMENICS G	GASE	SUIDE RAIL INSTALLATION TOOL	1/2 x 3 x 3 x . /38 x 240 250 1/2 x 3 x 3 x . /38 x 240 250 DESKA TEO.	105 37 154	AL EXTEUSION		
		M	N	,	4/04-01-000 4/04-01-001 4/04-01-002		CONTRE GRAS ASSIV	CENTER TEMPLATE	16 x 88 x . 188 SWEET . 50%.	205	<b>,</b>		
		. H	//	z	4106-01-003 4106-01-004 4106-01-005		1,	TAMISONAL SUMPAT!  TAMISVOZSE ANGLE  BASE SUMPORT	1/4 × 1/2 × .125 × 57 ANGLE  1/6 × Z × .125 × 20 CHANNEL	. 10	AL EXTRUSION	PIS	
	,	14 15	//	4	4/06-01-006 4/06-01-007 8/06-01-008			ROLLER. " "AKLE NUT	24 DA XI ROD .250 TVA X 18 LONG BOLT250 TVA X 18 LONG BOLT.	. 6 . 7 ! _	AL ALLOY	N FI	
		B		9	4/06-01-007 4/06-00-001 1/06-00-002		CENTER SAGE ASSY	WASHER / SAIM CENTER ROWER GUIDE	18 x 3 4 x . 28/ x 240 chimer. 18 x 12 x . 125 x 240 chimer.		NL ALLOY AL <del>CATRUSION</del>	国际	

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NUMBER	VUNDER	F	77	TEICATION ER	SOURCE M. MAKE B. BUY O. OFF THE SHELF	. <u>CONFIGURATION</u> H * HORIZONTAL V = VERTICAL C : COMMON	X · TEST		1.1	eint f sæc.	ug OR EMENT SPETE. R	TEST REGINTS.
Š	١ , , ,	38	\ \{\}	22		I <u>L EOUIPMENT</u> ENCLATURE	· · · · · · · · · · · · · · · · · · ·	•	Ŷ	\ \( \tilde{V} \)	3 8 8	15 25 3 2 5 2 5
18	9	33	3	63	7000	ENCENTURE .	<del>, </del>	DESCRIPTION	3	8	\$85	1213612118
14.	£3	। ४ । ८	ğ	. 44 .	MODULE	ASSEMBLY	ITEM		>	*	2 4 %	15/5/5/5/5/5/5/5/5/5/5/5/5/5/5/5/5/5/5/
F	1.2.3.4.6			476-00-000	LOWER CHERNICE OF	se (courb)	4 34ml					
}		HH		4/06-02-000		SIDE GAGE ASSY.			20	- 16		
1;		N N	<b> </b>							<u> </u>		1111111
	1			4/06-02-001	,	,	SIDE TEMPLATE	13x66x.188 SHOWY -57	L	AL. SMET	* ***	
		M N		4106-02-00Z			CHISONAL SUMMER (PANO)		ł	AC SHOW		
,	}	MH	1	4106-02-008 4106-02-004		,	DIAGONAL SLAMET (AFT)	11 x 19 x .060 SHEET -50%.	1.6	AL SHEET		
, }		MH	Z	4106-01 005			BASE SUPPORT	18x2x.125 x20 ammunes	1.5_	AL EXPENSION		
<b> </b>	,	MM	2	4/06-01-006	· [ ]	·	ROLLER	2 4 WA X1 ROO	٤,٤	AL ALLEY ROD		
		5 H	4	0/06-01-007		*	AKLE	. 250 DIA. K / 16 LONG BAT	./	AL ALLOY		]
		H H	!	4/06-02-005			DIRECVAL	10 x 16 x . 040 SHEET	.5	AL. SHEET	* * * * * * * * * * * * * * * * * * *	<b>]</b>
		Ø W	1	4104-01-008		•	DIASONA"	10 x 16 x 060 SHEET _	<b>.</b>	AL. SHEET		
<u> </u>		K K	1	4106-01-009		SIDE GASE ASSY.	WASHER / SHIPP ROLLER QUIDE	TOX3 KX. 250 X 240 CMNNEL	34	AL EXTRASSION		<u> </u>
1		NN	9	4/06-00-005	- LANEE CUERRANT GAST		SUPPORT BOLDET	AKOK IBB SHEET ST	.5-	AC, 545 <del>17</del>		<u> </u>
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## IVE HARDWARE UTILIZATION LIST

UMBER	MOER		KATION		1,F1CATION IER	SOURCE  M • MAKE  B • BUY  O • OFF THE SHELF	CONFIGURATION H: HORIZONTAL V: VERTICAL C: COMMON	X · TEST		7	200K   5PEC.	is or	REG	557		
13	Ž	५	30	5	ા કરા	OPTIONAL	EQUIPMENT	,		1 6	, X	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	3 3	3 3	8 7 5	}
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A	Ž	500	ŝ	OUR	24	MODULE	ASSEMBLY	ITEM		3	1 &	985	220	200	3 3 6	
F.	/2.3././. A	· //	c	,	4/07-00-000	ONS DV ENVELO	· ·	1	PESIGN IS TED. USE TIMORE	-	·					
		14	٥	,	4107-01-000	; j   · · ·	500 FPS A VELOCITY		TOP ADD ON MPAROACH.	li L	j					<b> </b>
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ł		"	C	′	4/07-02-000		1000 FPS & VOLOCITY	, ,	ADD ON TO BASIO EOU FFS	ļ.			$ \  \  $	11		
- 1		1	e	1	4/07-05-000	OMS'AVENELORE	1500 FRS A VELBERTY	,	1800 ON TO 1000 MPS.	ŀ	ļ.					
1	1.2.3.3.24	e. H	#	1	4/08-00-000	FLOGOLISHT X65%	] ]	<u> </u>	<u> </u>	2.4	;	i				
		8	*	,	4/08-00-001	BULKHAND.		POCKIMS FEBOOLASHT	LIONT SECRETION TED.	; z	1			11	.     ;	
	-	1,1	<b> </b>	2_	4/08-00-00Z	;		BRACKET	AKBKIOSO SMET	1.2	AL. ALLOY	j;				
		1	,,		-	<u>.</u>		WARE MEENESS		ľ		,	-	11		·
ĺ	1	1			4/08-00-003	' ''	•	,	1	li	•	<b>!</b>				
	ŀ	B	a	/	1/08-00-004	FLOODLIGHT - X0576 BU	ſ	SWITCH	\$ \$	i	<b> </b> ;				11	
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	12.1.22.0	<u>4</u>	c	4	4009-00-000	K STE PHYLORD SHEVICE	PAGE		,	2.5	ļ.					[ , ; ]
1	1.23319	14	c	/	4009-01-000	· [	SERVICE PANEL ASSY		Ì	2.8	þ			1	1	
	1				•	,	1		* 8 × 16 × 188 SHEET -5%		R. suer	Ϊ, ,			1	
		"	1		4009-02-001	, ,		ARNEZ-	WSL5W FEED-THAU CONNECTO	1.	M. SHOC	40N3827			ŀ	
1		₿	٢	9	4009-03-00Z	X6 576 PHYLAND HOVEL	SERVICE PANCE ASSY	CONNECTOR -PEED THRU	DUSTSCH RECTUR CO.		ł	CONTRACT SPEC				
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	123,214		н	1	4/10-00-000	ENVIROMENTAL CONTROL	, , ।	i	DONAN IS TED	1						
		6	N	,	4110-01-000	UNIT SET	HEAT EXCHANGER	l , ~	,	f.	•	SZ.	], [			
		1,4	N	,	N10-02-000		CONTROL & DISPLAY		;			POOR	$\  \ $			
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Ï		8	#		410-03-000		PURGE & TEST	! !				SO P				
		0	"	1	4/10-04-000		FLUIO LINES SET	! .	ľ		ľ					<u> </u>
,		M	Н	1	4/10-05-000	ENVIRONENTIK CONTROL UNIT SET:	INTERFACE FANGL ASSI									
		1			<u> </u>	CONT SCI.										

## IVE HARDWARE UTILIZATION LIST

ITEM NUMBER	WBS NUMBER	SOURCE CONFIGURATION	QUANITY	IDENTIFICATION NUMBER	CONFIGURATION  H * HORIZONTAL  V * VERTICAL  C * COMMON  L * EQUIPMENT  NCLATURE  ASSEMBLY	X . TEST A		WEIGHT	MATERAL/SPEC.	ERAMING OR PROCINCAINT SOED. NUMBER	SECONDANENT SECONDANES SURTE SECONDANES SURT S
-	05LFM0	80 % % 8 % % % % % % % % % % % % % % % %	+ + + + + + + + + + + + + + + + + + + +	4///-00-000 4///-01-000 4///-00-001 4///-00-002 4///-00-003 4///-00-000 40/2-00-000 40/2-00-001 40/2-00-003 40/2-00-004	FOTER ASSY - SWIVER.	CHRSONAL-SUPPORT  PLATE - SUPPORT  ANGLE - ATTHCH  DIFFONAL  BOLT  BEAM- SPREADER  ATTACH I UG  CLEVIS,  BOLT	NO. SOSNIZ - 1800 AEROL CO 2×2×. ZEO × 43 TVOS  11 × ZE ×. 375 PLATE  8×10 × 1 × 11 AWELE  6×6 × . 75 PLOTE  .50 DA × 2  MC10×21.9 × 204 CAMMEL  4×4 × .750 PLATE  74 DIA BOLT  .50 DIA × Z	19 (ca) EP 35 .2	L.A. COLIE. CAEGON STEEL. ASTA-ASDO GORGON STEEL ASTA A 232(1) GALGON STEEL ASTA-ASG  CAEGON STEEL ASTA-ASG CAEGON STEEL ASTA-ASG STEEL ALLOY STEEL ALLOY STEEL ALLOY	PELENED	
	eres .	M H H H H B H H H H B H	2 1 1 4 / 2 / 2	4015-00-000 4015-01-000 4015-01-001 4015-01-005 4015-02-000 4015-02-001 4015-02-005	PITTACH ASSY.  PONGUE ASSY	ILEH-B-OZAG TOW E  ATTACH INGLE  LUG  BOLT  TOW BAR (LER)  CROSS DARE  CLEUIS  RING	AR)  6×6×.375× & ANRIE  6×6×.75 PLATE  .50 DIA. × 2 LONG  3°0/A × 250 × 140° TUDE  3°0/A × 250 × 120° TUDE  4° DIA. BOLT  1.5 DIA × 6°0.D.	280 185 (16.0)	CARDON STEEL ASSTY - A36 CARBON STEEL ASSTY - A36 CARBON STEEL ASSTY - A36 CARBON STEEL ASSTY - A36 STEEL ALLAY STEEL ALLAY	DESERTED	

VUMBER	UMBER	URATION	۲	iFicarrad E.R	SOURCE M • MAKE B • BUY O • OFF THE SHELF		X · TEST A		24	ent   sæc.	U, OR ENEUT SPEC. T	TEST REGINTS.
TEM !	WBS A	VELL	non	DENT WMS	NON	NAL EQUIPMENT MENCLATURE	-	DESCRIPTION .	WEIG.	MATERIAL	ERWIII COCUR ITTGE	ELOPA PINA FENIN
Z	3	88	9	Ν×	MODULE	ASSEMBLY	ITSM	, .	\	\ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \	4 % %	3888888
•	1.233/C	c	,	4014-00-000	TV , X. 576 BULL			,			,	
	,	e	/	4014-01-000		TV CAMERIA ASSY				:-		
•		8 C	1	40/4-01-001			CAMORA, BRW.	CONO PROSEL 4410				
		8   c	.1	4014-01-002	•		LENS	CANNON 15-105 HM BODA				
		Ø C	1	1018-01-003		•	NENO-THE S PAN.	PERCO # 550	ļ			
		0 4		4018-01-004	· .	TU CANGEA ASSY.	VIDECON TUBE	2CA #0521		1		
		// C		1014-02-000	,	ZENOTE CONTRAC ONLY					,	
-	<u> </u>	<i>5</i> C		4014-02-001			THET & THE CONTROL	PELCO	•			<u> </u>
	:	B C		4014-02-003		PETTOTE CONTROL WIT	CONTROL PANEL			N SNEET	1	
	. •	МС		40/4-03-000		CANCEL INST.		· · · · · · · · · · · · · · · · · · ·				
		# c		4014-03-001		,	15ASE	15×10×.050		AL SHEET		
_		4 C	Z	40/4-03-002			BLKKE-	15 x 6 K.050	-	AC SNEET	••	
	1	H C	,	4014-03-003	.   1	CAMERA INSTE.	CABLE SET	Power, CONTEST & MONITOR				
; -		5 C	,	4014-04-000		TV MONITOR, ASSY	•				<b>*</b> .,	
•		8 C	,	4014-04-001	,		MONITOR UNIT	SONY # 900 - 9" SCEEN	·		0,9	
,		H C	′	4014-04-002	•	•	BASE	12×12 K2050	٠-	AL. SHEET	OF FO	<u> </u>
	••	H C	۲.	4014-05-003	TV, 4.576	TV MONITOR ASSY	BRACKET	.4 × 12 × 050		AL SHEET	OF FOOT	展
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UTOKK	NUMBER		NOTION	۷.	FICATION	SOURCE M • MA B • BU O • OF	9KE	H = H	KGURATION DRIZONTAL ERTICAL OMMON	X . TEST	<del></del>		encloses.	V, OR ENENT SOEC. E		ST 11175.	
	SS N	URCE	77/0	ine	ZENT.			NAL: EQ NENCLAT	UIPMENT_ URE		DESCRIPTION"	15161	1,07E	Servin Secura In Ge	EKER VEKER	375	4863
i	3	જુ (	ĝ į	Š	. 48	MODE	ULE	Ass	EMBLY'	ITEM		2	<u> </u>	485	803	55.55	S
		_	# .	/	4115-00-000	73/ .	X0 1307 BUL	CVENIO	1	•			<u>.</u>				
	•	<i>B</i>	<b>₩</b> .	/	4115-01-000			TV CAME	EA ASSY.	•	1			· ·			
	• •	5	<b>,</b>	/	4115-01-001					CATACRA, E & W	COND MODEL #410			~~			
		8	٠ ۲	1	4115-01-002	•				LENS	CANNON, 15-105 MN BOOKS	_				1111	Ì
		8	4	<b>′</b>	9115-01-003			·		TILT & PAN HEND	PELCO MODEL 550	-					İ
.]		8	"	1	4/15-01-004	ı	<del>-</del>	TV CAMES	THE MEST	VIDECON TUBE	200 # 852/	-	•			1111	
	,		1	- [	4/15-02-000			ZE MOVE C	ONTROL WIT					1, 1			l
- }			1		4//5-02-00/					EADN CONTROL	PERCO				111		ļ
		8		ı	4/15-ac-002	•				CONTRAL PANEL .	perco	-	4. 5				
İ	•	4	1	- 1	4115-02-003	•		CANOSH I	MISTE.	CONTRACT PAGE .		-	AL SHEET	:	$\{ \} \}$		
	, \	א אין	4	ź	4115-03-001 4115-03-002 9115-03-003	•		,		EMSE SUPPORT-VERTICAL TRACKOT	/2x /8 x . 050 2x 2 x . 060 x 60 AMGLE 8 x /8 x . 060 SWEET		AL. SHEET CHTEUSION AL SHEET	·			
		1	1		415-03-004			CANONCA A	MSTE.	CABUS SET	POLICE, COURSE & MONTHE			1.45			
			1	,	4115-04-000		.	ואפא עיץ	THE ASSY .	, ,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,						.	ļ
	•	8	ر ار	,	415-04-001			:	: '	MONITOL WAIT	SONY 900 , 9" SOMEN			., .			l
	, .	1	ν.	,	4115-04-002	•		·.		BASE	12 x 1, 2 x . 0 50	:	AC. SINTET				
		א ניו	/ 2	2	415-04-003	TV.	Xx 1307	TV MONN	rae MSSY	BEACKET	6 × 12 × . 050		AL. SHOWT				
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	I		11	-[		SOURCE CONFIGURATION	V. YEST REC	MTS.			1 1	-	
- 1	1	٠,	<b>!</b>	1	1 i. '.	M. MAKE H. HORIZONTAL		REQUIRED		[   <b>\</b> {\bar{\partial}{2}}	<b>5</b>	7857	∤. I
Į,	ا ره	tv.	13	:	1 8 1	B . BUY V . VERTICAL		TY FORTEST		1 1 1 m	9.	REGINTS.	
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			يُ ان ا	}	1 69	OPTIONAL EQUIPMEN	7		[3]	1 18 .	1 3 8 6	\$ 3 <b>3 3 3 7 9</b>	i i
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L			9	1		MODULE ASSEMBLY	ITEM			<u> </u>	44.5	28 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8	ł I
F		1.2.5.1.1.2	MH	3	4//6-00-000	PRYLORD BRY LINER	T .			×3 - /479	1		
- 1	ľ		" "	-	1776-000	· · · · · · · · · · · · · · · · · · ·		<b>i</b>	495	A: セッタ"			
			MM	14	4/16-00-001	] 1	FRAME	.90 K Z40 K. 050 X 100 Ner	z.z	ALUN ENTER	115043-3521-0055		
-1	ł	- •			1			1	. (	1 4 .			
†	-	•	H N	1	4//6-00-002	•     •	FRAME-END	75 x . 875 x . 043 x / 00 25E	1.5	[ ]	AND/0/38-0704		
1	1		11 11	0	4/16-00-003		SUPPORT-LONGERON	20 K 1.75 K . / 25 K 840 · TEE	11.4	]   .	ANDIO/36-2008		j
- 1	J		H H	8				SAGET	-1				
		•	" "	۱۳	4116-00-004		STUPPONT- CAMCERON	/X 2.00 X. 050 X 60 ANGLE	.#1	AL. EXTRUSION	<b>'</b> }		1 1
- {	ŀ	-	11 H	6	4116-00-005		BEICKET NO. 1 (LER)	4x4x.030 Sweet	1.1	AL SHEET	, 1		1 1
	-		HH	ء ا	4116-00-006			605	, ,	•			
-	- 1	•	"  "	-			BRICKET NO. 2 (LIR.	6 x/7 x. 040 .swær	.5				
- 1	ĺ		H H	6	4/16-00-007		BERCOT NOS (LER)		.#	.			
1	-	.'	77 H	ے ا	4//6-00-008		BRICKET, NO. 4 (LER)	4×7×.050 SHEET	1	AL. SHEET			
	- 1						,		(. 50)	L.C. STEEL			
٠			MH	4	4116-00-009		LINING NO.1	FORIZO RYEX 14 -NO.13	32.	EXPHNOED AND	*		
	Ţ	· [	17 11	8	416-00-010	1 1	LINING NO.Z	36 K/20 RHEX & - NO.15	24	:			]
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	- 1	:	`` ''	-,	-,,,	PRYLOAD BAY LIMBR	LINING NO. 3	24 A / 20 RIER /4 - NO./3.	16	EXAMMOTO ME	742		
	- 1			'	1 [		Į.	· ·		i		]	1.1
-	1	12.3.3.46	" C	′	10/7-00-000	PERFERSIT GRADICAL PERCIPOCAL PROPER	,	, ·	5			111111	
		·	MC	1.	4017-00-001		UNBILICAL PHINES	13 K/8 K. ZSO. SHEET	اح.	AL : MEET	]		
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ļ		: l	٦ ٢	750	4017-00-00Z	MOREMANT UNBICKE	ELECTRICAL COMMECTOR	LOCATION & NUMBER NOT TED.	Ιi	1 1			1
- 1		- 1	-		1					,	,	ა[ <u>წ</u> .]	1 1
	- 1	.Z.3.Z.Ld	HC	1	4018-00-000	· PROFESCHT WIBLICK FLUID PANES	<b>!</b> '		5-	,	] ]	う뙤!!!	l ' I
- [	- 1	•	4 e	1	4018-00-001		NHOILICAL PANEL	IIX IS K. ZSO SHEET	5	AL SMEET	]	(H) [ ] [ ]	
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- 1	ł	- I	4 6	780	4018-00-001	-PROPERSHY CHASTERED FLUID PRINCE	FLUID COUPLING	LOCATION & NUMBER ADS THE	<b>.</b> !			[4]	l - [
	1	.	- 1				1	· ·	;	' '	[		
- 1	/	12.3.3.1.6	11 C		1019-00-000	serioro des riscours	ļ ·		43	}	1	[科]	
- 1	- 1		8 6	6	4019-00-001		FLOODLICHT	GX/ZX4 BOX (LIGHT 180)	4:	l . •	]		
- }	ŀ	·	4 6		1019-00-00Z	·	BRACKET & UPPER	10 X /U X. OGO SMEET	٠.6	AL. SHEET	1	1 1 1 1 1 1	
1	- 1	- ŀ	7/6		0019-00-003	1 • 1	BEACKET) WANT	10 K 10 K . 060 5MSET	6	AL, SMEET			
j		, _ l	4 C	4	4019-00-004	1 4	BRACKET" LOWER	LOKZZ K.OGO JAHAT "	24	AL SWEET	i l	11111	i i
Ì	1		4 6	1	2019-00-005	1	BRACKET SLIGHTS	15 X 18 K. OGO SWEET	1.6	AC. SMET	1 1		i 1.
: L_		1	M 6	6	4019-00-0061	2000	WAT MACNES		raa.	L	<u> </u>		<u></u>
						,	SWITCH	•	•/		•		

UMBER	וועאפֿבּּ	,	JEATION	,		FICATION	<u>SOURCE</u> M • MAKE B • BUY O • OFF THE SHELF	M · MAKE H · HORIZONTAL X · TEST REQUIRED				PIAL   SPEC.	16 OR SERV.		E S		
LTEM A	WBS N	OURCE	ממעוני		ילאין המאשר		ONAL EQUIPMENT MENCLATURE	-	DESCRIPTION	WEIGH	IMTER	COUR COUR	VN5'073	72/2	- 51/5/	Nes.	
7	3	જે	9 6	,	44	MODULE	ASSEMBLY	ITEM.	•	>	,	483	220	38	305	38	
	1.2.1.2.1.6	11	. <i>,</i>	,	4/20-00-040	. Xo 1307 EXCHOND		•		925	,		П	П	$\prod$	$\prod$	
		14	ر <sub>ا</sub> ر	,	4/20-01-000		BULKHENO ASSY.		HOUR MASY IN DESIGN, PHOTESIAL STIPHOTES SHOWN BY BUSED OF								
		11	٠. · ا	,	4/20-02-000	. Ko 1307 BULKACHO	BULKHEAD ASSY.	X.576 BULLHOW BLAS		1							
	12.3.21.6	77	v   .	,	4/2/-00-000	Xo 1307 PM, OKKUZERI, FUEL PROVELS	ł			50		·					
		,	<u>.</u>  -	,	4/21-01-000		OKNOVENE MANNE MEST ( L. A)		<b>'</b> ,	184	<u> </u>						
ł			j.	J	4/21-01-001		, , , , , , , , , , , , , , , , , , ,	PAGNEL .	/2 F/Z K. ESO SMEET	3.6	AL. SHEET						
		8	1	ļ	4/2/-01 <b>-</b> 001			FLUID COUPLING (M)		//	SHAVLERS STEEL						
ŀ				- 1			OKIO/ZEJE PHINEZI, PHINEZ	ĺ	,	5.Z	37642	<u> </u>					
-		M		- 1	4/2/-02-000 4/2/-02-00/		DETATEOR PHONECIPATION	PRINEL	IZKIZ K. 250 SHEET	3.6	AL. SHEET						
-		H	1	ı				ĺ			ļ			Ш			
	,	0	.	- 1	4/2/-03-002	İ	·	COUPLING	1" 010.		SHINLESS						
-		0		<b>'</b>	4/21-02 -003			COUPLING .	78 DIA.	.6	STHINLESS	•					
			.									1					
		11/1	/ /	,	1/21-08-000		FUEL PANEL ASSV(ZS)		_	15.6							
	Ť ×	11/	/	,	4/21-08-00/			PANEL.	12 K12 K. 250 SPEET.	3.4	AL. SMSET						
	٠ ,	ا ه	y .	,	1/21-08-002	t		FLUID COUPLING(H)	3.5 "DIA	7.	STAINLESS					11	
	,	8	y   /		1/2/-03-003			FLUID COUPLING (M)	· 2.5" DIA	5	STRINCERS					41.	
-	: .	11	/ /	,	1/21-01-000	'	FUEL PANGE ASSINES)			14.6					1.1		
		,,,	١,	,  ,	1/21-04-001		ļ ļ	PANEL	ITKIZK . ESO SHEET!	26	AL SHORT						
		5	y ,	-	1/2/-04-002			FLUID COURING (M)	4 * DIA.	8	SMINCESS						
	:	5 4	<i>/</i>	,	421-04-003	<b>]</b> ,		FLUID COUPLING [N]	1/4" DIA.	2							
		B 1		- 1	1/21-04-004	Ke 1307 AL QUEZ	se/	FLUID CONPLINE (H)		/	STAINLESS						
-						FUEL MINELS		, ·	1								

## IVE HARDWARE UTILIZATION LIST

1		<b> </b>	T	Π			SOURCE	CONFIGURATION	<u> </u>	0415		<del>,</del>	]	T TI				
	ITEM NUMBER	NUMBER		MITION						icsraw e	M. MAKE B. BUY O. OFF THE SHELF	H : HORIZONTAL V : VERTICAL	K . TEST	PTIS. REQUIRED TY FOR TEST		1. / spec.	ak Svr sper	TEST REQUITS
	3	}	ų,	ממחני	Ÿ	4.5	77.6	· FOUIDMENT		· ·	1	<b>1</b>	388					
	43		18	131	ממשמו	1 2 2		<u> EQUIPMENT</u> MENCLATURE	4		,5/3,	7.6	1302	400000000000000000000000000000000000000				
	II	WBS	500	CONF	ò	LOEN	MODULE	ASSEMBLY	ITEM	DESCRIPTION	Š	N.	Sec 2	STEVEN STEVEN				
		1.5.1.	Н	н	/	4/22-00-000	FRYLORD PRESS SONULARS		,		65,00	•						
	. `		Н	/	1	WZZ-01-000		. BALLAST ASSEMBLY		· ,	62,52	  -	,					
	_		11	4	,	4/22-01-001			SUPPORT BEAM.	WF 11/2×16×15' I DEAM	1440	57EEL						
		-	M	//	/2	4/22-01-002	<b>.</b>	,	RENFORMS ROO	J"DIA X.IA FEET	37.4	•						
			M	,,	nso	4/22-01-003			BALCAS:	· ·	625KI	CONCLETS						
.			н	н	z	4/22-01-004			HOIST/END PLANE	15 X 31 X 1 PUTE	/3Z	STEEL.						
11			m	4	4	4/22-01-005		EALLAST ASSEMBLY	GUSSET	BXBXI PLATE	9	STEEL.	}	-				
			H	11	z	4/22-02-000		TRUMATION ASSY	.,,	. ,	69							
			11	#	,	4122-02-001	,	_ ' - ' - ' - ' - ' - ' - ' - ' - ' - '	TRUNNION "	DIA KIZ ROD.	zė	STEEL		111111				
$\ \cdot\ $	ļ	•	н	W	,	4/22-0Z-00Z		TELWARAN ASSY.	PLATE	IZXIZXI. PLATE	41	STEEL						
		! !	M	#	8	A/22-00-00/	PAYZONO MASS SHIKKATO	·	BOLT	1" aus. × 3"	.7	STEEL						
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YOCK	nungez		OTION		UTIFICATION 18ER	SOURCE M • MAKE B • BUY O • OFF THE SHELL	M. MAKE H. HORIZONTAL X. TEST REQUIRED				c/sac.	ok SWT SPET.		EST PAITS.
١	Ž	١,	3	7	(F)		- OPTIONAL E	QUIPMENT		Ş	8	38 ×	18 3	
	88	180	17/2/	3	W.B	· NO	MENCLATURE			19/3	. 32	4.00 4.00	400	3 3 3 3
7	<u> </u>	500	Ò	200	17.8	MODULE	ASSEMBLY	ITEM	DESCRIPTION .	WE	18181	2000	DEVE	SPA SPA SPA
- /	232/4	n	H	1.	4/24-00-000	Xo 1307 ELECTRICA	SSEVICE TAMES			9.4				
4		11	H	;	4/24-01-000		MANEL ASSY. (LH.)		·	4.7				
]		11	N	1	4/24-01-001			FANGL (LN)	12 x 12 x 250 SHEET	3.5	AL ALLOY	,		
-	•	8	H	1	4/24-00-00/			COMECTUE - POURE	36 SHELL SVZE COMMECTIVE	.6	AL ALLOY	. •		
-	•	8	H	2	4/24-00-00Z		PANEL ASSY (LN)	CONNECTOR - STOMAL	ZZ SHELÇ SIDE COMMETENE.	.3	AL ACCOP .			
-		11	#	1	4/24-0Z-000	,	FAREL ASSY (RH)			4.7		,		
1	•	11	#	1	4/24-0Z-00/			PANEL (RH)	YZX /ZX, ZSO SWEET	35	ACALCOY			
1	:	8	N	,	4/24-00-001	ļ ,		CONVECTOR - POWER	36 SHELL SIZE	-6	AL ALLOY			
1	•	z	"	2	4124-00-002	X. 1307 SERVICE AND	HANGE ASSY (RN).	CONNECTIVE - SASMAL	ZZ SHELL SIZE	.3	AL ALLOY.		1	
L		,,	,,	,	4120			,						
ľ	TOTAL CO.	77	"		4/25 00-000	KEEL FYTTWO , K-	FANO Y-Y LONGS	-:-	-7×8/cx2	2/-	-572cc	ロニノミンテング		
1.2	21.1.1.d 1.3.41.b	М	"	/	4/24-00-000	STABLIZING LONG NON-OSPLOYABO			IDENTICAL TO KIT 4003-0 EKCEPT SHEPLE PIN (PN &	0-00: 103-0	1-00A)			
1.5	233/9	М	c	/	4/27-00-000	CABLE SETS		,	IS NOT RESPUESTO.			•		
	:	11	ے	/	4127-01-000		Xo 1307 POWER CABLE SE	780	,			Ó.		
1	•	M	c	/	4/27-02-000	CABLE SETS	Xo 1307   T-0 CABLE SET	730					9	
	23.1.1.n	м	11	,	4128-00-000	T-O UMBILICAL PROVI	sions .				,	<b>,2</b>		
†	-	M	H	/	4/28-01-000		L.N. PANEL ASSEMBLY	730 .			, ,	2.6		
1	- l	м	W	,	A/28-02-000	T-O UMBRICAL PROV.	R.H. DAVEL ASSEMBLY.	780			i, 6	E A		
, /.z	23.2.l.c	μ,	N	/	4/29-00-000	T-o uribiçical Fiño X	F P35r.	. <del>79</del> 0 .				STATE OF SOLE SOLE SOLE SOLE SOLE SOLE SOLE SOLE		
					•		t = ==================================							



## APPENDIX B

HORIZONTAL IVE IN-FIELD
ASSEMBLY PROCEDURE

#### B-1: INTRODUCTION

zontal IVE and assumptions/guidelines employed. The in-field assembly procedure was developed to increase the degree of confidence in the basic IVE design approach by identifying possible problem areas associated with the IVE design. An iterative procedure was used resulting in a compatible design and in-field assembly procedure as discussed in Section B-3 of this appendix.

## B-2. COMMENTS.

The following comments apply to the procedures discussed in Section B-3 of this appendix.

- 1. The X, Y, Z coordinate system used in the IVE structure assembly procedure is the same as the Shuttle Orbiter coordinate system.
- · 2. Special tools/aids required during the assembly of the IVE structure consists of two (2) spreader/hoist bars and two (2) master alignment tools which are contractor furnished.

truse sand

- 3. No special tools are required to be provided by the

  User for the assembly of the structure. However, it
  is assumed that standard facility equipment such as
  an overhead hoist/crane, forklift, optical transit,
  targets, and levels will be available during assembly
  operations.
- 4. The assembly procedure is predicated upon the premise that the structure has been assembled, aligned, verified and disassembled by the Contractor prior to delivery to the User.
- 5. Upon completion of assembly and verification of alignment the nuts on all the bolted connections shall be "sealed" with a material similiar to locktite to prevent inadvertant loosening of bolts.
- o. It is anticipated that at the 100% drawing completion milestone an identification code will be devised and each part of the IVE structure marked. The coding will be incorporated in the in-field assembly procedure to facilitate assembly by the User.

#### B-3. IN-FIELD ASSEMBLY PROCEDURE

- · A. Section 2 Assembly. (See Figure B-1 View B)
  - A.1. Position 4 floor plates on floor to form rectangle approximately 16 feet by 20 feet.
  - A.2. Position 3 cross beams on support blocks between floor plates at 10 foot intervals.
  - A.3. Using overheat hoist, position right hand truss assembly on floor plates. Position end of cross beam between Horizontal attach plates at three places along bottom of truss assembly.
  - A.4. Bolt cross beams to truss assembly. (Do not torque bolts at this time).
  - A:5. Position and bolt right hand knee braces to truss assembly and three cross beams. (Do not torque bolts at this time).
  - A.6. Repeat Steps A.3, A.4 and A.5 for left hand truss assembly.
  - A.7. Detach overhead hoists from right and left hand truss assemblies.
  - A.8. Attach spreader bars to longeron at each end of Section assembly as shown in Figure B-1 View C.
  - A.9. Position diagonal tie rods below cross beams and attach to left and right truss assemblies.
  - A.10. Level bridge rails with floor screw jacks in the X and Y planes (Use facility optical equipment).
  - A.11. Square section assembly by adjusting tie rods.
  - A.12. Plumb truss assemblies in the vertical plane by adjusting knee braces.
  - A.13. Remove spreader bars from section assembly.

- A.14. Position master alignment tool at each end of section assembly as shown in Figure B-1 View D and check alignment of shear pin holes in rails.

   If no fit, repeat steps A.10, A.11 and A.12 as
  - required.
- A.15. Install tooling dowel pins as provided for at the bolted connections.
- A.16. Torque all bolts to specified values.
- $A_{\circ}17_{\circ}$  Position and attach two keel beams at  $Y_{\circ}$  centerline.
- A.18. Position and attach keel longerons to keel and cross beams. Check location using master alignment tool.
- A.19. Install tooling dowel pins in keel beams and \_\_\_\_longerons.
- A.20. Torque keel bolts to specified values.
- A.21. Remove master alignment tools from section assembly.
- A.22. Verify final alignment. (Assembly of Section 2 is \_\_\_\_ complete).
  - Section 3 Assembly. (See Figure B-1 View E)
  - Bil. Position 2 floor plates on floor 20 feet aft of and in line with Section 2 assembly.
- B.2. Position 3 cross beams on support blocks between floor plates at 10 foot intervals. (The 2 floor plates at Station X<sub>0</sub>1063.3 are shared by Sections 2 and 3).
- B.3. Using overhead hoist, position right hand truss assembly so upper longeron is in line with splice plate on Section 2 longeron and insert one bolt.

- B.3.a Adjust screw jacks to align both upper and lower splice plates between sections 2 and 3 and install remaining bolts. (Do not torque bolts at this time).
- B.4. Position end of cross beam between horizontal attach plates at three places along bottom of truss assembly and bolt cross beams to truss. (Do not torque bolts at this time).
- B.5. Position and bolt right hand knee braces to truss assembly and three cross beams. (Do not torque bolts at this time).
- B.6. Repeat steps B.3., B.3.a, B.4. and B.5 for left hand truss assembly.
- B.7. Detach overhead hoist from left and right truss assemblies.
- B.8. Attach spreader bar to longerons at aft end of Section 3 as shown in Figure B-1 View F.
- B.9. Repeat Steps A.9 through A.13 for Section 3.
- B.10. Position master alignment tool to span joint between Section 2 and 3 as shown in Figure B-View A. Position second alignment tool at af end of Section 3.
- B.11. Check alignment of shear pin holes in rails a each tool. If no fit, repeat steps A.10, A.1 and A.12 as required.
- B.12. Install tooling dowel pins as provided for at the bolted connections.
- B.13. Torque all bolts to specified values.
- B.14. Position and attach two keel beams to the cross beams at Y<sub>O</sub> centerline (Figure B-2 View A).
- B.15. Attach Section 3 keel longerons to keel beams and check alignment of keel fitting attach holes with master alignment tool across Section joint at Station  $X_0$  1063.3.

- B.16. Install tooling dowel pins in keel beams a longerons.
- B.17. Torque keel bolts to specified values.
- B.18. Remove master alignment tools from Section assembly.
- B.19. Verify alignment of rails and keel longerons fo Section 2 and 3.
- C. Section 1 Assembly. (See Figure B-2 View B)
  - C.1 Position 2 floor plates on floor 20 feet forward of and in line with Section 2 assembly.
  - C.2 Position 3 cross beams on support blocks between floor plates at 10 foot intervals. (The 2 floor plates at Station  $X_0819.66$  are shared by Sections 1 and 2).
  - C.3 Using overhead hoist, position right hand truss assembly so upper longeron is in line with splice plate at forward end of Section 2 and insert one bolt.
  - C.4 Adjust screw jacks to align both upper and lower splice plates between Sections 1 and 2 and install remaining bolts. (Do not torque bolts at this time)
  - C.5 Position end of cross beam between horizontal attach plates at three places along bottom of truss assembly and bolt cross beams to truss. (Do not torque bolts at this time).
  - C.6. Position and bolt right hand knee braces to truss assembly and three cross beams. (Do not torque bolts at this time).
  - C.7. Repeat Steps C.3, C.4, C.5 and C.6 for left hand truss assembly.
  - C.8. Detach overhead hoist from left and right truss assemblies.

- · C.9. Attach spreader bar to longeron at forward end of Section 1.
  - C.10. Repeat Steps A.9 through A.13 for Section 1.
  - C.10.a Remove spreader bar from Section assembly.
  - C.11 Position master alignment tool to span joint between Section 1 and 2 as shown in Figure B-2 View C. Position second alignment tool at forward end of Section 1.
  - C.12. Check alignment of shear pin holes in rails at each tool. If no fit, repeat Steps A.10, A.11 and A.12 as required.
  - C.13. Install tooling dowel pins as provided for at the bolted connections.
- ·C.14. Torque all bolts to specified values.
- C.15. Position and attach two keel beams to cross beams at  $Y_{\rm O}$  centerline.
- C.16. Attach Section 1 keel longerons to keel beams and check alignment of keel fitting attach holes with master alignment tool across Section joint at Station X<sub>O</sub>819.66.
- C.17. Install tooling dowel pins in keel beams and longerons.
- C.18. Torque all bolts to specified values.
- C.19. Remove master alignment tools from Section 1 , assembly.
- C.20. Verify alignment of rails and keel longerons

  Sections 1, 2 and 3 using transit and targets

  exed to pin holes. The assembly of Sections

  and 3 is complete.



- MS/PS/OOS Support Structure. (See Figure B-2 View D)
- D.1 Position left hand welded support assembly at Station X<sub>0</sub>576 and Y<sub>0</sub>=100 to index with upper and lower gusset attach plates on forward end of Section 1.
- D.2-1. Position 3 floor plates under screw jacks at connectorners of support assembly. (The floor plate at the forward corner of Section 1 is shared Times with the support assembly).
- D.3-10 Level and align L.H. support assembly with screw struct jacks.
- D.4 Bolt gusset plates and vertical member of suppor assembly together at X<sub>0</sub>576.
- D.5 Install tooling dowel pins in gusset plates.
- D.6 Repeat Steps D.1 through D.5 for right hand Staring support assembly.
  - side antet.
- D.6.a Install floor beams connecting LH & RH support assemblies.
- $D_{\rm b} T_{\rm col}$  Install center floor panel between support assemblies at  $Z_{\rm O}$ =419.
- D.8. Install lower support assembly braces connecting of all LH and RH support assemblies.
- D.9 Bolt handrails to floor panels on the left and right support assemblies.
- X 576 Bulkhead Assembly. (See Figure B-2 View D)
- E.1 Using overhead hoist, position bulkhead assembly on top of MS/PS support assembly with aft face of bulkhead at station X<sub>0</sub>=576.
- E.2 Bolt lower attach angle of bulkhead to horizontal member of support assembly. (Do not torque bolts at this time).

- E.3 Position and bolt 2 diagonal braces to bulkhead and floor structure. (Do not torque bolts at this time).
- E.4 Center bulkhead on  $Y_0$ =0 scribe line and align aft surface of bulkhead in the vertical plane by adjusting diagonal braces.
- E.5 Install tooling dowel pins as provided for in bolted connections.
- E.5a Torque bolts to specified values.
- E.6 Attach handrails to outboard edge of bulkhead structure.

### F. X<sub>0</sub>576 Bulkhead Assembly. (See Figure B-2 View E)

- F.1 Assemble 3 panels of bulkhead together in a horizontal position.
- F.2 Using overhead hoist, position bulkhead assemble at Station  $X_01307$  to mate with support angle (L and R side) attached to aft end of Section 3 assembly and bolt together. (Do not torque bolts at this time).
- F.3 Center bulkhead on  $Y_0=0$  scribe line and align forward surface of bulkhead in a vertical plane at Station  $X_0=1307$ .
- F.4 Bolt lower half of bulkhead to attach structure on aft end on Section 3 assembly. (Do not torque bolts at this time).
- F.5 Install tooling dowel pins as provided for at the bolted connections.
- F.6 Torque all bolts to specified values.

### G. Payload Wire Tray Assembly

G.l Install wire tray attach clips to the knee braces on left and right sides of Section 1, 2 and 3 assemblies. Index to scribe lines on knee brace.

- G.2 Position master alignment tool at each end of
  Section 1 assembly engaging rails and keel fitting.
  Attach wire tray index template to alignment tool.
- G.3 Position 20 foot sections of wire tray assemblies between attach clips on knee braces on left and right side of Sections 1, 2 and 3.
- end of Section 1 and bolt in place. Also bolt tray to center knee brace.
- G.5 Repeat Steps G.2, G.3 and G.4 for Sections 2 and 3.

### H. Payload I/F Elements

Install payload I/F elements as required. Include MS, PS, OOS enclosures, mid-body payload I/F's and optional equipment.



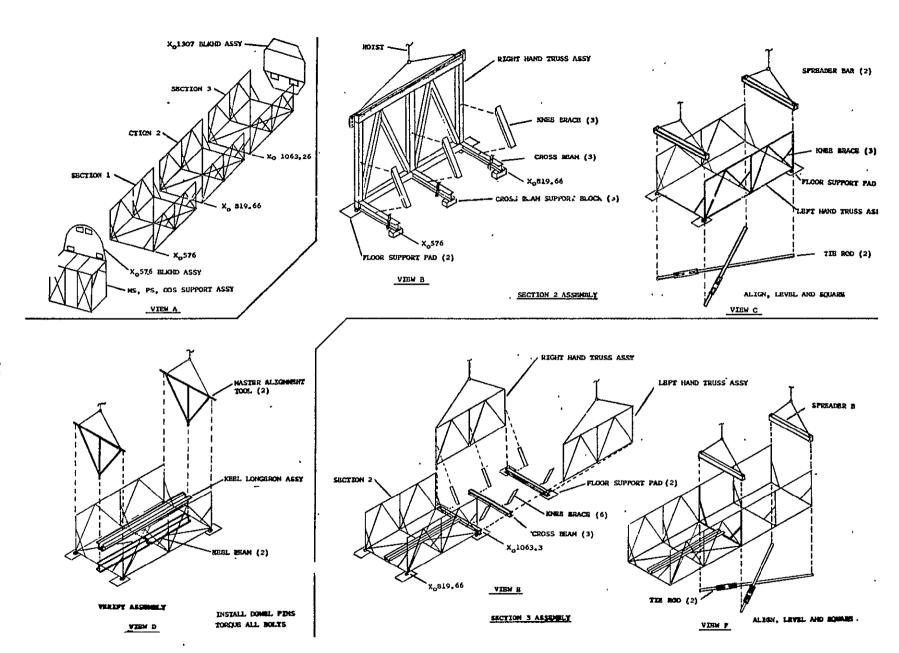


FIGURE B-1 HORIZONTAL IVE IN-FIELD ASSEMBLY PROCEDURE

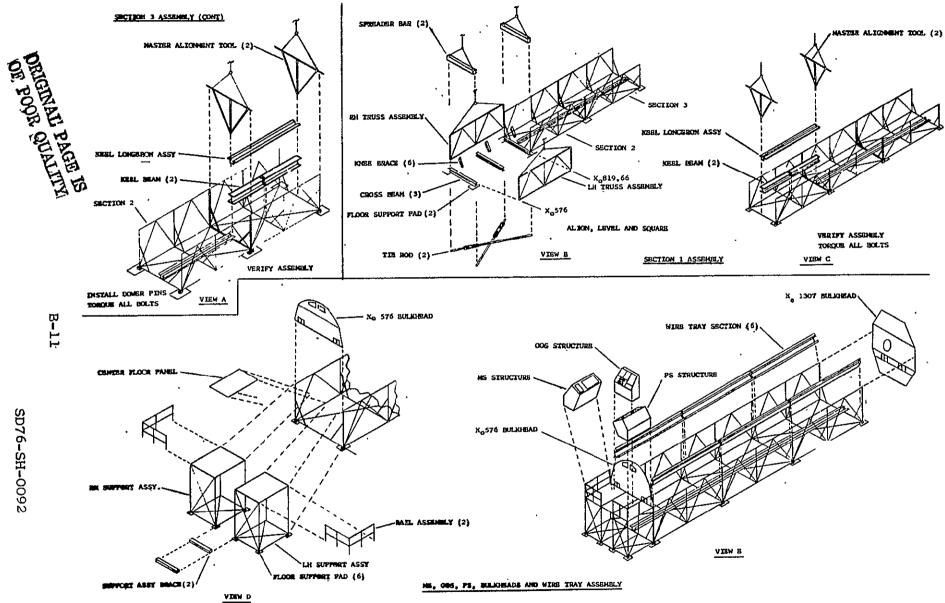


FIGURE B-2 HORIZONTAL IVE IN-FIELD ASSEMBLY PROCEDURE





APPENDIX C

PAYLOAD INTEGRATION BASELINE FUNCTIONAL FLOW BLOCK DIAGRAMS AND OPTIONS

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### APPENDIX C

PAYLOAD INTEGRATION BASELINE FUNCTIONAL FLOW BLOCK DIAGRAMS AND OPTIONS

### C.1 CONTENTS

This appendix contains the detailed descriptions or the payroad integration functional flow block diagrams as described in Section 9.0 of the report. A baseline and two optional flows are included for the following five payloads representative of the broad payload spectrum:

- 1. Solar Maximum Mission
- 2. Solar Physics Dedicated Missi
- 3. Module With Pallet (Spacelab)
- 4. Large Space Telescope
- 5. Mariner Jupiter Orbiter/IUS

Contents include terminology used, baseline payload integration functional flow block diagrams, and interface checkout matrices identifying differences to the baseline flows for the two options.

### C.2 TERMINOLOGY

The following terminology and definitions apply to the data presented in the figures and tables in Section 3, of this appendi:

<u>Cargo</u> The total complement of payloads carried on any one flight.

Payload The total complement of specific instruments, space equipment and support hardware carried in the Orbiter to accomplish a certain task in space.

<u>Subsystem</u> The next functional subdivision of a system.

Assembly The next functional subdivision of a component (two or more parts or subassemblies).

Subassembly An assembly within a larger assembly.

<u>Part</u>

A basic element of component, assembly or subassembly.

Acceptance Tests

Acceptance tests are the required formal tests, conducted to demonstrate acceptability of an item for delivery.

Block Numbers in FFBD's

This indicates a sequential number of a block, as shown in the Functional Flow Block Diagram (FFBD) for a given payload processing operation.

Description

Title identification of the blocks in the FFBD's, the functions to be performed, and the operation level (subsystem, system, payload and cargo.)

I/F Knowledge

Four different categories of interface related requirements are reflected. They are as follows:

- (1) Orbiter I/F Not Required This category does not interface with Orbiter:
- (2) Orbiter I/F Knowledge Required This category does not interface
  with Orbiter, but knowledge of
  Payload/Orbiter interface data is
  required.

- (3) Direct Orbiter Function Simulated -In this category some of the Orbiter functions are required to be simulated.
- Direct Orbiter I/F This category requires that the payload interfaces directly with Orbiter.

Baseline Location Operational site location for the function identified as determined from the NASA payload data base.

### Option 1 (User Oriented

Operational site location for the function identified with maximum integration to be accomplished prior to payload delivery to the launch site.

Option 2 aunch site orient Operational site location for the function identified with maximum integration to be accomplished at the launch site.

### Simulators

Identifies integration functions requiring usage of IVE or GSE simulators.

[VĒ

Identifies interface functions, in the interface checkout matrix tables that. Scan be accomplished by standardized IVE.

Special Facility

A special facility (thermo/vac, vibration or other) is required to perform a given function.

### Remarks

Either Site - A given function may be performed at either User or Launch Site. The decision where to perform the function will depend on the results of a trade stud of other relevant functions, showing that the chosen site will not affect the operat However, if the function does not depend on a trade study, it can be performed at the site equipped with the required GSE simulators or IVE, taking into consideration the effects of factors like handling, transportation, risk, availability of personne etc.



### Remarks (Cont)

Both Sites - A given function is to be performed at both sites (User & Launch). Final decision whether the function is to be performed at both sites depends on I/F verification philosophy such as: Whether minimum checkout is performed initially at User site, and maximum checkout at Launch site, or vice verse. Transportation and subsequent checkout and I/F verification requirements will influence the decision.

### Trade Study

A study, required to determine the preferable site to perform a given function. Determination of site wil depend on evaluation of the factors, presented in the Criteria Table, such as risk, cost, schedule impact, availability of personnel and equipment, etc.

# C.3. PAYLOAD INTEGRATION ANALYSIS - FUNCTIONAL FLOW BLOCK DIAGRAMS AND INTERFACE CHECKOUT MATRICES

The FFBD's and I/F checkout matrix tables are included for the five payloads identified above. These data support the discussion in Section 9.0 of this report. Data presented in Figures C-1 through C-5 and Tables C.1 through C.5 reflect the next level definition of the payload integration processes described in Figure 9-2.

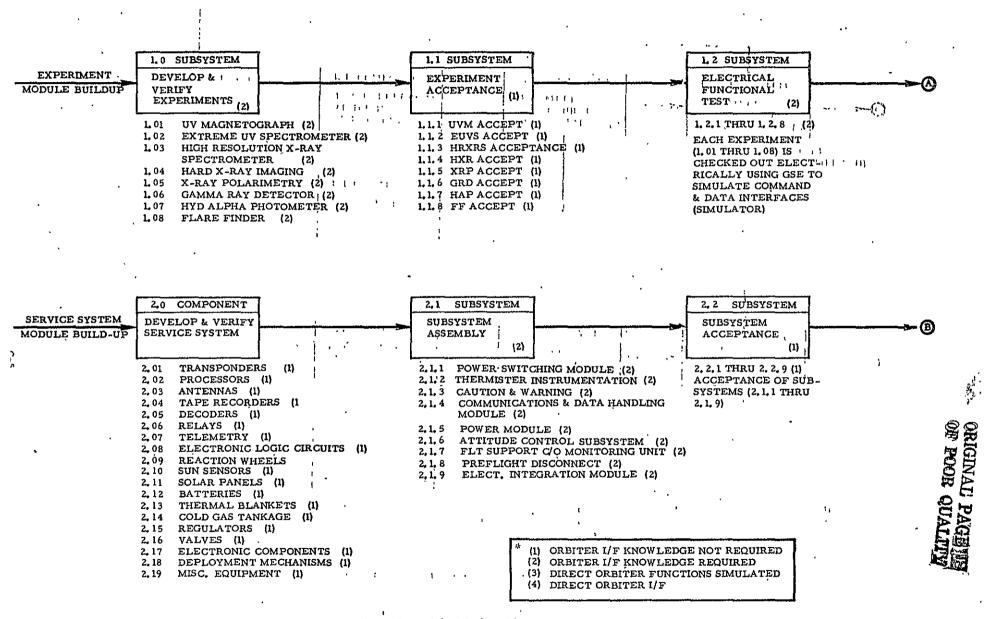


Figure C-1. Solar Maximum Mission FFBD

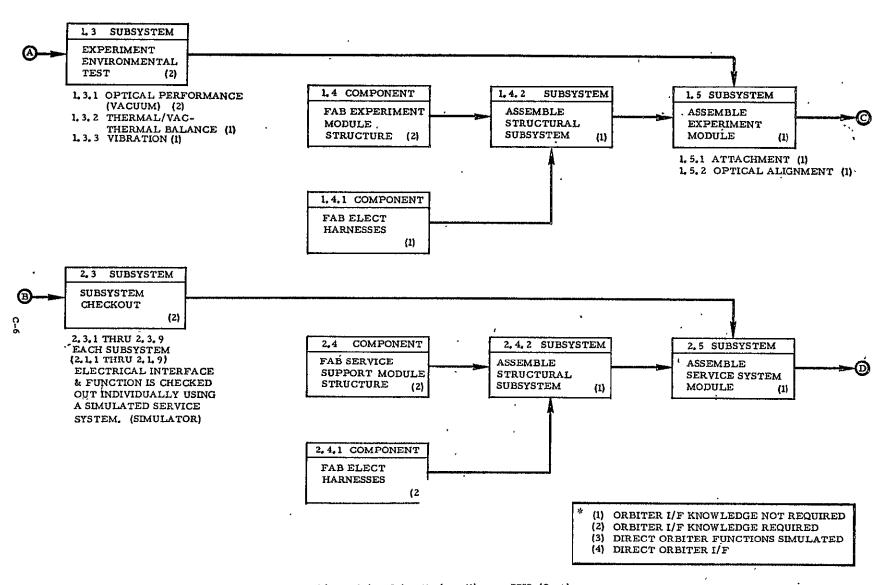


Figure C-1. Solar Maximum Mission FFBD (Cont)

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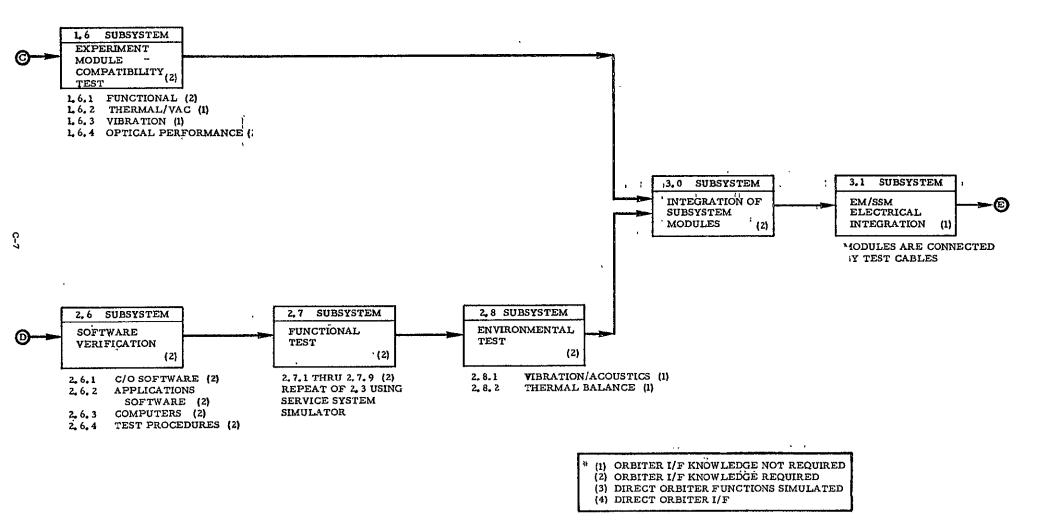


Figure C-1. Solar Maximum Mission FFBD (Cont)

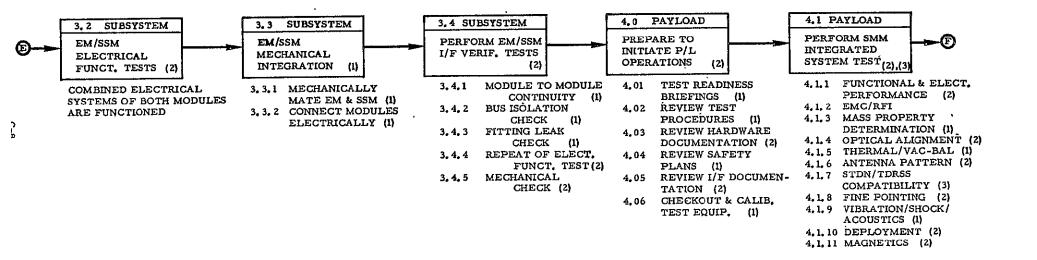


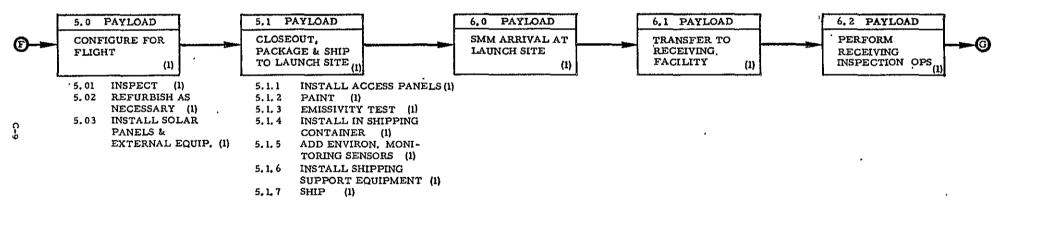
Figure C-1. Solar Maximum Mission FFBD (Cont)

<sup>(1)</sup> ORBITER I/F KNOWLEDGE NOT REQUIRED
(2) ORBITER I/F KNOWLEDGE REQUIRED

<sup>(3)</sup> DIRECT ORBITER FUNCTIONS SIMULATED

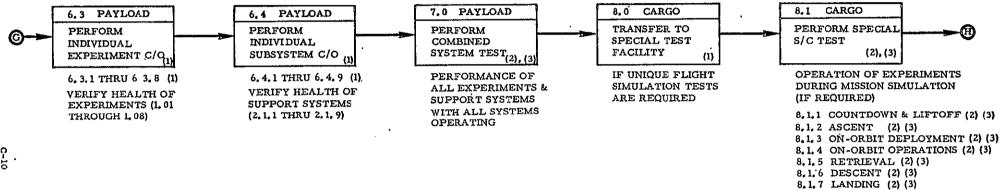
<sup>(4)</sup> DIRECT ORBITER I/F





- \* (1) ORBITER I/F KNOWLEDGE NOT REQUIRED (2) ORBITER I/F KNOWLEDGE REQUIRED
- (3) DIRECT ORBITER FUNCTIONS SIMULATED
- (4) DIRECT ORBITER I/F

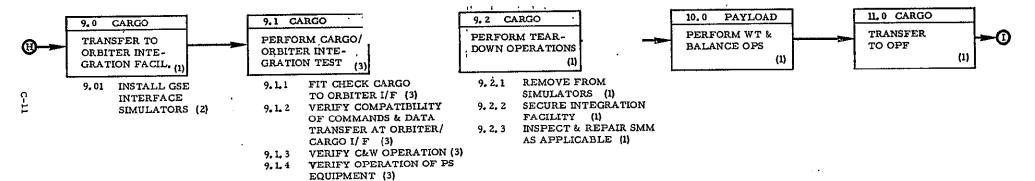
Figure C-1. Solar Maximum Mission FFBD (Cont)



- \* (1) ORBITER I/F KNOWLEDGE NOT REQUIRED
- (2) ORBITER I/F KNOWLEDGE REQUIRED
- (3) DIRECT ORBITER FUNCTIONS SIMULATED
- (4) DIRECT ORBITER I/F

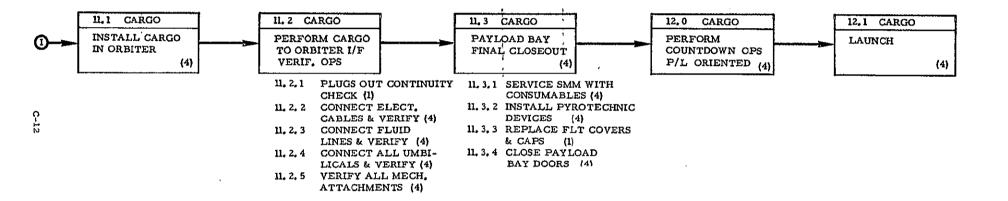
Figure C-1. Solar Maximum Mission FFBD (Cont)





- \* (I) ORBITER I/F KNOWLEDGE NOT REQUIRED
- (2) ORBITER I/F KNOWLEDGE REQUIRED
- (3) DIRECT ORBITER FUNCTIONS SIMULATED
- (4) DIRECT ORBITER I/F

Figure C-1. Solar Maximum Mission FFBD (Cont)



- \* (1) ORBITER I/F KNOWLEDGE NOT REQUIRED
- (2) ORBITER I/F KNOWLEDGE REQUIRED
- (3) DIRECT ORBITER FUNCTIONS SIMULATED
- (4) DIRECT ORBITER I/F

- Figure C-1. Solar Maximum Mission FFBD (Cont)

	ORIGINALI PAGE IS OF POOR QUALITY	Table	C.1 SMM ]	DITERFACE CH	rckout matrix			<b></b>		
	1 20		TIT	REPACES		CHFKCOM FQUIPM	ent			
BLOCK NO.	PESCRIPTION	* I/F KNOWLEDGE	Baseline Location	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITE ORIENTED	GSR CIV.	IVE	SPECIAL FACILITY		РЕМАРКО
1.0	Develor & verify experiments	(5)	User	X					Various	sources
1.01	UV magnetograph	(2)		х						
1.02	Extreme UV spectromater	(5)		x				<u> </u>		
1.03	High resolution X-Ray spectrometer	(2)		x						
1.04	Pard X-Ray imaging	(2)		x						
1.05	YRay polarimetry	(8)		х						
1.76	Gapma ray detector	(5)		χ						
3.07	Nydrosen A'pha photometer	(n)		х						<u> </u>
1.08	Flare finder	(8)		х						V
1.1	Fyperiment acceptance	(1)		x						
1.1.1	UV magnetograph acceptance	(1)		у						
1.1.2	Extreme UV spectrometer acceptance	(1)		χ						
1.1.3	HRXRS acceptance	(1)		x						
1.1.4	Hard X-Ray imaging acceptance	(1)		Х						
1.1.5	X-Ray polarimetry acceptance	(1)		Х						
1.1.6	Camma ray detector acceptance	(1)	,	х						
1.1.7	Hydrogen Alpha photometer acceptance	(1)		у			ļ	<u> </u>	<b> </b>	
1.1.8	Flare finder acceptance	(1)	+	Y		_,	<u> </u>	<u> </u>	<u> </u>	
	* (1) OBBITED I/E HOT DESCRIBED.		(	Pinicial C	ORTITER OFFITER I/F					

Table C.1 (Cont) SHM INTERFACE CHECKOUT MATRIX

		T	nti	REACES		CHPKCO EQUIPM	t/test ent		
BI.OCK	desésialica	* I/P KNOWLEIGE	BASELINE LOCATION	OPTION I USER ORIENTED	OPTION 2 LAUNCH SITF ORIENTED	G9# OM.	IVE	SPECIAL FACILITY	RPMARECS
1.2	Electrical functional tests	(2)	User	х		х			
1.2.1	UVM electrical functional tests	(2)		x		x			
1.2.2	EUVS electrical functional tests	(2)		x		х			
1.2.3	HEXES electrical functional tests	(2)		х		х			
1.2.4	HXRI electrical functional tests	(2)		x		х			
1.2.5	XRP electrical functional tests	(2)		х		х.		•	
1.2.6	GRD electrical functional tests	(2)		x		· x			
1.2.7	HAP electrical functional tests	(2)		x		x			•
1.2.8	FF electrical functional tests	(2)		x		X			
1.3	Experiment environmental tests	(2)		×					
1.3.1	Optical performance (vacuum)	(2)		x				x	
1.3.2	Thermal/Vac-thermal balance	(1)		x				х	· ·
1.3.3	Vibration	(1)		x				х	
1.4	Fabricate experiment module structure	(2)		x					
1.4.1	Fabricate electrical harnesses	(1)		x		x			ORB/P/L I/F harnesses included in SSM harnesses
1.4.2	Assemble structural subsystem	(1)		x					
1.5	Assemble experiment module	(1)		x					
1.5.1	Attachment	(1)		. х					
1.5.2	Optical alignement	(1)	<b>†</b>	x			,		
- Transfer or a control of the contr	ORBITEP I/F HOT REQUIRED			FUNCTIO	ORBITER ORBITER ORBITER				•
	S OBSILED I'L KJOATEDCE BEGINEED		.(	DIRECT	OPPITER I/F				

ITTERPACES.

## Table C.1 (Cont) SMM INTERFACE CHECKOUT MATRIX

	, ·		TrTI	RFACES		CHPKCOU EQUIPM	T/TEST Ent		
BLOCK NO.	, PESCRIPTION	* 1/F KNOWLEDGE	BASELINE LOCATION	OPTION 1 USBR ORIENTED	OPTION 2 LAUNCH SITH ORIENTED	GSW	IVE	SPECIAL PACILITY	REMARKS
2.13	Thermal blankets	(1)	User	х		,			Various sources
2.14	Cold gas tankage	(1)		x				<u> </u>	
2.15	Regulators	(1)		х				<u> </u>	
2.16	Valves	(1)		x					
2.17	Electronic components	^(1)		x					
2.18	Deployment mechanisms	(1)		· x					
2.19	Miscellaneous equipment	(1)		х					<b>†</b>
2.1	Subsystem assembly	(5)		х		,			
2.1.1	Power switching module	(5)		х					
2.1.2	Thermistor instrumentation	(2)		х	•				
2.1.3	Caution & warning	(2)		х					
2.1.4	Communications & data handling module	(2) ·		х					
2.1.5	Power module	(5),		х					
2.1.6	Attitude control subsystem	(5)		х					
2.1.7	Flight support c/o monitoring unit	(2)		х					
2.1.8	Preflight disconnect	(2)		x.					
2.1.9	Electrical integration module	(2)		x		<u> </u>	<u> </u>	<u> </u>	
				`				<u> </u>	
				A III Kibirah	ORSITER		<u> </u>	<u> </u>	<u> </u>
	ORBITEP I/F NOT REQUIRED		(	FUNCTI	ON" CIMULATED				
	OPBITEP I/F KIOJLEDGE REQUIPED			DIRTCT	OPPITER I/F				

Table C.1 (Cont) SMM INTEREACE CHECKOUT MATRIX

		i	1	PTERFACES	!	CHFKCOU EQUIPM	r/test ent:	į	
BI.OCK	Precription	# I/F RNOWLEDGE	BASELIN · LOCATI	OPTION 1 E USER ON ORIENTED	OPTION 2 LAUNCH SITF ORTENTED	GSF.	IVE	SPECIAL PACILITY	<b>РИМА</b> РИЗ.
.2	Subsystem acceptance	(1)	User	x	1				
.2.1	Power switching module acceptance	(1)	,	х	'				1
.2.2	Thermistor instrumentation acceptance	(1)	<u> </u>	х					
2.2.3	Caution & warning acceptance	(1)		х			<u> </u>	<u> </u>	
2.2.4	C&DH module acceptance	(1)		х					í
2.2.5	Power Module acceptance	(1)		х	,		·	'	
2.2.6	Attitude control subsystem acceptance	(1)	•	х	,				
2.2.7	Flt support c/o monitoring unit acceptance	(1)		х					
2.2.8	Preflight disconnect acceptance	(1)		х			<u> </u>	<u> </u>	
2.2.9	Electrical integration module acceptance	(1)		х					
2.3	Subsystem checkout	(2)		х	,	х			
2.3.1	Power switching module checkout	(5)		x		х			
2.3.2	Thermistor instrumentation checkout	(2)		х		х		<u> </u>	
2.3.3	Cs ution & warning checkout	(2)		х		х			
2.3.4	C&DH module checkout	(5)		х		X			
2.3.5	Power module checkout	(2)		x		х			
2.3.6	Attitude control subsystem checkout	(2)		х		, x		<u> </u>	
2.3.7	Flt support c/o monitoring unit checkout	(2)		х	<u> </u>	х		<b></b>	3
2.3.8	Preflight disconnect checkout	(5)		х		×			
	ORBITEP I/F NOT REQUIRED			(3) FUNCTO	CONT FIMULATE	)			•
	OBSITED I/E KIOATEROE ESDATEED		•	(I) DIRECT	r oppiter i/f				

Table C.1 (Cont) SHM INTERFACE CHECKOUT MATRIX

	•		INT	BRFACES		CHFKCOU! EQUIPM			
BLOCK	PRECRIPTION	* I/F	Baseline Location	OPTION 1 USER ORIENTED	OPTION 2 Launch Site Oriented	GSR GM.	ive	SPFCIAL FACILITY	REMAPES
2.3.9.	Electrical integration module checkout	(2)	User '	^X		x			
2.4	Fabricate SSM structure	(2)		х				<u> </u>	,
2.4.1	Fabricate electrical harnesses	(2)	,	х		x	,	<u> </u>	ORB/P/L I/F knoledge for I/F Harnesses
2.4.2	Assemble structural subsystem -	(1)	3	х					
2.5	Assemble service system module	(1)		х					,
2.6	Software verification	(5)	*	х		x	-		Can be accomplished at remote loca
2.6.1	Checkout software	(5)	1	х					
2.6.2	Applications software	(2)	í	x				<u> </u>	· ·
2.6.3	Computers and processors	(2)	1	x				<u> </u>	
2.6.4	Test procedures	(5)		·x				ļ	<b>*</b>
2.7	SSM functional test	(2)	<u>'</u>	x	х	χ	х		To be accomplished at both sites
2.7.1	Power switching module checkout	(2)		х	x	х	х		
2.7.2	Thermistor instrumentation checkout	(2)	!	x	х'.	х	х		•
2.7.3	Caution & warning checkout	(5)	, ,	x	х	x	х		
2.7.4	C&DH module checkout	(Ş) · .	. !   ;	x	х	x	х		
2.7.5	Power module checkout	(2)	, !	, x	х	X	х	<b></b>	
2.7.6	Attitude control system checkout	(2).		х	х	х	<u>x</u>	<u> </u>	
2.7.7	Flt support e/o monitoring unit checkout	(5)		х	x	х	х		
2.7.8	Preflight disconnect checkout	(5)	<b>,</b>	Х	X ORSITER	x	x	<u> </u>	<b>*</b>
	ORBITEP I/F NOT REQUIRED		(	3 FUNCTI	ONG CIMULATED				•
à	OPBITED I/F KNOWLEDGE REQUIEND		(	TOFRIC (F	OPPITER I/F				

Table C.1 (Cont) SMM INTERFACE CHECKOUT MATRIX

			Tri	ng aces		CHPKCON EQUIPM	t/test ent			
BLOCK NO.	PESCRIPTION	* I/F NOWLEDGE	BASELINE LOCATION	OPTION 1 'USER ORIENTED	OPTION 2 LAUNCH SITF ORTENTED	G97 TIM.	IVE	SPECIAL FACILITY	PRMATECT	
2.7.9	Electrical integration module checkout	(2)	User	х	х	х	х		At either or both sites	
2.8	Environmental test	(1)		х		x				
2.8.1	Vibration/scoustics	(1)		x		х		x		
2.8.2	Thermal balance ·	(1)		x		х .		x	,	
3.0	Integration of subsystem modules	(5)		x ·	х	х	х		Trade study	
3.1	EM/SSM electrical integration	(1)		x					Connect cables-no power	
3.2	EM/SSM electrical functional tests	(2)		х	х	x	x		Trade study	
3.3	EM/SSM mechanical integration	(1)		хх			· · ·	.:		
3-3-1	Mechanically mate EM & SSM	(1)		х	x	х	×		Trade study	
3.3.2	Connect modules electrically	(1)		х	х	х	X		<u> </u>	
3.4	Perform EM/SSM interface verification tests	(2)		х	х	X	х			
3.4.1	Module to module continuity	(1)		x	х	x	х			
3.4.2	Bus isolation check	(1)		х	х	_х	х			
3.4.3	Fitting leak check	(1)		x	x	x	x			
3-4-4	Repeat of electrical functional tests	(2)		х	x	x	x		At either or both sites	
3.4.5	Nechanical fit check	(2)	*	· x	х	х	х			
				3) -DIRFOR	ORSITEK		1			
	OBBITED I/E MOLHBONIED  OBBITED I/E MOLHBONIED		(	9 FUNCTI	ON" SIMULATED	1				

Table C.1 (Cont) SMM INTERFACE CHECKOUT MATRIX

			TIT	BRO'ACIES		CHPKCOU EQUIPM	t/test Ent		
BLÒCK NO.	PESCRIPTION	* 1/F RHOVLETGE	BASELINE LOCATION	OPTION 1 USBR ORIENTED	OPTION 2 LAUNCH SITF ORTENTED	GSR GIM.	IVE	SPPCIAL PACILITY	RPMAPEG
<b>4.</b> 0	Prepare to initiate payload operations	(5)	User	х	X				
4.01	Test readiness briefings	(1)		х	х				`
4.02	Review test procedures	(1)		х	х		<u> </u>	·	Both sites
4.03	Review hardware documentation	(2)		х	х				
4 Ok	Review safety plans	(1)		x	x				
4.05	Review interface documentation	(2)		x	X				
4.06	Checkout & calibrate test equipment	(1)		х	х	·x	х .	·	
4.1	Ferform SMM integrated system tests	(1), (2),(3)		х	х	х	х		
4.1.1	Functional & electrical performance	(2)		х	х	x	х		<b>*</b>
4.1.2	ENC/RFI tests	(5)		х	x				Monitor EMI/RFI during funct. test
4.1.3	Mass property determination	(1)		x				х	Special facility
4.1.4	Optical alignement	(2)		х	х	x	х		Both sites.
4.1.5	Thermal/wac balance	(1)		х .		x		х	Special facility required
4.1.6	Antenna pattern	(2)		х	x	x	x		Both sites
4.1.7	STDN/TDRSS compatibility	(3)		х	х				Direct link communications
4.1.8	Fine pointing	(5)		x	x	x	х		Both sites
4.1.9	Vibration/shock/acoustics	(1)		х	χ	х		x	Special facility required .
4.1.10	Deployment	(2)		х	х	х	х		
4.1.11	Hagnetics	(2)	*	х ;		х			
	* (1) ORBITEP T/F NOT REQUIRED			3) DIRPOR	ORSITER ON CIMULATED				•
	OPSITEF I/F KNOWLENGE REQUIFED		(	TOFRIC C	OPPITER I/F		,		

Table C.1 (Cont) SHN INTERFACE CHECKOUT MATRIX

<del>,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,</del>			n.r.	RPACES		CHFKCOU EQUIPM	r/test ent		
BLOCK	PRSCRIPTION	* 1/F KNOWLEDGE	Baseline Location	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITF ORIENTED	gs# nim.	IVE	SPECIAL FACILITY	REMARKS
6.3.3	High resolution X-Ray spectrometer checkout	(1)	Launch Site		x	X	x		Post shipping requirement
6.3.4	Hard X-Ray imaging checkout	(1)			x	x	х		
6.3.5	X-Ray polarimetry checkout	(1)			х	X	х		
6.3.6	Gemma ray detector checkout	(1)			х	x	х		
6.3.7	Hydrogen Alpha photometer checkout	(1)			х	x			
6.3.8	Flare finder checkout	(1)	<b> </b>  _		х	х	х		
6.4	Perform individual subsystem checkout	(1)		<u> </u>	х	х	х		
6.4.1	Power switching module checkout	(1)		<u> </u>	х	х	х		
6.4.2	Thermistor instrumentation checkout	(1)		<u> </u>	х	х	х		
6.4.3	Caution & warning module checkout	(1)			x	x	x		
6.4.4	C&DH module checkout	(1)			х	х	х		
6.4.5	Power module checkout	(1)			х	х	х		
6.4.6	Attitude control system checkout	(1)			χ	х	х		
6.4.7	Flt support c/o monitoring unit checkout	(1)			х	х	х		
6.4.8	Preflight disconnect checkout	(1)			x	х	x		
6.4.9	Electrical integration module checkout	(1)			х	х	х		<u> </u>
7.0	Perform combined system test	(2), (3)		х	х	х	х		7.0,8.1,&9.0 tests could be combin & tests conducted in IVE. Need trade study.
	*(1) ORRITEP I/F NOT REQUIRED	1	1 (	DIRPOR FLUCTI	ONSITER ON CIMULATED				
	OPBITEF I/F KIGVLERGE REDUTFER		. (	DIRECT	OPPITER I/F		,		

DRIGINAL PACE IN STALLING

### Table C.1 (Cont) SME INTERFACE CHECKOUT MATRIX

			र ॄ मत	BRPACES	(	CHFKCOU!	r/Test ent		
BI <sub>r</sub> ock No.	PRSCRIPTION	* I/P	BASELINE LOCATION	OPTION 1 USBR ORIENTED	OPTION 2 LAUNCH SITF ORIENTED	GSP GIM.	ive	SPECIAL PACILITY	RPMARKT
8.0	Transfer to special test facility	(1)	Launch Site		. x	, x	x	:	7.0,8.14 9.0 tests could be combined tests conducted in IVE. Reed
8.1	Perform special S/C test	(2), (3)			x	х	χ	·	trade study
8.1.1	Countdown and liftoff simulation	(2), (3)	•		х	х	ж		
8.1.2	Ascent simulation	(2), (3)		,	' x	х	, х	-	
8.1.3	On orbit deployment simulation	(2), (3)			х	x	, x		
8.1.4	On orbit operations simulation	(2), (3)		<u>'</u>	x	х	X		
8.1\5	Retrieval simulation	(2), (3)			x	х	x		
8.1.6	Descent simulation	(2), (3)			x	х	х		·
8.1.7	Inding simulation	(2), (3)			, x	x	х	·	
					<u>                                     </u>				
9.0	Transfer to orbiter integration facility.	(1)			х				O&C (MSOB)
9.01	Install GSE interface simulators	(5)		х	, x	х	х		Orbiter simulator required
9.1	Perform cargo/orbiter integration test	(3)		x	х	х	х		7.0,8.1,29.0 tests could be combine & tests conducted in IVE. Meed trade study for cost effectiveness
9.1.1	Fit check cargo to orbiter interface	(3)		х	х	х	х		blade Blady 101 Coast Circolliness
9.1.2	Verify compatibility of command & data X-fer	(3)		x	х	х	х	<u> </u>	
	at orbiter/cargo interface	<u> </u>							
9.1.3	Verify caution & warning operation	(3)		х	х	х	х		User site could be used if Orbiter
9.1.4	Verify operation of PS equipment	(3)	<b>*</b>	х	х	х	х	,	Simulator is available
	• (1) ORBITEP T/F NOT REQUIRED			3) DIRPOR FUNCTI	ORSITER ON CIMULATE	<u> </u>	<u> </u>	<u> </u>	L
	OPPITED I/F KNOVLEDSF REQUIEDD	6	(	TOFRIG C	OPPITER I/F				

Table C.1 (Cont) SHM INTERFACE CHECKOUT MATRIX

BI-OCK	PESCRIPTION			Irri	ireaces		CHPKCOUT/TEST EQUIPMENT		SPPCIAL PACILITY	REMADECI
		* I/F	Baseline Location		OPTION 1 USBR ORIENTED	OPTION 2 LAUNCH SITF ORIENTED	car Gar	IVE		
9.2	Perform tear down operations	(1)	Launc	h Site	х	x	х	х		User site could be used if orbiter simulator is available
9.2.1	Remove from simulators	(1)			х	х	x	х		
9.2.2	Secure integration facility	(1)			¹ x	x	x	ж		·
9.2.3	Inspect & repair SMM, as applicable	(1)			х	х	х	x		
10.0	Perform weight & balance operations	(1)			х .	х		,	х	Specialized facility required
11.0	Transfer to OFF	(1)				х				
11.1	Install cargo in orbiter	(4)				x				
11.2	Perform cargo to orbiter I/F verification ops.	(4)				х				
11.2.1	Plugs out continuity check	(1)				х				
11.2.2	Connect electrical cables & verify	(4)				х		•	, , ,	,
11.2.3	Connect fluid lines & verify	(4)				х		<u> </u>		
11.2.4	Connect all umbilicals & verify	(4)				х				
11.2.5	Verify all mechanical attachments	<b>(</b> 4)				х				,
11.3	Payload bay final closecut	(4)				х				,
11.3.1	Service SMM with consumeables	(4)			k	x				
11.3.2	Install pyrotsehnic devices	(4)				х				
11.3.3	Replace flight covers & caps .	(4)	<b>*</b>			x				
	* (1) ORBITEP I/F NOT REQUIRED  OPRITER I/F KNOTLEDGE REQUIEDD		в Маррия	<u>۔</u> ان	y fuction	ORBITER ON CIMULATED OPPITER Î/F				,

bi.ock no.	out out	A straight with the straight of the straight o	· 	wegaju njilara ujus manja lang			ing to a summer	•	
	Precription	* I/F KNOWLEDGE	Baseline Location	USER ORIENTED	UFITON 2 LAUNCH SITF ORIENTED	GSW WIM.	IVE	SPECIAL PACILITY	РЕМАРИЗ
11.3.4	Close payload bay doors	<b>(</b> 4)	Launch Site		X		1		
12.0	Countdown operations	(4)			х	:			1
12.1	Launch	(4)			х				
								<b> </b>	
				· · · · · · · · · · · · · · · · · · ·	,				
~.,									
<del>,</del>			•	•					
	* (1) ORBITEP I/F NOT REQUIRED			3 DIRPOR FUNCTION	URSITER ONG CIMULATED		<u> </u>	<u> </u>	
* *. *	CS) OBSILED I'L KJOAFEUCE BECALLED		(		OPPITEP 1/F				

TablesC.1 (Babe)

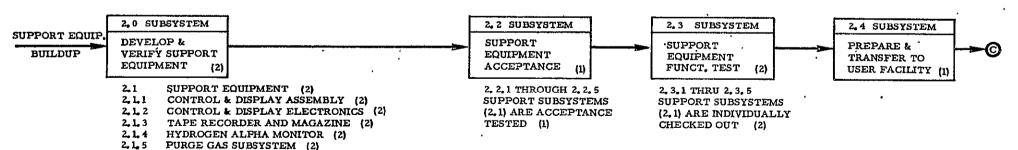
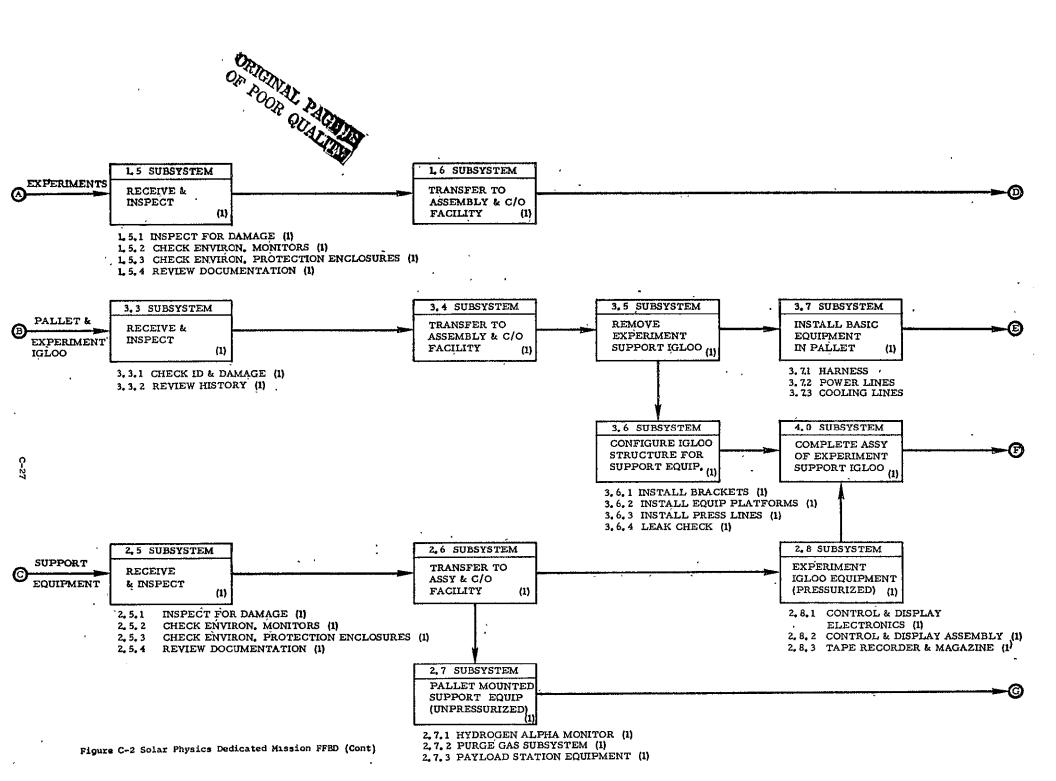


Figure C-2 Solar Physics Dedicated Mission FFBD



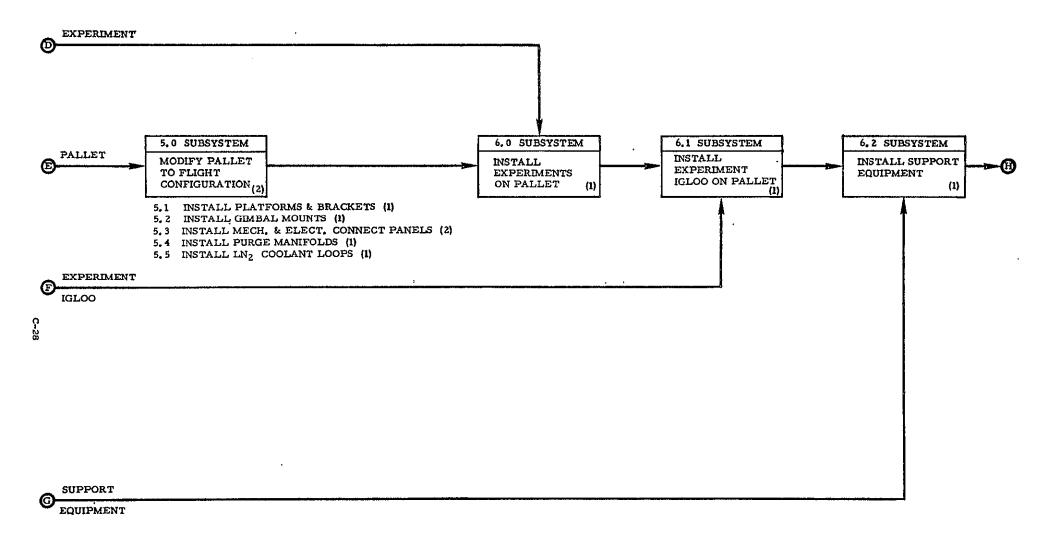


Figure C-2 Solar Physics Dedicated Mission FFBD (Cont)

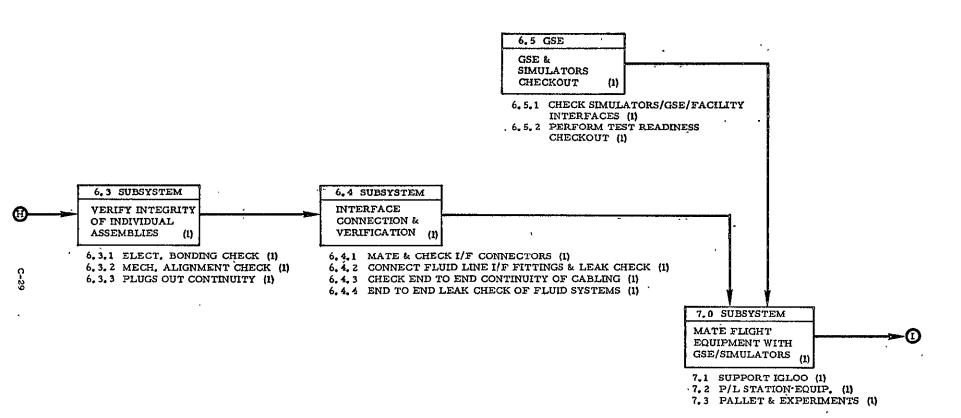
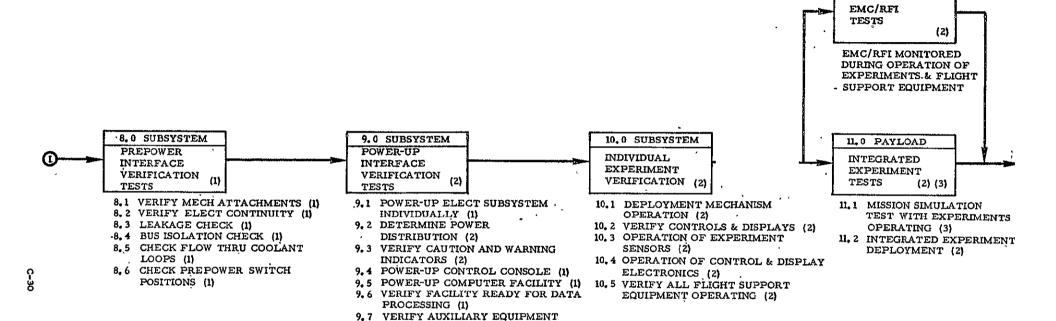


Figure C-2 Solar Physics Dedicated Mission FFBD (Cont)



12,0 PAYLOAD

Figure C-2 Solar Physics Dedicated Mission FFBD (Cont)

OPERATIONS (1)

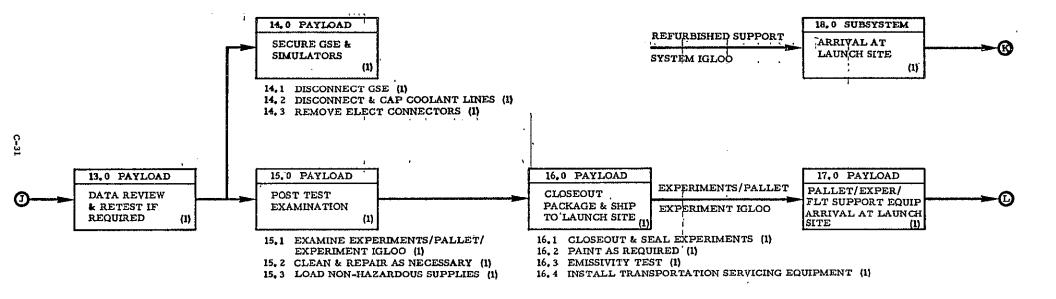


Figure C-2 Solar Physics Dedicated Mission FFBD (Cont)

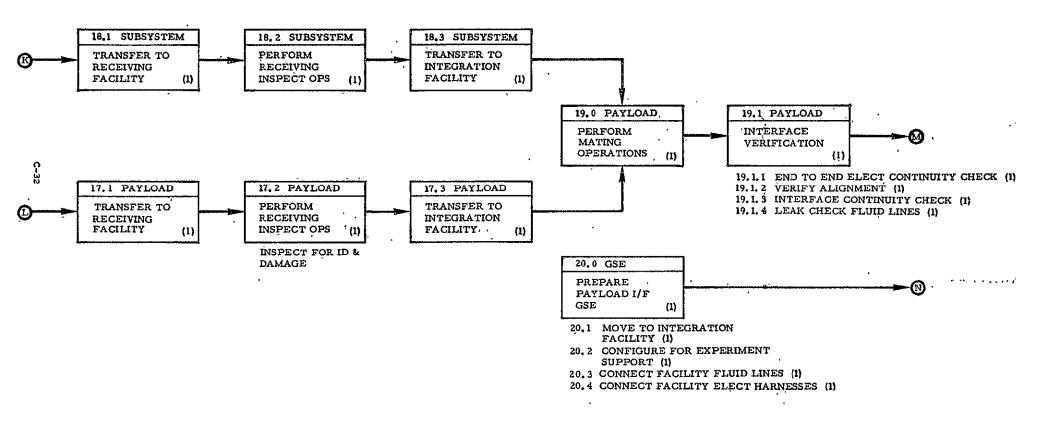


Figure C-2 Solar Physics Dedicated Mission FFBD (Cont)

## ORIGINAL PAGE ES OF POOR QUALITY.

OPERATIONS (2)

SUBSYSTEMS (2)

EQUIPMENT (1)

21. 6

21. 4 VERIFY UMBILICAL INTERFACES

21.5 VERIFY OPERATION OF DATA MANAGEMENT

21.7 VERIFY OPERATION OF OPERATOR CONSOLE (1)

VERIFY OPERATION OF COMPUTER PERIPHERAL

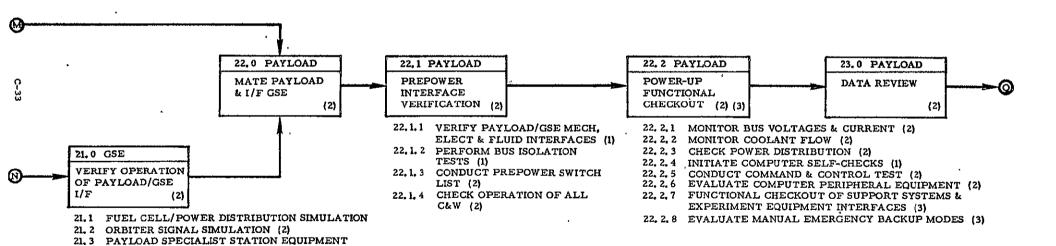


Figure C-2 Solar Physics Dedicated Mission FFBD (Cont)

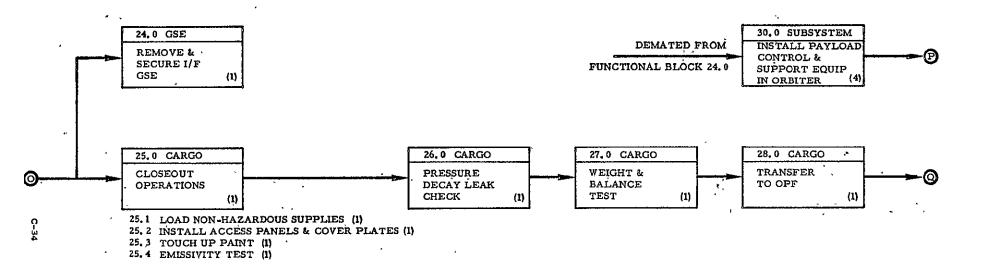


Figure C-2 Solar Physics Dedicated Mission FFBD (Cont)

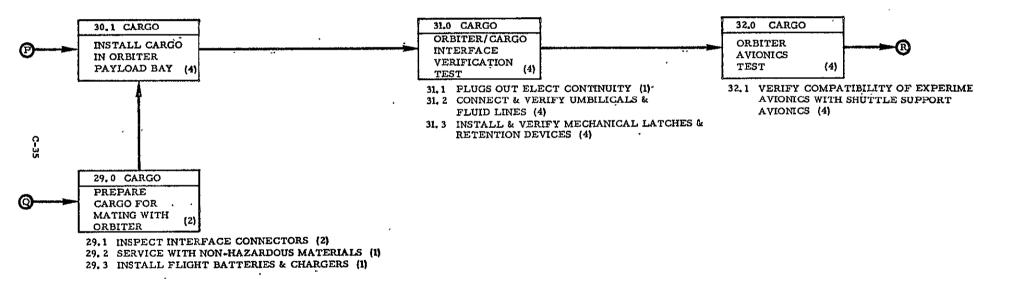


Figure C-2 Solar Physics Dedicated Mission FFBD (Cont)

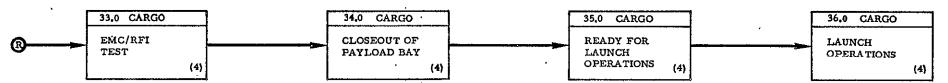


Figure C-2 Solar Physics Dedicated Mission FFBD (Cont)

Table C.2 Solar Physics Dedicated Mession Interface Checkout Matrix

	1		]	THEFFACES		CHFKCOIT EQUIPM	r/test ent	<u> </u>	
BLOCK	Prescription	# I/F	BASELI LOCAT		OPTION 2 LAUNCH SITE ORTENTED	GSB GIM.	IVE	9PFCIAL FACILITY	REMAPES
1.0	Develop & verify experiments	(5)	User	×	,				Various sources
1.1	Experiments	(2)		х					
1.1.1	100 em photoheliograph	(5),		x					
1.1.2	UV spectrograph	(2)		х			`	<u>                                     </u>	
1.1.3	Extreme UV spectro heliometer	(5)		χ				<u> </u>	
1.1.4	Spectrometer/spectro heliograph	(2)		x					
1.1.5	Soft X-Ray telescope/spectrograph	(2)		х					
1.1.6	Eoft X-Ray spectrometer/spectro heliograph	(5)		х				<u> </u>	
1.1.7	Grid collimator acquisition photometer	(5)		х					
1.1.8	Hard X-Ray imaging	(5)		х					
1.1.9	X-Ray continuum spectrometry	(5)		×	·				
1.1.10	X-Ray burst detector	(5)		х					
1.1.11	X-Ray/gamma ray spectrometer	(5)		x					
1.1.12	Gamma ray spectrometer	(2)		x				ļ	
1.1.13	Solar X-Ray polarimeter	(2)		х					
1.1.14	Bragg reflection crystal polarimeter	(2)		x				<u>                                     </u>	
1.1.15	Solar neutron experiment	(2)		х.				<u> </u>	ļ
1.1.16	High energy gamma ray & neutron detector .	(2)		x	<b>  </b>			<b></b>	
1.1.17	Externally occulted coronagraph	(5)	<b>*</b>	X DIRECTO	ORSTTER				<u> </u>
	ORBITEP I/F NOT REQUIRED			3 FUNCTI	ON CIMULATED			*	·
	OPSITED I/E KIGHEIME BENUIFEN			DIRWT	OPPITER I/F				

Table C.2 Selar Physics Dedicated Nission Enterface Checkout Matrix (Cont)

<del></del>			Tri	TRACES		CHFKCOU EQUIPM	t/test ent		***************************************
BLOCK	PESCRIPTION	* I/F	Baseline Location		OPTION 2	GSK GSK	IVE	SPECIAL FACILITY	RIMARKS
1.2	Froeriment accentance	(1)	Her	х ;				·	
1.2.1 thr	ough 1.2.17 experiments, same as in (1.1) are	(1)		х					
	acceptance tested								
1.3	Experiment functional tests	(2)		х					· · · · · · · · · · · · · · · · · · ·
1.3.1 thr	ough 1.3.17 experiments, same as in (1.1) are	(2)		х		х	<u> </u>		
	individually checked out						<u> </u>		
1.4	Prepare & transfer to User facility	(1)		x					
1.5	Receive & inspect	(1)		х					
1.5.1	Inspect for damage	(1)		х	,				
1.5.2	Check environmental monitors	(1)		х			<u> </u>	<b></b>	
1.5.3	Check environmental protection enclosures	(1)		х					
1.5.4	Review documentation	(1)		х					
1.6	Transfer to assembly & checkout facility	(1)		х			`		
				 			<u> </u>		
2.0	Develop & verify support equipment	(5)		x			<del> </del>	ļ	
2.1	Support equipment	(2)		×	ļ		<b></b>	<b> </b>	
2.1.1	Control & display assembly	(2)		X			<u> </u>	<u>                                     </u>	•
2.1.2	Control & display electronics	(5)		х	<u> </u>		<b></b>		
2.1.3	Tape recorder & magazine	(2)	*	х			<u> </u>	1	
	* ORBITEP I/F NOT REQUIRED		7		ONSITEM ONS CIMULATED				
	OPPITER I/F KIGHLEDGE REQUIFED		Ċ	TOURIG (	OPPITER I/F				

Table C.2 Solar Physics Dedicated Mission Interface Checkout Matrix (Cont)

. :			11**11	RFACES		CHFKCOU	t/test ent		
BLOCK	PESCRIPTION	# I/F KNOWLENGE	Baseline · Location	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITF ORIENTED	GGB GBD.	IVE	SPFCIAL PACILITY	PEMARES
2.1.4	Evdrogen Alpha monitor	(2)	Uner	х		-			
2.1.5	Purge gas subsystem	(5)		· x		-			
2,2	Support equipment acceptance	(1)		x					
2.2.1 thr	cugh2.2.5 support subsystems, same as in (2.1)	(1)		х					
,	are acceptance tested			,					
2.3	Support equipment functional test	(2)		x		х			,
2.3.1 thr	ough 2.3.5 support subsystems, same as in (2.1)	(5)		x		x			
	are individually checked out								
2.4	Frepare and transfer to User facility	(í)		х					<u> </u>
2,5	Receive & inspect	<u>(1)</u>		х					,
2.5.1	Inspect for damage	(1)		х					
2.5.2	Check environmental monitors	(1)		х		المرابعة الكان الماريق والمرابعة			,
2.5.3	Check environmental protection enclosures	(1)		x				-	
2.5.4	Review documentation	(1)		x			-		
2.6, /	Transfer to assembly and checkout facility	(1)		х					
2.7	Pallet mounted support equipment(unpressurized)	(1)		x					
2.7.1	Hydrogen Alpha monitor	(1)		x				,	1
2,7.2	Purge gas subsystem	(1)		x		<u> </u>			
2.7.3	Payload station equipment	(1)	.*	· x		· · · · · · · · · · · · · · · · · · ·			·
	ORRITER T/F NOT REQUIRED		(	JUNEAU TECHNOLOGY	ORSTITER ON CIMULATED				) (
	(2) OPBITED I/F KNOWLEDDE REDUIFED		(	TOFRIG (4	OPPITER I/F			•	•

Table C.2 Solar Physics Dedicated Mission Interface Checkout Matrix (Cont)

	· · · · · · · · · · · · · · · · · · ·		i. Irti	SECFACIES		CHFKCOU.	r/test ent		
BI-OCK	Pescription	* I/F	BASELINE LOCATION	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITF ORIENTED	GSS SIM.	IVE	SPECIAL PACILITY	REMARKS
2.8	Experiment igleo equipment (pressurized)	(1)	Vser	х					
2.8.1	Control & display electronics	(1)		х				1	
2.8.2	Control & display assembly	(1)		х				·	
2.8.3	Tape recorder and magazine	(1)		х					
3.0	Receive pallet & experiment igloo (return of post flight hardware)	(1)		. x					
3.1	Refurbish pellet & igloo	(1)		×					
3.1.1	Service cooling loops	(1)		x					
3.1.2	Functional check of power conditioning equip.	(1)	`	х					
3.1.3	Remove all harnesses	(1)		х					
3.1.4	Inspect & repair applicable harnesses	(1)		x					
3.1.5	Remove non applicable brackets	(1)		x					
3.1.6	Examine mating surfaces	(1)		х,					
3.1.7	Paint .	(1)		x					
3.1.8	Emissivity test	(1)		x	,	·			
3.2	Prepare & transfer to User facility	(1)		x					
3.3	Receive & inspect	(1)		х					
. 3.3.1	Check ID & damage	(1)	<b>†</b>	х					
•	OPSITED I/F NOT REQUIRED OPSITED I/F KNOWLEDGE REQUIESD	,	(	ייייאוריז -	ORRITER ON' CIMULATED OPPITER I/F				



Table C.2 Solar Physics Dedicated Mission Interface Checkout Matrix (Cont)

·		1	1	INN	REPACES	1	· · CHFKCOU	T/TEST		-
BLOCK NO.	přacsípaion	* I/F	BASI	; ELIME CATION	OPTION 1 ) USER ORIENTED	OPTION 2 LAUNCH SITF ORIENTED	GSFC CML	: IVE	SPPCIAL PACILITY	RIMARIE
3.3.2	Beview history	(1)	Uae	er	x			i i		
3.4	Transfer to assembly & checkout facility	(1)		1	, х	'				
3,5	Remove experiment support igloo	· (1)		. '	, х			í		
3.6	Configure isloo structure for support equip.	(1)		. 1	, X	-		<u>'</u>		
3.6.1	Install brackets	(1)		,	. х			'		
3.6.2	Install equipment platforms	, (1)		,	x					
3.6.3	Tratall prhasure lines	(1)	,	1	' X	1		,		
3.6.4	Leak check	(1)			х	,				,
3.7	Install basis equipment in pallet	(1)		,	x		•	,		
3.7.1	Harnesses	(1)		,	x	,				•
3.7.2	Furge lines	(1)			x					,
3.7.3	Cooling lines	(1)			x	,				*
¥•O.	Complete assembly of experiment support igloo	(1)			X					
							å.	<u></u>		
5.ò	Modify pallet to flight configuration	.(2)			x			<u> </u>		
5.1	Install platforms and brackets	(1)		,	х					
5.2	Install gimbal mounts	(1)			х.				ļ	
5-3	Install mech. & electr. connect panels	(5)	1		X DERFOR	ORSITER	} 	<u> </u>		
	ORBITEP I/F NOT REQUIRED	` •		C	FUNCTIO	ma cimulated				•
	OPBITED I/F KNOVLEDGE REQUIESD				DIRTCT C	OFFITER I/F				

Table C.2 Solar Physics Dedicated Mission Interface Checkout Matrix (Cont)

<del>,</del>			, गर्न	BRPACES		CHFKCOM EQUIPME			
BLOCK NO.	PESCRIPTION	* 1/F KNOWLEDGE	Baseline Location	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITF ORIENTED	GS% GIM.	IAE	SPECIAL PACILITY	REMARKS
5.4	Install purge manifolds	(1)	Vser	х					
5.5	Install LM2 coolant loops	(1)		х					
	-								
6.0	Install experiments on pallet	(1)		х					`
6.1	Install experiment igloo on pallet	(1)		х	ļ <b>!</b>		<del></del>		
6.2	Install support equipment	(1)		х					
6.3	Verify integrity of individual assemblies	(1)		x .				<u> </u>	
6.3.1	Electrical bonding check	(1)		х				<u> </u>	
6.3.2	Mechanical alignment check	(1)		х				ļ	
6.3.3	Plugs out continuity	(1)		х		<del> </del>			
6.4 -	Interface connection & verification	(1)		х			 		
6.4.1	Mate & check interface connectors	(1)		, x				J	
6.4.2	Connect fluid line I/F fittings & leak check	(1)		х				<u> </u>	
6.4.3	Check end to end continuity of cabling	· (1)		x	<b> </b>			_	
6.4.4	End to end leak check of fluid systems	(1)		x	<u> </u>			<b>.</b>	
6.5	GSE & simulators checkout	(1)		x	x	х	x	<u>  </u>	Trade study
6.5.1	Check simulators/GSE/facility interfaces	(1)		¥.	x	x	х	<u> </u>	
6.5.2	Perform test readiness checkout	(1)	<b>*</b>	I I	х	х	x	<b></b>	, +
· ·		<u> </u>	L,	5 DIRPOT	ORSITER			1	
	* (1) ORRITEP I/F NOT REQUIRED		(	3 FUNCTI	ONSITER ONS CIMULATED				
	OPBITER I/F KHOVLETXIF REQUIRED		(	TOTRECT	OPPITER I/F				



Table C.2 Solar Physics Dedicated Mission Interface Checkout Matgin (Cont)

			nin	REACES		CHFKCOU. EQUIPM	r/Test Ent -		
BLOCK NO.	DESCRIPTION	* I/F	Baseline Location	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITF ORIENTED	GSR Sim.	IVE	SPECIAL FACILITY	Rinaapigs
7.0	Nate flight equipment with GSE/simulators	(1)	User	x	. х	x	x		Trade study
7.1	Support igloo	(1)		х	х	х	х		
7.2	Payload station equipment	(1)		х	x	х	x	·	
7+3	Pallet & experiments	(1)		x	х	Х	х		
8.0	Pre-power interface verification tests	(1)		х	х	х	x		
8.1	Verify mechanical attachments	(1)		τ̈́χ	х	х			
8.2	Verify electrical continuity	(1)		х	х	х	x		
8.3	Leakage check	(1)		х	х	X		-	
8.4	Bus isolation check	(1)		х .	х	x			
8.5	Check flow through coolant loops .	(1)		x	. х	х ,	х	Ì	
8.6	Check pre-power switch positions	(1)		x	. x	х	X		
9.0	Power-up interface verification tests	(5)		x , , ,	х	х	х		
9.1	Power-up elect. subsystems individually	(1)		x	х	x	х		
9.2	Determine power distribution	(5)		<b>x</b> ′ · . ,	. х	х	х.		
9.3	Verify Caution & warning indicators	(2)		x	x	x	x		
9.4	Power-up control console .	(1)		х	x	×	<u> </u>	<b> </b>	
9.5	Power-up computer facility  * 1 ORBITEP I/F NOT REQUIRED	(1)	<u> </u>		X. ORBITEH WC CIMULATED	x	<b>x</b>	1	<u> </u>
	OBSILES I\k KAOATSIXE ESUALES		C	DIRRT	OPPITER 1/F	-			

Table C.2 Solar Physics Dedicated Mission Interface Checkout Natrix (Cont)

			neti	IRFACES		CHAKCON	r/test int		
BLOCK NO.	pescription	* I/F KNOWLEDGE	Baseline Location	OPTION 1 USER ORIENTEU	OPTION 2 Launch Sitt Oriented	Gam Vin.	IVE .	9PFCIAL FACILITY	REMATECT
9.6	Verify facility ready for data processing	(1)	User	х	x	x	x		Trade study
9.7	Verify auxiliary equipment operating	(1)		х	х	х	х		
10.0	Individual experiment verification	(5)		x	х	X	X		
10.1	Deployment mechanism operation	(2)		х	X	X			
10.2	Verify controls & displays	(5)		х	х	х	х .		Trade study
10.3	Operation of experiment sensors	(2)		х	х	χ	х		
10.4	Operation of control & display electronics	(2)		х	х	х	×		
10.5	Verify all flt support equipment operating	(2)		х	х	х	x		
11.0	Integrated experiment tests	(8).(3)		Х	. x	x	х		Both sites; trade study
11,1	Mission sim. test with experiments operating	(3)		x	¥	х	x		
11,2	Integrated experiment deployment	(5)		x	х	x	х		
12.0	EMC/RFI tests (EMC/RFI monitored during	(5)		х	х	х	х		
	operation of experiments & flight support equipment)			<del>,</del>			<u> </u>		
13.0	Data review & retest, if required	(1)	•	х	x	. <b>x</b>	х		
	* (1) ORRITEP T/F NOT REQUIRED	<u>[</u>		DIRPCT FUNCTI	URTITER ON CIMULATED		l	<u> </u>	
•	OPBITED I/F KNOWLEDGE REQUIPED			DIRWIT	OPPITER I/F				

			TI:TI	RFACES		CHPKCOU EQUIPM			
BLOCK	Prescription	* I/F	Baseline Location	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITE ORIENTED	GSR GIM.	IVE	SPECIAL FACILITY	REMARKS
14.0	Secure GSE & simulators	(1)	User	х	x	x	x		Either site
14.1	Disconnect GSE	(1)		х	х	х	х		
14.2	Disconnect & cap coolant lines	(1).		x	X	Х	x		
14.3	Remove electrical connectors	(1)	<del></del>	x	х	х	х		. :
15.0	Post-test examination	(1)		x	χ				
15.1	Framine experiments/pallet/experiment isloo	<u>(1)</u>		х	х				
15.2	Clean & repair, as necessary	(1)		Х	X		·		
15.3	Load non-hazardous supplies	(1)		x	X				
16.0	Closeout, package & ship to Launch Site	(1):		х					
16.1	Closeout & seal experiments	(1)		x				-	
16.2	Paint as required	(1)		х					
16.3	Emissivity test	(1)	<u> </u>	х	<u> </u>				1
16.4	Install transportation servicing equipment	(1)	<del>,                                    </del>	x					
17.0	Pallet/expers/flt. support equip. arr. @IS	(1)	Launch site		, x		,		
17.1	Transfer to receiving facility	(1)			х	<u> </u>		<b> </b>	
17.2	Perform receiving inspection operations	(1)	<u> </u>	5 DIRPOT	X ORSITER		<u> </u>	1	
	OPRITED I/F MOT REQUIRED  OPRITED I/F KNOWLEDGE REQUIFED		. (	y fueri	ON" TIMULATED OPPITER I/F	ı			

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Table C.2 Solar Physics Dedicated Mission Interface Checkout Matrix (Cont)

				INTERFACI	S .		CHPKCOU EQUIPM	t/test ent			•
BLOCK NO.	DESCRIPTION	* I/F KNOWLEDGE	BASELI LOCAT	NE USI	on 1 Tr Tred	OPTION 2 LAUNCH SITP ORIENTED	GSS GSS	IVE	SPFCIAL PACILITY		RIMARKS
17.3	Transfer to integration facility	(1)	Launch s	ite	Χ .	х					
18.0	Refurbished support system igloo arrival	(1)				х					
	at Launch Site							<u> </u>		<u> </u>	
18.1	Transfer to receiving facility	(1)				x					
18.2	Ferform receiving inspection operations	(1)				x		<u> </u>			
18.3	Transfer to integration facility	(1)			,	x ·	-	· · · · · · · · · · · · · · · · · · ·			
19.0	Perform mating operations	(1)				х					
19.1	Interface verification	(1)				х	x				
19.1.1	End to end electrical continuity check	(1)				x	х	x		-	
19,1,2	Verify alignment	(1)				x	<u> </u>				
19.1.3	Interface continuity check	(1)				х	x				
19.1.4	Leak check fluid lines	(1)	-			х	x				
20.0	Prepare payload interface GSE	(1)			x	х	x x	х		Eithe	y site
20.1	Move to integration facility	(1)			X	х	x	X			
50.5	Configure for experiment support	(1)			<u>x</u>	х	х	<u> </u>			
20.3	Connect facility fluid lines  * (1) ORBITEP I/F NOT REQUIRED	(1)		(3)	X X	X ORSITER	<u> </u>	х	,	<u>'</u>	ę <u> </u>
	OPBITEP I/F KNOWLEDGE REQUIERD				ECHCATO	OFFITER I/F					



Table C.2 Selar Physics Dedicated Mission Interface Checkout Matrix (Cont)

				D. THREACES		CHPKCOU EQUIPM	t/test ent			
BLOCK NO.	drscription	* I/F	BASELII LOCATI		OPTION 2 LAUNCH SITF ORIENTED	GSF Sim.	ive	* SPPÇIAL PACILITY		RIMATECI
20.4	Connect facility electrical barness	(1)	Launch st	ite X	х	x	х		Eithe	r site
21,0	Verify operation of payload/GSE interface	(2)		, х	х	х	х			
21.1	Fuel cell/power distribution simulation	(5).		х	х	X	x			
21.2	Orbiter signal simulation	(2)		х	х	x	х			
21.3	Paylond specialist station equipment operation	(2)		x	X	<u>,</u> x	х			
21.4	Verify umbilical interfaces	(2)		х	x	x	x			
21.5	Verify oper. of data management subsystems	(2)		х	х	х	х			
21,6	Verify oper. of computer periph. equipment	(1)		х	х	x	x			
21,7	Verify operation of operator console	(1)		X	х	х	х			<b>†</b>
22.0	Nate payload & interface GSE	(5)		х	х	х	x		Eithe	r site; trade study
22.1	Pre-power interface verification	(2)		x	х	х	х			
22,1,1	Verify payload/GSE mech. elect. & fluid I/F	(1)		x	х					
22.1.2	Perform bus isolation tests	(1)	·	x	x	х ·	х	ļ	·	
22,1,3	Conduct pre-power switch list	(5)		x	x	x	x	ļ		
22,1,4	Check operation of all caution & warning	(5)		x	×	х	x	ļ		
22.2	Power-up functional checkout	(2),(3)		х	x	х	x	1		
22.2.1	Monitor bus voltages & current	(5)	<b> </b>	x	х	х	х	<b>!</b>	,	
22.2,2	Monitor coolant flow	(5)		x	x	x	x			
23.2.3	Check power distribution  * 1 ORRITED I/F NOT REQUIRED	(5)	1 +		X ORSITER OPTENULATED	Х	x	<u> </u>	<u></u>	
	S OBBITED I'E KION'EDOE BEGINDEN			TOWNS OF	OPPITER I/F					

Table C.2 Solar Physics Dedicated Mission Interface Checkout Matrix (Cont)

				INTERPACES		CHYKCOU EQUIPM	T/TEST Ent		REMARKO	
BLOCK NO.	Prscription	MONTRIGE + 1/b	BASELI LOCAT		Option 2 Launch Sitt Oriented	GSB	IVE	SPFCIAL FACILITY		
22.2.4	Initiate computer self-checks	(1)	Launch s	ite X	x	x	х		Either site; trade study	
22.2,5	Conduct command & control test	(2)		х	х	. х	x			
22.2.6	Evaluate computer peripheral equipment	(5)		x	х	x	х	<u> </u>		
23.2.7	Functional c/o of support systems and experiment equipment interfaces	(3)		x	x	X	x			
22.2.8	Evaluate manual emergency backup modes	(3)		х	x	<u>x</u>	x			
23.0	Data review	(2)		х	х				Trade study, see 22,2	
24.0	Remove and secure interface GSE	(1)		x	х	x	х			
25.0	Close-out operations	(1)			x					
25,1	Load non-hazardous supplies	(1)			х					
25.2	Install access panels & cover plates	(1)			x				•	
25.3	Touch-up paint	(1)			х					
25.4	Entasivity_test	(1)			х					
26.0	Pressura dacay leak check	(1)	<b>*</b>		х					
	* (1) ORBITEP I/F NOT REQUIRED		<u></u>	3 DIRPOT	ORSITER ONS SIMULATED		<u> </u>			
	OPBITER I/F KNOWLEDGE REQUIFED			O DIRECT	OPPITER I/F					

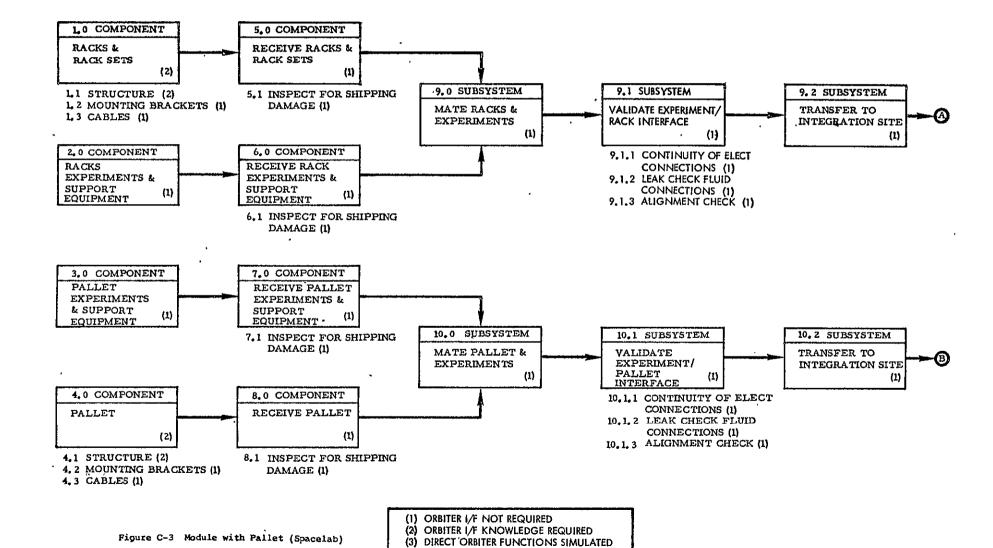
Table C.2 Solar Physics Dedicated Mission Interface Checkout Matrix (Cont)

<del> </del>	DRSCRIPTION	TETRIFACES				CHPKCOUT/TEST EQUIPMENT			,
BLOCK NO.		: * I/F	BASELINE LOCATION	OPTION I USER ORIENTED	OPTION 2 LAUNCH SITF ORIENTED	GST GST	IVE	SPECIAL	, REMATECS
27.0	Volght & balance test	<b>'</b> (1)	Launch site	,	, x	, ;		. ж	
	·	;	·				:		; 1
28.0	Transfer to OFF	(1)	!		' x	<u> </u>	'		
29.0	Prepare cargo for mating with orbiter	(5)		ł	x			<b> </b>	
29,1	Inspect interface connectors	(2)	<u> </u>		x			ļ	
29.2	Service with non-hazardous materials	(1)	<u> </u>		` x				
29.3	Install flight batteries and chargers	(1)		<u> </u>	X			<b>}</b>	
	* ,			,		<del></del>		ļ	
30.0	Install payload control & support equip. in Or		<b></b>		x				
30.1	Install cargo in orbiter payload bay	(4)			х	<del>,</del>	-		
31.0	Orbiter/carso interface verification test	(4)			x				•
31.1	Flugs out electrical continuity	(1)			х				
31.2	Connect & verify umbilicals & fluid lines	(4)			х				
31.3	Install & verify mech. latches & retention dev	(4)	<b>*</b> '		x				
· · · · · · · · · · · · · · · · · · ·									
	* (1) ORRITEP I/F NOT REQUIRED		(	3 DIRECT FUNCTI	ORSITER ON CIMULATED		•		
	OPSITED I/F KNOWEDOE REQUIESD	•	(	TOFFILE (T	OFFITER I/F				

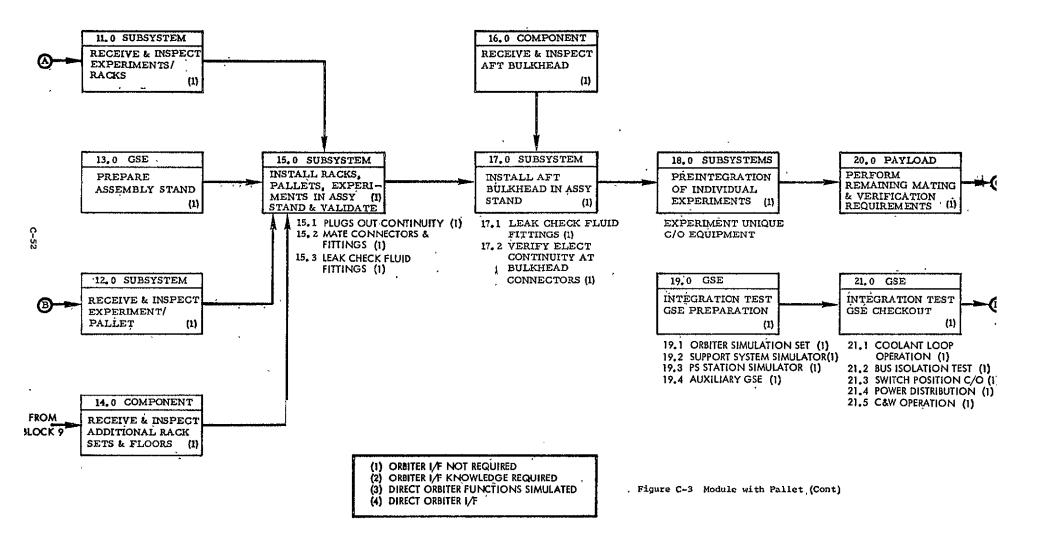
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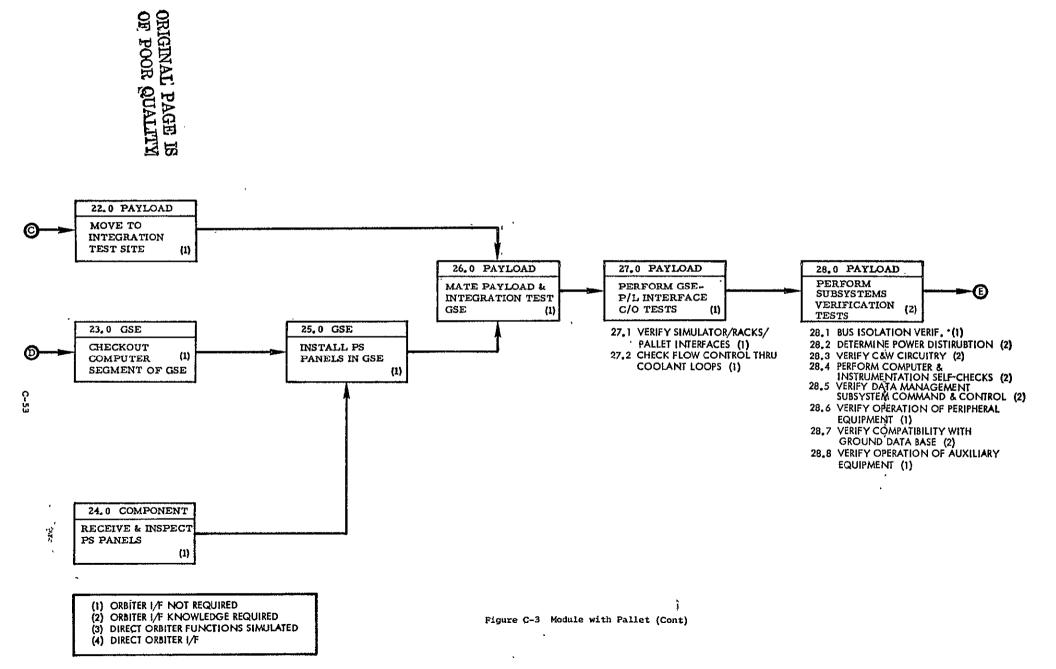
Table C.2 Solar Physics Dedicated Mission Interface Checkout Matrix (Cont)

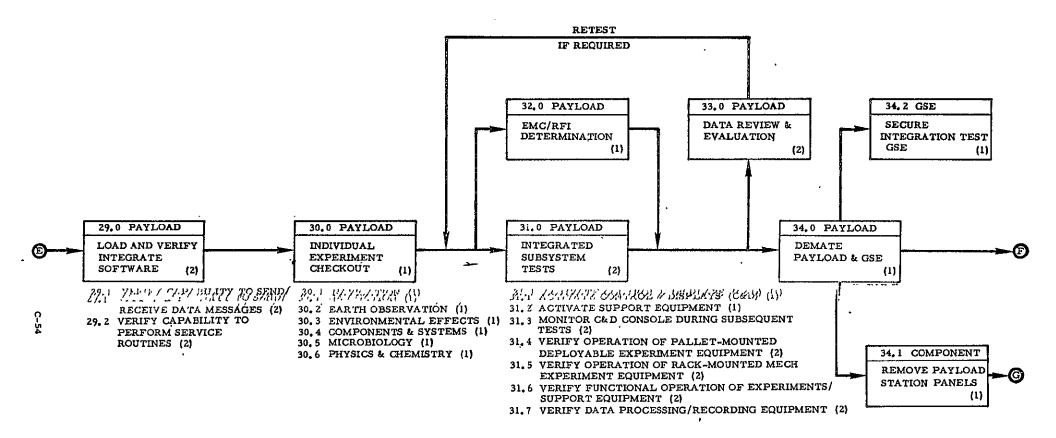
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	Precription	TP TARWACKS				CHYKCOUT/TEST EQUIPMENT		-	
BLOCK NO.		* 1/P	Baseline Location	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITH ORIENTED	GSW Sim.	IVE	SPECIAL FACILITY	RIMAPICS
32.0_	Orbiter avionics test	(4)	Taunch site		x				
32.1	Verify compatibility of experiment avionics with shuttle support avionics	(4)			Y				
<del></del>		1 (3)						-	
33.0	EMC/RFI test	(4)			x ·				
34.0	Closeout of payload bay	(4)			х				
35-0	Ready for Leunch Operations	( <u>4</u> )	· · · · · · · · ·		×				
36.0	Launch Operations	(4)			х				
						,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,			
			:						
					ļ			<u> </u>	
					<b> </b>				
	* (1) ORRITEP T/F NOT REQUIRED	<u> </u>		DIRPOR	ORTITER CETATUMIT OK		1	<u> </u>	
OPBITEP I/F KTOVLEDGE REQUIFED  DIRECT OPPITER I/F									



(4) DIERCT ORBITER I/F

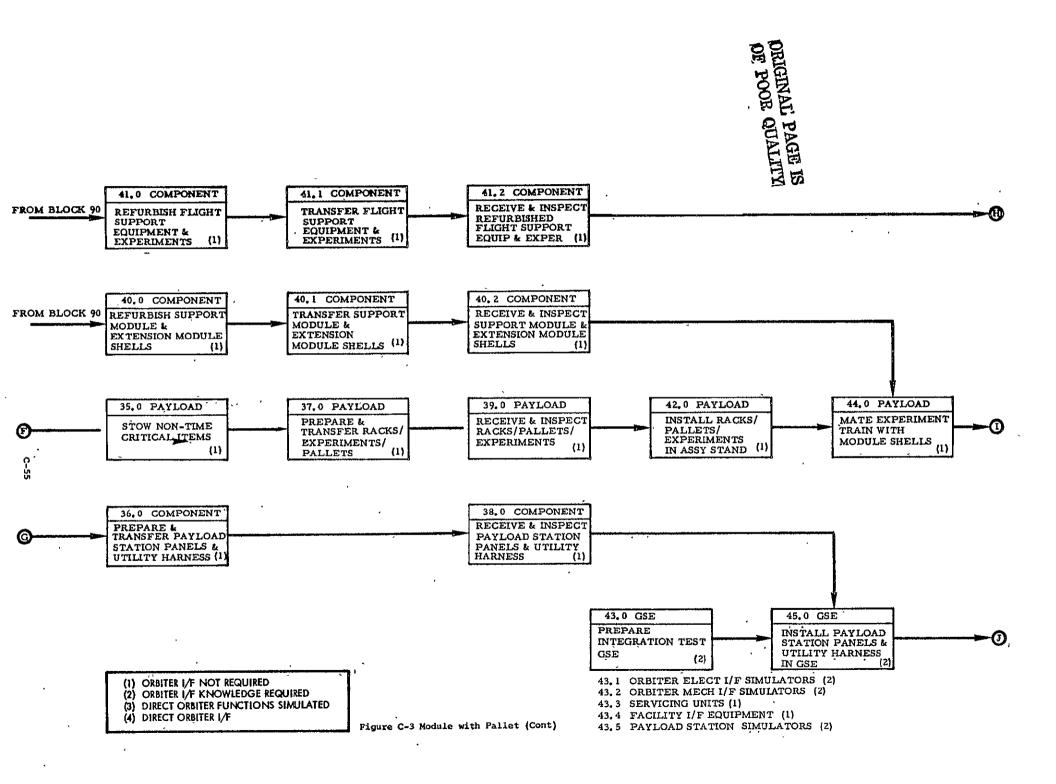






- (1) ORBITER I/F NOT REQUIRED
- (2) ORBITER I/F KNOWLEDGE REQUIRED
- (3) DIRECT ORBITER FUNCTIONS SIMULATED
- (4) DIRECT ORBITER I/F

Figure C-3 Module with Pallet (Cont)



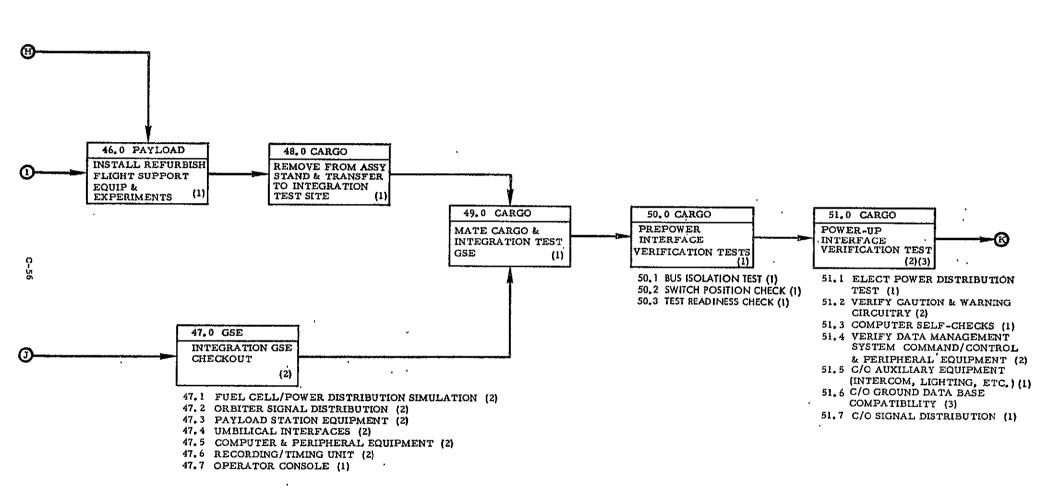


Figure C-3 Module with Pallet (Cont)

(1) ORBITER I/F NOT REQUIRED
(2) ORBITER I/F KNOWLEDGE REQUIRED
(3) DIRECT ORBITER FUNCTIONS SIMULATED

(4) DIRECT ORBITER I/F

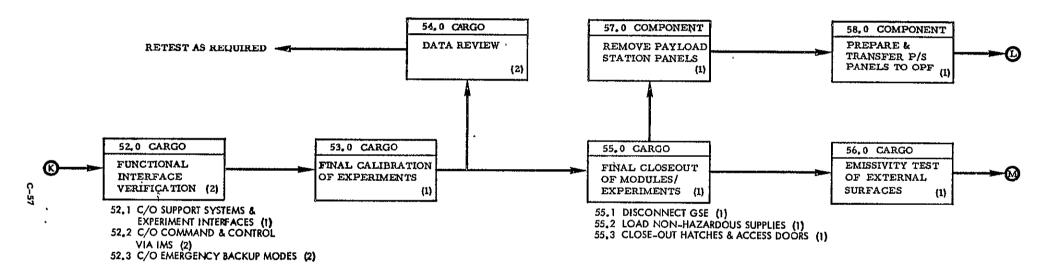
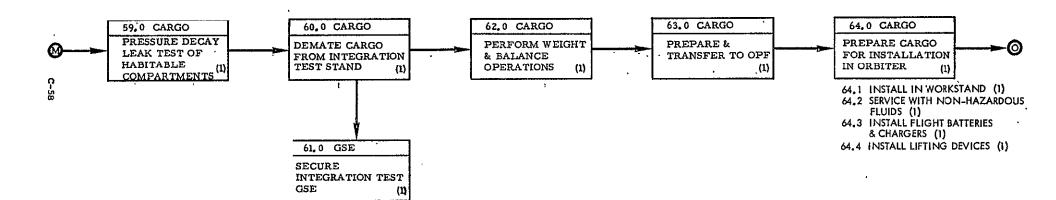


Figure C-3 Module with Pallet (Cont)

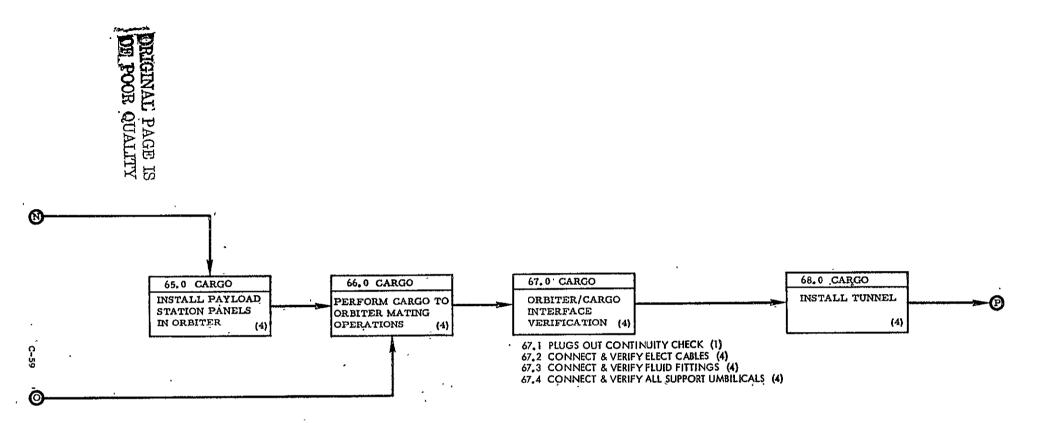
<sup>(1)</sup> ORBITER I/F NOT REQUIRED
(2) ORBITER I/F KNOWLEDGE REQUIRED
(3) DIRECT ORBITER FUNCTIONS SIMULATED
(4) DIRECT ORBITER I/F





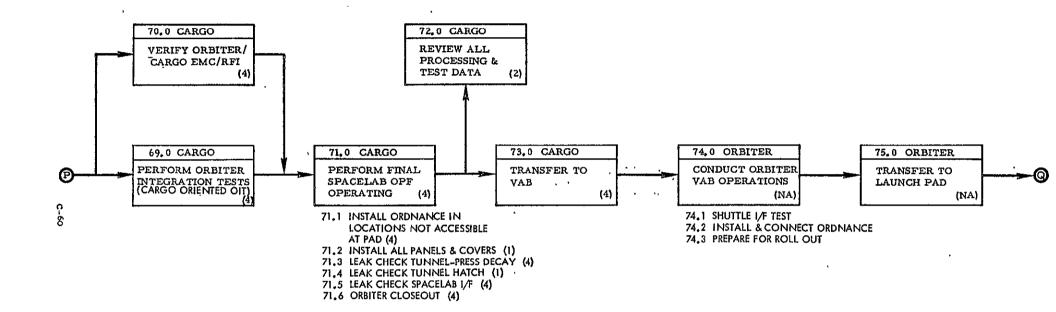
- (1) ORBITER I/F NOT REQUIRED
- (2) ORBITER I/F KNOWLEDGE REQUIRED
  (3) DIRECT ORBITER FUNCTIONS SIMULATED
- (4) DIRECT ORBITER I/F

Figure C-3 Module with Pallet (Cont)



- (1) ORBITER I/F NOT REQUIRED
  (2) ORBITER I/F KNOWLEDGE REQUIRED
  (3) DIRECT ORBITER FUNCTIONS SIMULATED
  (4) DIRECT ORBITER I/F

Figure C-3 Module with Pallet (Cont)



(1) ORBITER I/F NOT REQUIRED

(2) ORBITER I/F KNOWLEDGE REQUIRED
(3) DIRECT ORBITER FUNCTIONS SIMULATED

(4) DIRECT ORBITER I/F

Figure C-3 Module with Pallet (Cont)

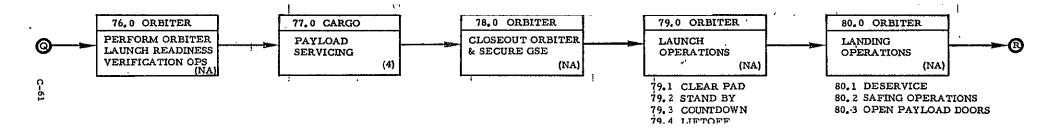
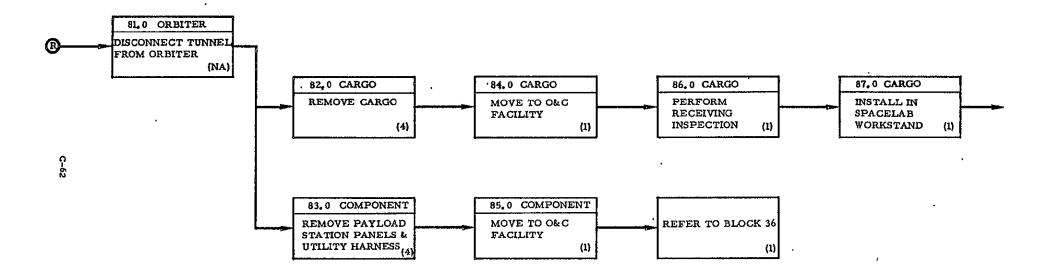


Figure C-3 Module with Pallet (Cont)

- (1) ORBITER I/F NOT REQUIRED
  (2) ORBITER I/F KNOWLEDGE REQUIRED
  (3) DIRECT ORBITER FUNCTIONS SIMULATED
  (4) DIRECT ORBITER I/F

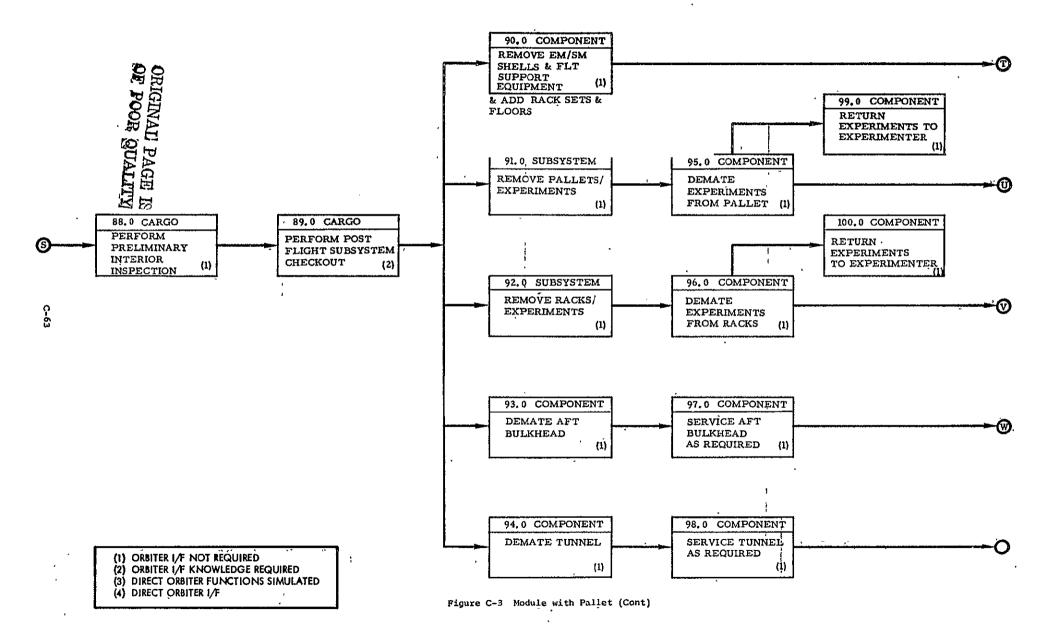


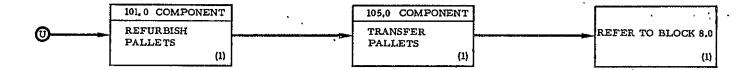
Pigure C-3 Module with Pallet (Cont)

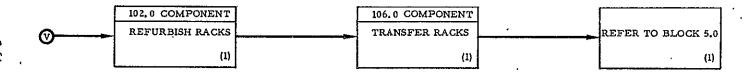
- (1) ORBITER I/F NOT REQUIRED
- (2) ORBITER I/F KNOWLEDGE REQUIRED

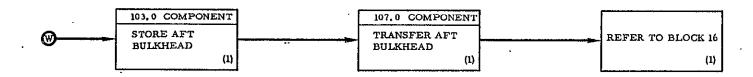
  (3) DIRECT ORBITER FUNCTIONS SIMULATED

  (4) DIRECT ORBITER I/F









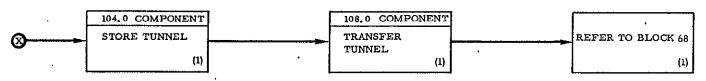


Figure C-3 Module with Pallet (Cont)

- (1) ORBITER I/F NOT REQUIRED
  (2) ORBITER I/F KNOWLEDGE REQUIRED
  (3) DIRECT ORBITER FUNCTIONS SIMULATED
  (4) DIRECT ORBITER I/F

# DRIGINAL PALE IN

## TABLE C.3 MODULE WITH PALLET (SPACE)

PA TR	·		· Text		; ; ", , ,	CHIPKCOU	t/test ent,			
elock no.	description	FI/F ENOWLEDGE	raseline Location	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITH ORIENTED	GSW.	IVE	SPECIAL I		MAPES
1.0	Racks and rack sets	(2)	User	х				,	Várious	sources
1.1	Structure	(5)		×				,	-	1
1.2	Mounting brackets	(1)	,	x		:			,	
1.3	Cables	(1)		, x	·	, ,	,			'
2.0	Racks experiments & support equipment	(i)		х		,	;			
3.0	Pallet experiments & support equipment	(1)		x						
4.0	Pallet	(5)	,	х		``				
4.1	Structure	(5)		х						
4.2	. Nounting brackets	(1)		х	·					
4.3	Çables	(1).		х		a			it.	
5.0	Receive racks & rack sets	(1)		, x	·		.,			
5.1	Inspect for shipping damage	(1)		х					<u>-</u>	
6.0	Receive racks experiments & support equipment	(1)		x			: ,			
6.1	Inspect for shipping damage	(1)	*	X X	DRAMMER		<u> </u>	1	<u> </u>	7
6.1	Inspect for shipping damage  * (1) ORBITEP I/F NOT REQUIRED  OPBITEP I/F KNOWLEDGE REQUIRED	(1)	(	3) DIRFCI FUNCTI	ORSITER ONC CIMULATED OPPITER I/F	, , , , , , , , , , , , , , , , , , , ,		<u>. L </u>	L	V

Table C.S (Cont) MIDULE WITH PALLET (SPACELAB)

<del></del>			net	RVACES		- CHPKCOU EQUIPM	t/test Ent		
BLOCK	description	* I/F	raseline Location	OPTION 1 USER ORIENTED	option 2 Launch Site Oriented	GST SIM.	IVE	SPECIAL FACILITY	REMARKO
7.0	Receive pellet experiments & support equipmen	(1)	User	x					Various sources
7.1	Inspect for shipping damage	(1)	<u> </u>	х					<u> </u>
8.0	Receive pallet	(1)		х					
8.1	Inspect for shipping damage	(1) ·		x					*
9.0	Mate racks and experiments	(1)		x					
9.1	Validate experiment/rack interface	(1)		х			<b> </b>	_	
'9-1-1	Continuity of electrical connections	(1)	,	х			ļ		
9.1.2	Leak check fluid connections	(1)		х			ļ	_	
9.1.3	Allgregent check	(1)	,	х					
9.2.	Transfer to integration site (user)	(1)	3	x	· · · · · · · · · · · · · · · · · · ·				
10.0	Mate collet and experiments	(1)		x					
16.1	Validate experiment/pallet interface	(1)	\ <u></u>	x		·	<b>}</b>	<b> </b>	
16.1.1	Continuity of electrical connections	<u>(2)</u>	1	х			ļ		
10.1.2	Leak check of fluid connections	(1)		х			<u> </u>		
10.1.3	Alignement check	(1)		х				ļ	
10.2	Transfer to integration site  * (1) ORBITEP T/F NOT REQUIRED	(1)	• •	X DIRPOR	ORSITEH	·		1	
	OPETTEP I/F KIOVLEDGE REDUIFED		(	DIRWI	oppiter i/F				

님	······································			77:T	irpaces		CHERCOUT	/TEST		
	BLOCK NO.	pescription	* I/F RNOWLEDGE	Baseline Location	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITF ORIENTED	GSS SIM.	IVE	SPPCIAL PACÍLITY	REMARKS .
	´11.0	Receive & inspect experiments/xxcks	(1)	Uner	х					
	12.0	Receive & inspect experiment/pellet	(1)		<u>x</u>			,	•	
	13.0	Prepare assembly stand	(1)		х			<i>,</i>		
	14.0	Receive & inspect additional rack sets & floo	r (1)		х					
	15.0	Tostall racks, ballets, experiments in assembly stand & validate	(1)		Х					·
	15.1	Plugs out continuity	(1)		x					
ł	15.2	Mate connectors and fittings	(1)		х					
	15.3	Leak check fluid fittings	(1)		Х					,
	16.0	Receive and inspect aft bulkhead	(1)		х					
	17.0	Install aft bulkhead in assembly stand	(1)		x					
	17.1	Leak check fluid fittings	(1)			ļ				
	17.2	Verify electrical continuity at bulkh. conn.  1) ORBITEP I/F NOT REQUIRED  2) ORBITEP I/F KNOWLEDGE REDUTRED	(1)	(	シ functi -	ORSITER ONT FIMULATED OPPITER I/F				

			neir	TRPACES	•	CHPKCOU EQUIPM				
BLOCK NO.	PRSCRIPTION	* I/F	Baseline Location	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITY ORIENTED	GSB SM.	IVE	SPECIAL FACILITY	RIMAPICS	
18.0	Pre-integration of individual experiments	(1)	User	х	х			}	Unique C/O equipment required. Either site:trade study.	
19.0	Integration test GSE preparation	(1)		X	<u> </u>	x			Fither site	
19.1	Orbiter simulation set	. (1)		х	x	x				
19.2	Support system simulator	(1)		х	x	х				
19.3	PS station simulator	(1)		×	x	х				
19.4	Auxiliary GSE	(1)		x	x	x	,		<b>+</b>	
20.0	Perform remaining mating & verif. requirements	(1)		х	х		,		Either site: trade study	
21.0	Integration test GSE checkout	(1)		х	х				Eithen eite	
21,1	Coolant loop operation	(1)		Y	у	·				
21,2	Bus isolation test	(1)		x	х			<u> </u>		
21.3	Switch position checkout	(1)		Х	х		<u> </u>	ļ		
21,4	Power distribution	(1)		x	х					
21.5	Caution & warning operation	(1)		х	х	• • • • • • • • • • • • • • • • • • •	,		<u>†</u>	
22.0	Hore to integration test site	(1)	ţ.	х	x				Either site	
	* (1) ORPITED 1/F NOT REQUIRED			DIRPOR FUNCTI	ORBITER ONT CIMULATED			1		
	OPBITER I/F KNOWLEDGE REQUIERD			DIRWIT	OPPITEP I/F					

# SS POOR OUTLINESS IS

### Table C.3'(Cont) MODULE WITH PALLET (SPACELAB)

			Trif	REACES	ì	CHFKCOFF EQUIPM	r/Test Int		_	
BLOCK NO.	description	MUCMTRIĆE * I\E	Baseline Location	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITF ORIENTED	ggr Sim.	IVE	SPFCIAL PACILITY	RPMAPKS	
23.0	Checkout computer segment of GSE	(1)	User	Y	х				Either site	
23:1	Console operation	(1)		х	" x			ļ		
23.2	Commuter/instrumentation selfcheck	(1)		Х	x					
23.3	Verify command/control sapability	(1)		٠x	х				·	
S#.0	Receive & inspect PS vanels	(1).		х	х .					
25.0	Install PS panels in GSE	(1)		х	x	,				
26.0	Note payload and integration test GSE	(1)		. X .	x				V-	
27.0	perform GSE/payload interface c/o tests	(1)		х	х	х	Х		Either site; trade study	
27.1	Verify simulator/racks/pallet interfaces	(1)		x	х	х	x			
27.2	Check control lools flow control	(1)		х	х	x	х		<u> </u>	
28.0	Perform subsystem verification tests	(5)		х.	х	x.	х :		Trade study	
28.1	Bus isolation verivication	(1)		. х	х	X	х			
28.2	Determine power distribution	(5)		· x	x	x	x			
28.3	Verify caution & warning circuitry  * (1) ORBITEP I/F HOT REQUIRED	(2)	<u> </u>	х 3) <u>рікесе</u> 5 <b>г</b> елесе	X 'URSITER ONO SIMULATED	x	х	<u> </u>	L	
	OPSITED I/F KNOWLENGE REQUIESD.			_	OPPITER I/F					

Table C.3 (Cont) MODULE WITH PALLET (SPACELAB)

			Tr'ti	TREACES		СН-КСОГЛ ЕQUІТНІ	r/Test ·			
BLOCK NO.	PRSCRIPPION	* I/F	Baseline Location	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITF ORIENTED	695 °I'4.	IVE	SPECIAL PACILITY	<del>БРМ</del> АРКЛ	
28.4	Perform computer & instrumentation selfchecks	(2)	lleer	х	х	У	X		Trade study	
28.5	Verify data management subsystem com.& control	(1)		X	γ	У	X			
28.6	Verify operation of peripheral equipment	(1)		Y	х	Х	· x			
28.7	Verify compatibility with ground data base	(3)		у	x	х	Y	ļ		
28.8	Verify operation of auxiliary equipment	(1)		Υ	у	Υ	хх		*	
29.0	Load and verify integrated software	(2)		х .	у	Υ	Х		Fither site	
29.1	Verify capability to send/receive data message	. (S)		Y	Υ	×	X			
59.2	Verify capability to perform service routines	(2)		х	ү	у	×			
30.0	Individual experiment checkout	(1)		x	X				Unique experiment equirment	
30.1	Navigation	(3)		Y	γ					
30-2	Farth_chservation	(1)		y	Y					
. 30.3	Environmental effects	(1)		Y	\ \			_		
30.4	Components & systems	(1)		×	x		<u></u>	ļ		
30.5	Microbiology	(1)		, x	у					
30.6	Physics and chemistry	(1)	<b>V</b>	Y	Y	3			<u> </u>	
				22 DI RM-1	ORFITESR					
	* (1) ORRITED 1/F NOT REQUIRED		(	3) FUNCTI	ONT TIMULATED					
	OBSILED I\L KIOALEDDE ESDALESIA		(	DIRWIT	OPPITEP I/F					



Table C\_3 (Cont) | MODULE WITH FALLET (SPACELAB)

	,		T:TI	HO'ACES	1	CH KCOU EQUIPM	t/test ent			
BLOCK NO.	Prscription	MONTEIGE * 1/E	Baseline Location	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITF ORTENTED	G9W SIM.	IVE	SPECIAL PACILITY	REMARKS	
31.0	Integrated subsystem tests	(2)	User	Ř	х	х	х .		Trade study	
31.1	Activate control and displays	(1)		'x	х	x	x			
31.2	Activate support equipment	(1)	ı	х	x	х	х	-		
31.3	Monitor CaD console during subsequent tests	(2)		x	х	х	х			
31.4	Verify operation of pallet-mtd deployable experiment equipment	(2)		х	x ·	· х	x			
31.5	Verify operation of rack-mtd mechanical experiment equipment	(2)		х	х	Х	х			
31.6	Verify functional operation of experiments/support equipment	(5)	1	Х	Х	X	, X			
31.7	Verify data processing/recording equipment	(5)	,	x X	, х	X	х'		•	
32.0	EMC/RFI determination	· (1)	, ,	x	; X				Special equipment required	
33.0	Data review and evaluation	(5)		, , x	Х	•			Can be accomplished enywhere	
34.0	De-mate psyload & GSE	(1)	· · ·	x	X X					
34.1	Remove payload station panels	(1)		X	' Y		· '		· .	
34.2	Secure integration test GSE  * (1) ORBITEP I/F NOT REQUIRED	(1)	♦	X DIRPOR FUNCTION	X ORBITER CETATUMIE ON		<u> </u>			
	OPBITED I/F KNOWLEDGE REQUIERD		(	_	OPPITER I/F					

Table C.3 (Cont) MODULE WITH FALLET (SPACELAB)

			·	Int	IRFACES		CHPKCOUT EQUIPM	/TEST			-
BI-OCK NO.	Pescription	* I/F KNOWLENGE		ELINE XATION	OPTION 1 USBR ORIENTED	OPTION 2 LAUNCH SITF ORIENTED	GGZ CGZ	IVE	SPECIAL PACILITY	R	FMARKO
35.0	Stow no n time-critical items	(1)	Us	er	. х	х		<u>,</u>		Either site	
36.0	Prepare & transfer P/L stajon panels & util.har	a. (1)			x	x					
37.0 -	Prepare & transfer racks/experiments/pallets	(1)			X	x					
38.0	R & I navloed station panels & utility harmess.	(1)			X	х					
39.0	R & I racks/pallets/experiments	(1)		,	x	х					
40.0	Refurbish SM & extension module shells	(1)	Launc	h site	\	х		ļ			
40.1	Transfer SM & extension module shells	(1)				х					
40.2	R & I SM & extension module shells	(1)			·····	х					
41.0	Refurbish FS equipment & experiments	(1)				х		<u> </u>		<del>-</del>	
41.1	Transfer FS equipment & experiments	(1)				х		•			
41.2	R & I refurbished FS equipment & experiments	(1)				x					
42.0	Install racks/pollets/experiments in easy stand	(1)				х					, <u>,</u> ,
43.0	Prepare integration test GSE	(5)			х	х	x	х		Either site	; trade study
43.1	Orbiter electrical I/F simulators	(2)	L		· x	х	х	х			
43.2	Orbiter mechanical I/F simulators	(5)			x	х	x	х			
43.3	Servicing units	(1)			x	х	x	x			
43.4	Facility I/F equipment	(1)			х	x	x	x			
43.5	Payload station simulator	, (5)	<u> </u>		х	х	x	х		·	<u> </u>
եե Օ	Maje experiment train with module shells	(1)	. 1	,	<u>у</u>	URSITER		·	<u> </u>	Either site	·
*	ORPITEP T/F NOT PROUITED			C	FUNCTIO	ONT CIMULATED					
	OPBITED I/F KNOWLEDGE REQUIRED				DIRTO	OPPITER I/F				•	

Table C.3 (Cont) MODULE WITH PALLET (SPACELAB)

			· I	THE PACES		CHFKCOU			
BLOCK NO.	PĘSCRIPTION	* 1/P KNOWLEDĆE	Baselin Locati		OPTION 2 LAUNCH SITE ORIENTED	G98 Sim.	IVE	SPECIAL PACILITY	REMAPES
<u>45.0</u>	Install P/L staion panels & utility Harness in	ae (s)	Launch si	te X	х				Either site
46.0	Install refurbished FS equipment & experiments	(1)			х				
47.0	Integration GSE checkout	(5)		x	х	х	х		Either site
47.1	Fuel cell/power distribution simulation	(5)		x	x	X	· x		
47.2	Orbiter signal distribution	(2)	1	х	х	х	x		
47.3	Payload station equipment	(5)		x	x	<u>x</u>	х		
47.4	Umbilical interfaces	(2)		x	x	х	x		
47.5	Computer & peripheral equipment	(3)		x	<u> </u>	Х	· x		<u> </u>
47.6	Recording/timing unit	(5)		x	x	х	х		
47,7	Operator console	(1)		х	x	х	x		<b>,</b>
48.0	Remove from assy stnd & transfer to int.test at	te (1)			x				
49.0	Mate cerso & integration test GSE	(1)			x	хх	x.		GSE simulator Vs IVE trade study
50.0	Pre-power interface verification tests	(1)		See remarks	x		x		If accomplished at User site, demate, ship, remate and reverify
50.1	Bus isolation test	(1)			x .		x		at launch site
50.2	Switch nomition check	(1)	<u> </u>		x		x		
50-3	Test readiness check	(1)			х		х ′		*
51.0	Power-up interface verification test	(2).(3)		<u>                                     </u>	x	X	X		
51.1	Flectrical power distribution test	(1)			х.	х	х.	<u> </u>	
51,2	Verify caution Evarning circuitry  (1) ORBITED I/F NOT REQUIRED	<u>(5)</u>	₩	3 DIRFET	X PRITERU CETAJUMIC CA	х	Х.		
	OPSITEP I/F KIGHLEDGE REGULERD			DIRWT	OPPITER I/F				

Table C.3 (Cont) MODULE WITH PALLET (SPACELAB)

			nei Teri	ERPACES		CHFKCOU				
BLOCK NO.	, Pescription	* 1/F KNOWLEDGE	BASELINE LOCATION	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITP ORIENTED	GSE SIM.	IVE	SPECIAL FACILITY	REMARKS	
51.3	Computer self checks	(1)	Launch site		x	x	х		. If accomplished at User site, demate, ship, remate and reverify	
51.4	Verify data management system command/control.	(2)			x	x	x		at launch site	
51.5	Checkout auxiliary equipment (intercom, lighting, etc.)	· ( <u>1</u> )			x	X	х.		· ·	
51.6	C/O ground data base compatibility	(3)			х	х	х			
51.7	Checkout signal distribution	(1)			х	ж	х			
52.0	Functional interface verification	(2)			х	х.	х			
52.1	C/Osupport systems & experiment interfaces	(1)			xx	х	х			
52.2	Checkout command & control via IMS	(2)			х	х	х			
52.3	Checkout emergency backup modes	(2)			х	х	х		•	
53.0	Final calibration of experiments	(1)		,	х				Experiment unique equipment	
54.0	Tata review	(2)			x				Can be accomplished anywhere	
55.0	Final closeout of modules/experiments	(1)			X					
55.1	Disconnect GSE	(1)			х			,		
55.2	Load non-hazardous supplies	(1)			x					
55.3	Closeout hatches & access doors	(1)		! 	. x .	٠.		<u> </u>		
56.0	Emissivity test of external surfaces	(1)			x					
57.0	Remove payload station panels	(1)			х					
58.0	Prepare & transfer P S panels to OFF	(1)			х.					
59.0	Pressure decay leak test of habitable commertment  (1) ORBITEP I/F NOT REQUIRED	(1)	+ (	3) DIRPOT	ORSITER N° CIMULATED					
	OPBITER I/F KNOVLETOF REQUIERD		Č		OPPITER I/F					

Table C.3 (Cont) MODULE WITH PALLET (SPACELAE)

H		1	IIII	re aces		CH*KCOU EQUIPM	t/test ent			
ORIGINALI PART	description	* 1/P	Baseline Location	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITE ORIENTED	GST SIM.	IVE	SPECIAL PACILITY	REMARKS	
60.0	Demate cargo from integration stand	(1)	Isunch site		x					
61.0	Secure integration test GSE	(1)			х	······································				
62.0	Perform weight & balance operations	ω			X			·x		
63.0	Prepare & transfer to OPF	(1)			х			хх	Special handling equipment	
64.0	Prepare cargo for installation in orbiter	(1)			х					
64.1	Install in workstand	(1)			x					
64.2	Service with non-bazardous fluids	(1)			x					
64.3	Install flight batteries and chargers	(1)			х	····				
64.4	Install lifting devices	(1)			х				<b>*</b>	
65.0	Install payload station panels in orbiter	(4)			х					
66.0	Perform cargo to orbiter mating operations	(4)		·	х					
67.0	Orbiter/cargo interface verification	(4)			х				Orbiter responsibility	
67.1	Plugs out continuity check	(1)			х					
67.2	Connect & verify electrical cables	(4)			X					
67.3	Connect & verify fluid fittings	` (h)			x					
67.4	Connect & verify all support umbilicals	(4)			x					
68.0	Install tunnel	(4)			х					
69.0	Perform orbiter integration test	(14)			х					

OPSITED I/F KNOVLEDOF REDUIFED

DIRECT OPPITED I/F

Table C.3 (Cont) MODULE WITH PALLETS (SPACELAB)

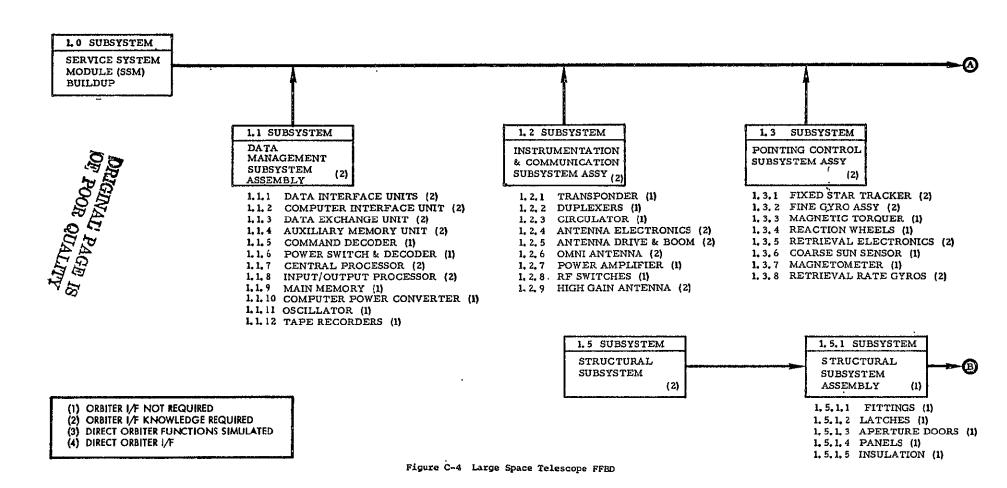
			]	PTARPACES		CHFKCOU EQUIPM	t/test Ent		
BI-CCK NO.	DESCRIPTION	* i/p mowledge	Baselin Locati		OPTION 2 LAUNCH SITP ORIENTED	GSE SIM.	IVE	SPECIAL FACILITY	REMARKS
70.0	Verify orbiter/cargo EMC/RFI	(4)	Launch si	te	х				Orbiter responsibility
70.1	Monitor during OIT functional tests	(4)			х				
71.0	Perform final spacelab OFF operations	(4)					<u> </u>	<u> </u>	Use OPF coningent
71.1	Install ordnance in locations not accessible at	pad(4)			х		<u> </u>		
71.2	Install all panels and covers	(1)			X		<u> </u>		
71.3	leak-check tunnel pressure decay	(4)			х		<u> </u>		
71.4	Leak check tunnel hatch	(1)			x		<u> </u>		
71.5	Leak check spacelab interface	(4)			х				
71.6	Orbiter closeout	(4)			х				
72.0	Review all processing & test data	(5)		,	х				
73.0	Transfer to VAB	(4)			x	-			Orbiter responsibility
74.0	Conduct orbiter VAB operations	na			×				•
74.1	Shuttle interface test	NA			х				
74.2	Install & connect ordnance	NA NA		`	Х				
74.3	Prepare for roll out	NA			х				
75.0	Fransfer to launch pad	NA.			х		,		
76.0	Platform orbiter launch readiness verif. ops.	NA			x				·
77.0	Payload servicing	(4)			х		<u> </u>		As required
78.0	Closeout orbiter & secure GSE	na ·	<u> </u>	711 00000	X ORRITER				
	* (1) ORBITED I/F NOT REQUIRED  OPSITED I/F KNOWLEDGE REQUIESE	•		FUNCTI	OFFITER 1/F				

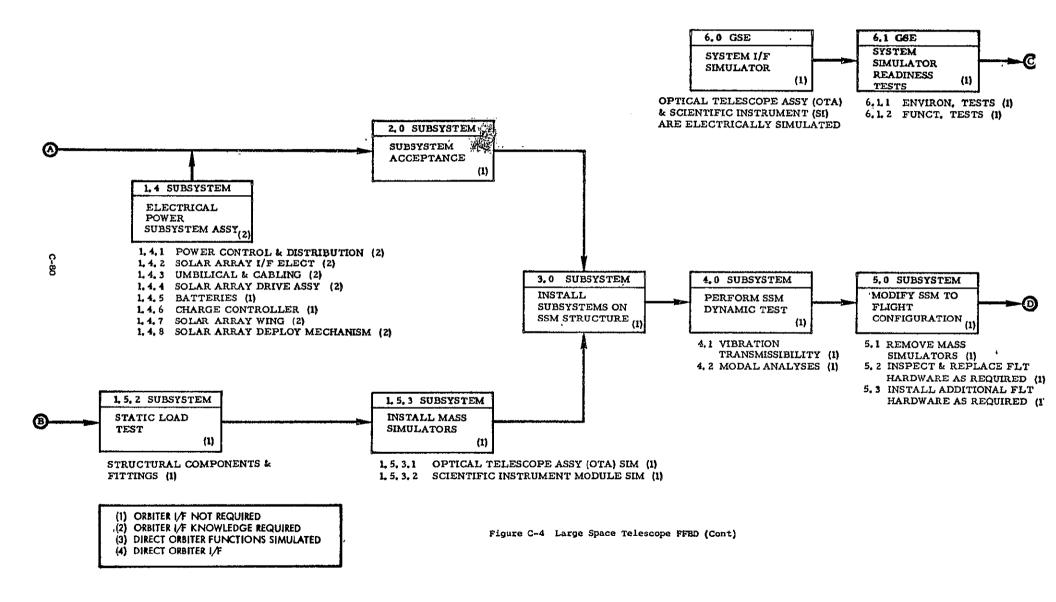
### Table C.3 (Cont) MODULE WITH PALLETS (SPACELAB)

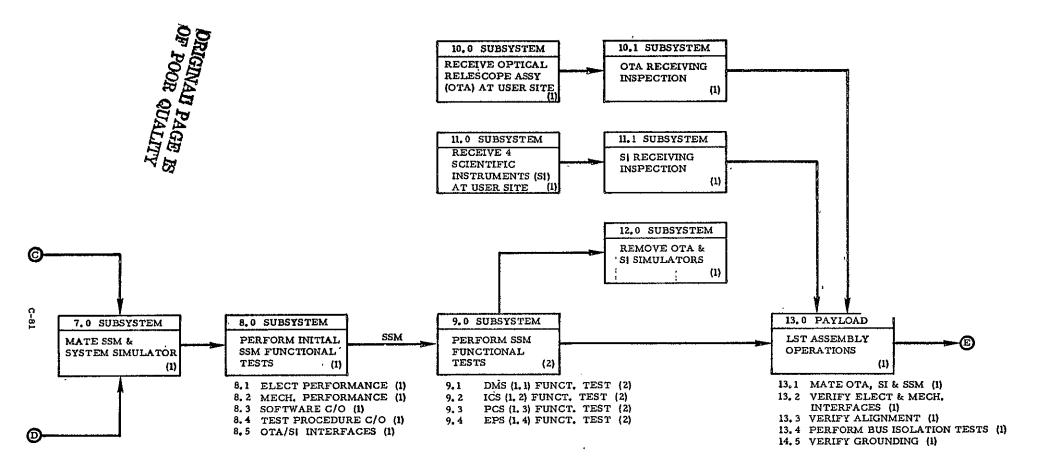
ı		,		n:h	IRFACES	i į	CH*KCOU EQUIPM	t/test ent	, ,	
BLOCK NO.	Precription	* I/F		eline Pation	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITY ORIENTED	. , GST SIM.	: IVE	SPRCIAL PACILITY	RIPMATECT
τ̃9∙0	Launch operations	NA	Launch	i site	·	x	,		ŧ	· ,
79.1	Clear pad	NA.	·	1	r	х	¥		ì	1
79.2	Stend by	NA	,			x		· · ·		
79.3	Countdown	NA NA	5	,	1	-; x	•	1		!
79.4	Liftoff	, NA	·	,	i. a	' x			:	:
80.0	Landing operations	, NA	,	;		x		}	1	,
80.1	De-mervice	, NA	į			l x		}		,
80.2	Safing operations	, NA	'	l .		¹ x		j.	4	,
80.3	Onen rayload doors	NA '	ì	-	i	x				•
81.0	Disconnect tunnel from orbiter	NA ·	,		!	x	· ·	; -	;	·
82.0	Remove cargo	(4)	,	·	,	÷х		:		,
83-0	Remove P/L station manels & utility harness	(4)				· x		1 .		
84.0	Move to 0 & C facility	(1)	;	1	:	! x	٠	:	í	1.
85.0	Move to O & C facility	(1)	<u> </u>	`		х				
86.0	Perform receiving inspection	(1)				x		,	1	
87.0	Install in spacelab workstand	(1)				·x		· ·		;
88.0	Perform preliminary interior inspection	(1)	,		,	х		} ,	ţ	, .
89.0	Perform nost-flight subsystem checkout Remove EM/SM shells & FS equipment	(2)		,		. х х				
	* 1 ORBITEP I/F NOT REQUIRED  OPBITEP I/F K'IOVLEDGE REQUIRED	1		<u>ا</u>		URSITER N° SIMULATED OFFITER 1/F	,			

Table C.3 (Cont). MODULE WITH PALLETS (SPACELAB)

	PESCRIPTION		TPT!	RFACES		CHPKCOU EQUIPM	tytest Ent		RFMAPK/I
BI.OCK NO.		* I/P	raseline Location	OPTION 1 USBR ORIENTED	OPTION 2 LAUNCH SITF ORIENTED	GST SIM.	IVE	ADECIPITA ADECIAT	
21.0	Remove pellets/experiments	(1)	Iaunch aite		x				
92.0	Remove racks/experiments	(1)			х	,	<u> </u>	<u> </u>	·
93.0	Demate aft bulkhead	(1)			х				
94.0	Demate tunnel	(1)			х				
95.0	Demate experiments from pallet	(1)			x				
96.0	Demate experiments from racks	(1)		····	x				
97.0	Service aft bulkhead as required	(1)			х		<u> </u>		
98.0	Service tunnel as required	. (1)		•	_х				
99.0	Return experiments to experimenter	(1)		x					
100.0	Return experiments to experimenter	(1)		x			'		
101.0	Refurbish pellets	(1)		·	х				
102.0	Refurbish racks	(1)			x				المراجع والأراج والمراجع والمراجع والمراجع والمراجع والمراجع والمراجع والمراجع
103.0	Store aft bulkbead	(1)		,	х				
104.0	Store tunnel	(1)			х				······································
105.0	Transfer pallets	(1)		····	x				
1.06.0	Transfer racks	(1)		<del></del>	х		<u> </u>		
107.0	Transfer aft bulkhead	(1)		,	х	<del></del>			
108.0	Transfer tunnel	(1)	♦		х.				
	* (1) ORBITEP I/F NOT REQUIRED			DIRFCT FUNCTION	ORSITER ON" SIMULATED		<u> </u>		







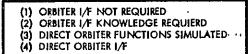
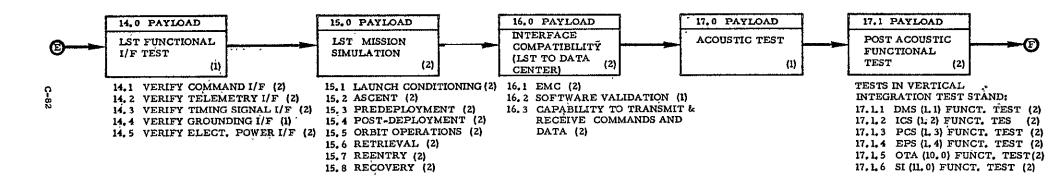


Figure C-4 , Large Space Telescope FFBD (Cont)



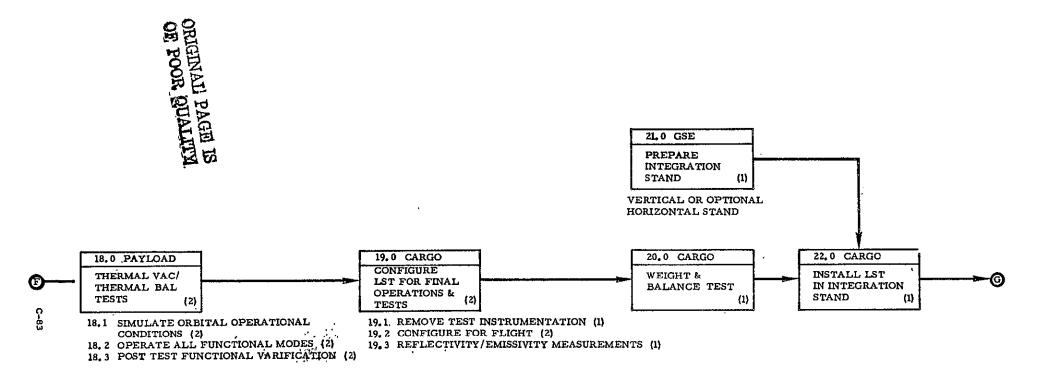
(1) ORBITER I/F NOT REQUIRED

Figure C-4 Large Space Telescope FFBD (Cont)

<sup>(2)</sup> ORBITER I/F KNOWLEDGE REQUIRED

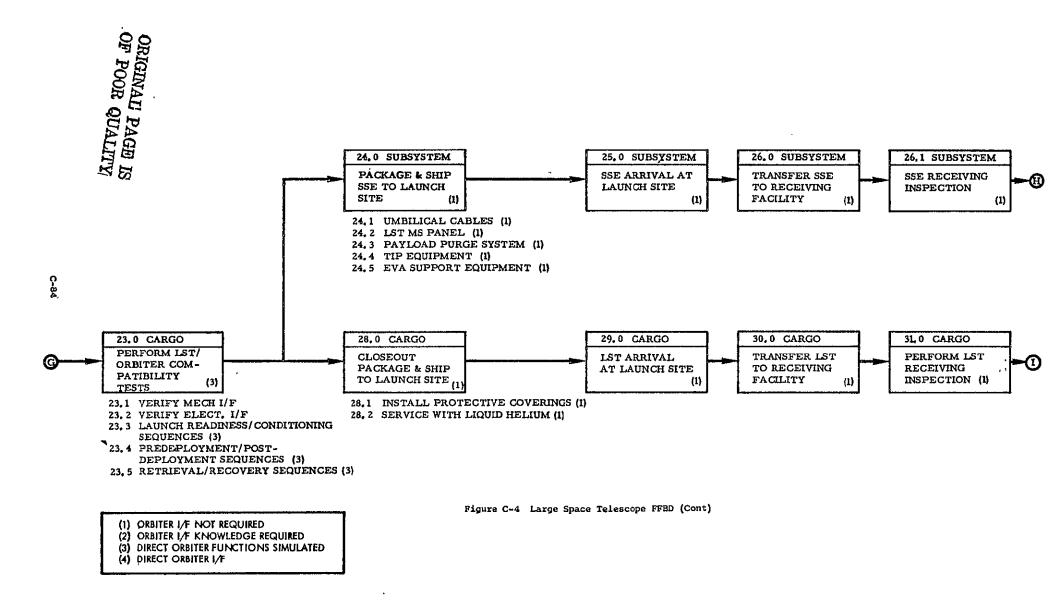
<sup>(3)</sup> DIRECT ORBITER FUNCTIONS SIMULATED

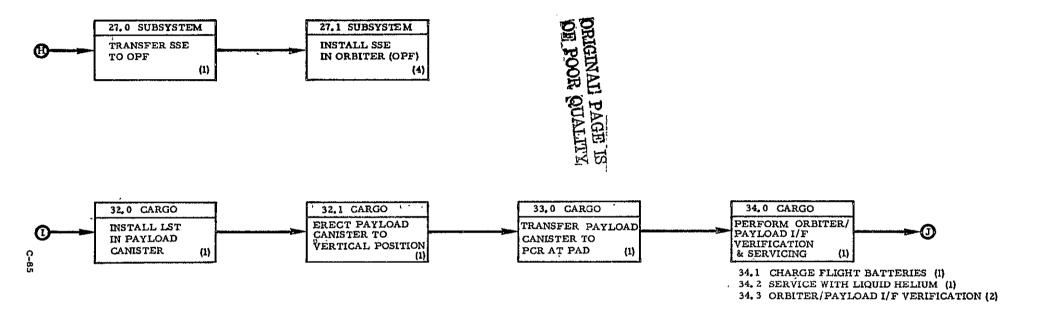
<sup>(4)</sup> DIRECT ORBITER I/F



- (1) ORBITER I/F NOT REQUIRED
  (2) ORBITER I/F KNOWLEDGE REQUIRED
- (3) DIRECT ORBITER FUNCTIONS SIMULATED
  (4) DIRECT ORBITER I/F

Figure C-4 Large Space Teleccope FFBD (Cont)





- (i) ORBITER I/F NOT REQUIRED
- (2) ORBITER I/F KNOWLEDGE REQUIRED
  (3) DIRECT ORBITER FUNCTIONS SIMULATED
  (4) DIRECT ORBITER I/F

Figure C-4 Large Space Telescope FFBD (Cont)

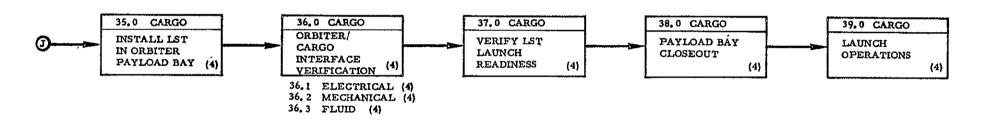


Figure C-4 Large Space Telescope FFBD (Cont)

- (1) ORBITER I/F NOT REQUIRED
  (2) ORBITER I/F KNWOLEDGE REQUIRED
  (3) DIRECT ORBITER FUNCTIONS SIMULATED
  (4) DIRECT ORBITER I/F

OF POOR QUALITY

### Table C.4 LET INTERFACE CHECKOUT MATRIX

				Teria	RPACES		CHIPKCOU	r/Test '		,	
BI.OCK	DESCRIPTION	* I/P	BAI L	Seline OCATION	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITH ORIENTED	098 r 91M.	ive	'SPECIAL FACILITY		REMARKS
1.0	Service system module (SSM) buildup	, <b>(</b> 2)	Use	r ;	х,	1				Various	sources
1.1	Data management subsystem assembly	, (S)	,	,	x		f	ŀ		<u>,                                     </u>	l
1.1.	l Pata interface units	, (S)			, X						1
1.1.	2 Computer interface unit	(5)			<u>x</u>	i		:			
1.1.	3 Data exchange unit	(5)	;		i x			,			
1.1.	4 Auxiliary memory unit	, (5)			×	1	;		,	,	
1.1.	5 Command decoder	(1)		,	• х						,
1.1.	S Power switch & decoder	(1)			x						
1.1.	7 Central processor	(2)	·	,	` x				,	,	
1.1.	3 Input/output processor	(2)			· x			,		:	
1.1.	9 Main memory	(1)			х	,	,	·			
1,1,	10 Gomputer power converter	(1)		,	· x			,			
1.1.	. Oscillator	(1)			, x			;		<u> </u>	
1.1.	12 Tape recorders	(1)			Х						
1.2	Instrumentation & communication subsystem assy	. (2)			. х						
1.2.	L Transponder	(1)			ж			٠	`		1
1.2.	Puplexers	(1)		ì		` .		,			
1.2.	3 Circulator .	(1)		,	х	<u> </u>				<u> </u>	
1.2.		(5)		<u>',</u>	X DIRFTER	ORSITER				,	4
	(1) ORRITEP I/F NOT REQUIRED				9 FUICTIO	ON CIMULATED					
	OPBITER I/F KNOWLEDGE REQUIEND			C	DIRECT	OPPITER 1/F					

Table C.4 (Cont) LST INTERFACE CHECKOUT MATRIX

	· · · · · · · · · · · · · · · · · · ·		INT	INFACES		CHPKCOU.	/Test			
BLOCK	precedention	* I/F	Baseline Location	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITE ORIENTED	G97 SIM.	IVE	SPECIAL PACILITY	HINAPKS	
1.2.5	Antenna drive and bosm	(5)	User	х					Various sources	
1.2.6	Orni antenna	(5)		×						
1.2.7	Power amplifier	(1)		х				·		
1.2.8	HF switches	(1)		x				<u> </u>		
1.2.9	High gain antenna	(2)		х	`			<u> </u>		
1.3	Pointing control subsystem assembly	(2)		x						
1.3.1	Fixed star tracker	(2)		x				<u> </u>		
1.3.2.	Fine gyro assembly	(5)		х				ļ		
1.3.3	Magnetic torquer	(1)		X		•		<b></b>		
1.3.4	Reaction wheels	(1)		х			·			
1.3.5	Retrieval electronics	(5)		x					 	
1.3.6	Coarge sun sensor	(1)		х						
1.3.7	Mignetometer	(1)		x				<u> </u>		
1.3.8	Retrieval rate gyros	(2)		х				<u> </u>		
1.4	Electrical power subsystem assembly	(5)		x				<b>.</b>		
1.4.1	Power control & distribution	. (2)		x				<b></b>		
1,4,2,	Solar array electrical interface	(5)		x				<b>1</b>		
1.4.3	Unbilical & cabling	(5)		x	<b> </b>			<b></b>		
1,4,4	Solar array drive assembly	(5)	<u> </u>	X SV ULRPUT	ORSPIER	·			<u> </u>	
	* (1) ORBITEP I/F NOT REQUIRED		(	3) FUNCTI	CSTATUMES TWO					
	OPBITTER I/F KYOAT,EIGH REGUIERD			TYPRICE (T	OPFITER I/F				-	

# Table C.4 (Cent) LST INTERFACE CHECKOUT MATRIX

	·	, , , , ,	Triff	TRPACES		) CHEKCOU EQUIPM	r/test ent	1	:	
BIOCK NO.	description	knowledge + 1/b	BASELINE LOCATION	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITF ORIENTED	G9%	IVE	SPECIAL PACILITY		
1,4,5	Batteries	(1)	Vser.	х				<u> </u>	Various sources	
1,4,6	Charge controller	(1)	,	x		,	•	<u>'</u>		
	Sol ar-array wing	(5)	·	х	,					
1.4.8	Solar array deploy mechanism	(2)		х						
1.5	Structural subsystem	(5)		х						
1.5.1	Structural subsystem assembly	(1)		x					`	
1.5.1.1	Fittings	(1)		х			· · · · · ·	<u>'</u>		
1.5.1.2	Intches	(I)		х						
	Aperture doors	(1)		х				ļ		
1.5.1.4	Panels	(1)		х				ļ		
1.5.1.5	Insulation	(1)		х.					V	
1.5.2	Static load test of struct. components & fitt.	(1)		х				ļ		
1,5,3	Install mass simulators	(1)		х		х		ļ	·	
1.5.3.1	Cotical telescope assy (CTA) mass simulator	(1)		х		х		ļ		
1.5.3.2	Scientific instrument module mass simulator	(1)		х	<u> </u>	х	ļ	ļ		
							4	ļ	<u> </u>	
2.0	Subsystem acceptance	_u_	*	x			<u> </u>	<u> </u>	<u>, , , , , , , , , , , , , , , , , , , </u>	
· · · · · · · · · · · · · · · · · · ·				,				<b></b>		
	* (1) ONBITEP T/F NOT REQUIRED			3) <u>presen</u>	ORSITER	<u> </u>		<u>.                                    </u>		
	OPBITER I/F KNOWLEDGE REQUIFED		(		OPPITER I/F					

Table C.4 (Cont) LET INTERFACE CHECKOUT MATRIX

			- III	RFACES		CHPRCOUT EQUIPM	/Test HT			
BLOCK	Description .	* 1/F KNOWLEDGE	raseline Location	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITF ORIENTED	GSE SM.	IVE	SPRCIAL FACILITY	REMARKS	
8.0	Perform initial SSM functional tests	(1)	User	х	· x	хх	X		Rither site	
8.1	Electrical performance	(1)		х	х	x	x			
8.2	Mechanical performance	(1)		х	X	x	x			
8:3	Software checkout	(1)		х	х	X	X			
8.4	Test procedures checkout	(1)		х	х	х -	х			
8.5	OTA/SI interfaces	(1)	-	х	х	х	хх		-	
9.0	Perform SSM functional tests	(2)		х	х	<u> </u>	Х			
9.1	DMS functional test	(5)	•	, X	χ	X	х			
9.2	ICS functional test	(2)		х	х	x	х	[i		
9.3	FCS functional test	(2)		Ř	х	X	X			
9.4	EPS functional test	(5)	·	х	χ	х	х		<b>*</b> · · · · · · · · · · · · · · · · · · ·	
10.0	Receive OTA at user site	(1)		x				`		
10.1	OTA receiving inspection	(1)		х						
11.0	Receive 4 scientific instr. at user site	(1)		( X						
11.1	ST receiving inspection	(I)	¥	x					•	
<del>,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,</del>	ORBITEP I/F NOT REQUIRED		. (	3) DIRFCT FUNCTI	ORSITER ON CIMULATED		<u> </u>	<u>L</u>		
	ODBITED I/F KNOWLEDGE REQUIEST		(	TOFRIC (I	OPPITER I/F					

Table C.4 (Cont) LST INTERFACE CHECKOUT MATRIX

			TI:T	BRFACES		CHPKCOVI EQUIPME	r/Test NT		
BLOCK	Prechipation	* I/F KHOWLEDGE	Baseline Location	OPTION 1 USBR ORIENTED	OPTION 2 LAUNCH SITF ORIENTED	GSW M.	IVE	SPECIAL FACILITY	RPMADECT
12.0	Remove OTA & SI simulators	(1)	User	х.	х				Either site
13.0	IST assembly overations	(1)		х	х			,	
13.1	Mate OTA, SI & SSM	(1)		х	x				
13.2	Yerify electrical & mechanical interfaces	(1)	,	х	хх				
13.3	Verify alignement	(1)	,	х	x	·			
13.4	Perform bus isolation tests	(1)	1	х	х		•		
13.5	Verify grounding	(1)		x	х			· · · · · ·	•
14.0	LST functional interface test	(2)	,	х	Х	х	x		Trade study
14.1	Verify command interface	(5)	k 4	x	х	X	х		
14.2	Verify telemetry interface	(5)	,	x	х	х	х		
14.3	Verify timing signal interface	(5)		х	χ	х	х		
14.4	Yerify grounding interface	(1)		х	х.	x	x		
14.5	Verify electrical power interface	(2)		х	х	х	х.		
- !	;	1			,		,	2	
15.0	LST mission simulation	(5)		Х	x	X t	x <sup>1</sup>	-	
15.1	Launch conditioning	(5),		<u> </u>	x ,	x '	х і	!	
15.2	Agcent	(2)		X	х	x :	x	,	<u> </u>
	Pre-deployment	·(2)	. ♦	x	х,	х	х		. ♦

Table C.4 (Cont) LST INTERPACE CHRCKOUT MATRIX

···· <u>}</u>			i i i i i i i i i i i i i i i i i i i	REPACES	· ·	CH*KCOV EQUIPM	/Test ent		
BLOCK NO.	description	* I/P NNOWLEDGE	Baseline Location	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITE ORIENTED	GSE GM.	IVE	SPECIAL PACKLITY	REMARKS
15.4	Pogt-deployment	(2)	Vser	х	x	х	х		Trade study
15.5	Orbit operation	(5)		х	х .	. x	х		<u> </u>
15.6	Retrieval	(2) '		·x	х	Х	x		
15.7	Recentry	(2)		x	х	х	Х	`	
15.8	Recovery	(5)		×	х	х	. x		
16.0	LST to data center interface compatibility	(5)		х	х	X	x		
16.1	емс	(3),		x	х	х	х		
16.2	Software validation	(1)		х	. x	Х	Х	<u> </u>	
16.3	Capability to transmit & receive commands & dat	. (5)		<u>x</u>	х	хх	X		
17.0	Acoustic test	(1)		х	. x			. х	
17.1	Post-accustic functional test	(2)		х	х	х .	X	1	,
17.1.1	DMS functional test	(2)		_ x	х	х	x	<u> </u>	
17.1.2	TCS functional test	(8)		X 1	<u>, x</u>	х	x	<u> </u>	,
17.1.3	FCS functional test	(2)	ļ	x	, x	<u> </u>	, х	1	
17.1.4	FPS functional test	(5)		Х	<u>x</u>	<u> </u>	х		
17.1.5	OTA functional test	(5)		x	х	x	x	<b> </b>	
17.1.6	SI functional test * (1) ORBITEP I/F NOT REQUIRED	(5)	(	X DIRFUT FUNCTI	X URSITER ON SIMULATED	Х	х	<u> </u>	<u> </u>
	S OPPLITED I/P KNOWLEDGE REDUIERD				OPPITEP I/F				

Table C.4 (Cont) IST INTERFACE CHECKOUT MATRIX

			I?"M	IRFACES		CHPKCOU EQUITM	t/test ent		
BLOCK, NO.	DESCRIPTION	* 1/F	Baseline Location	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITF ORIENTED	GEM.	IVE	SPFCIAL FACILITY	RINATEGS
18.0	Thermal wac/thermal balance test	(2)	User	х ′	x			ж	Trade utudy
18.1	Simulate orbital operational conditions	(5)		х	x			x	
18.2	Operate all functional modes	(5)		х,	X	x	х	x.	
18.3	Post-test functional verification	(5)		х	х	хх	х		
19.0	Configure LST for final operations & tests	(2)		х.	x ·				Fither.site
19.1	Remove test instrumentation	(1)		х	х		<u> </u>		
19.2	Configure for flight	(5)		х	<u> </u>				
19.3	Reflectivity/emissivity measurements	<u>(1)</u>		x	х		·		
20.0	Weight and balance test	(1)		х	x			X	•
21.0	Frepare vertical or optional horizontal integration stand	(1)		x	х		х	х .	Assuming IVE is adaptable to ass
22,0	Install LST in integration stand	(5)		x	х				
23.0 `	Perform LST/orbiter compatibility tests	(3)		x	x	<u>, , , x </u>	x		Trade study
23.1	Verify mechanical interface	(3)		х	х	, <b>X</b>		1	,
23.2	Yerify electrical interface  * (1) ORBITEP I/F NOT REQUIRED	(3)	•		X URBITER ON CIMULATED	x	х	<u> </u>	<u> </u>
	OPBITED I/E KNOWLEDGE REQUIFED		Ò	_	OPPITER I/F				

				Irti	IRPACES '	٠ ، } ,	, СНРКСООТ	i/Test	.,.,	
OF POOR PAGEN	prechiption '	* I/F KNOWLETGE	BAS	eline Cation	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITE ORIENTED	)	j i.	SPECIAL PACILITY	( REMARKO
23.3	Launch readiness/conditioning sequences	(3)	Ué	er !	'x	'x	x	<b>x</b> :	:	Trade study
23.4	Pre-deployment/post-deployment sequences	(3)			х '	х	х	x		
23.5	Retrieval/recovery sequences	(3)			'x	·x	х	·x	ş	<b>+</b>
		,				,		;		,
24.0	Package & ship SSE to Launch site	(1)		1	x	,		1		
24.1	Umbilical cables	· (1)	<u> </u>	i	<sup>i</sup> x	i		,		
24.2	LET MS panel	(1)	<u> </u>	;	x.			:		<u> </u>
24.3	Payload purge system	(1)		· t	Х			ı	•	
2h .h	TTP equipment	(1)			, X	!	`	:	<u> </u>	
24.5	EVA support equipment	(1)		<u> </u>	; x		<u> </u>			
		'	,		1			;		·
25.0	SSE arrival at Launch site	(1)	Launc	h site		; x				
			,	<u> </u>	· '	,	<u> </u>			
26.0	Transfer SSE to receiving facility	(1)	ļ		'.	х		:		
26.1	SSE receiving inspection	(1)	<u> </u>	ļ	' 	х				
						,				<u> </u>
27.0	Transfer SSE to OPF	(1)	<u> </u>	ļ	<u> </u>	, x	<u> </u>	<u>'.</u>	х	
27.1	Install SSE in orbiter (OFF)	(4)		Ÿ		x		,	x	,

Table C.4 (Cont) LET INTERVACE CHECKOUT MATRIX

•••			ni	RPACES		CHPKCOU EQUIPM	t/Test ent		
BLOCK NO.	Přscription	* 1/F	BASELINE LOCATION	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITH ORIENTED	G9% SIM.	IVE	SPPCIAL FACILITY	RPMATECS
28.0	Closecut, package & ship to Launch site	(1)	User	x	,				
28.1	Install protective covers	(1)		x	,				
28,2	Service with liquid helium	(1)	₩	<u>x</u>					
89.0	LST arrival at Launch site	(1)	Inunch site		х		,		
30.0	Transfer Lat to receiving facility	(1)			х			х	h;
31.0	Perform LST receiving inspection 1	(1)			x			x	
32.0	Install payload in canister	(1)			х	,			
32.1	Erect psyload canister to vertical position	(1)		,	<u>x</u>				
33.0	Transfer Payload canister to PCR at pad	(1)	,	1,7	х			х	
34.0	Perform orbiter/payload interface verification and servicing	(1).(2)		x	- x	х	X		Both sites; trade study
34.1	Charge flight batteries	(1)		х	х	····			
34.2	Service with liquid helium	(1)		х	х		` 		•
74. 국	Orbiter/mayload interface verification  * 1 ORBITED I/F NOT REQUIRED	(2)	•	X DIRPOR FUNCTIO	X ORSITEN ORSITEN ORSITEN	X	х		<u> </u>
	OPPITED I/F KNOWLEDGE REQUIESD			DIRWT	OPPITER I/F				

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Table C.4 (Cont) LET INTERFACE CHECKOUT MATRIX

	T 38		TET	REFACES	į	CHPKCOU EQUIPM	T/TEST			
BLOCK NO.	PESCRIPTION :	* I/F KNOWLETGE	BASELINE LOCATION	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITE ORTENTED	G98 NM.	IVE '	PACILITY PACILITY	REMARKS	
35.0	Install LST in orbiter bay	<b>(</b> 4)	Launch site	,	х					
					,	` :				
36.0 ·	Orbiter/cargo interface verification	<u>'(4)</u>	<u> </u>		x					
36.1	Electrical	(4)			x	j				
36.2	Mechanical	(4)			x	:	<u> </u>			
36.3	Fluid	(4)		,	x	,				
37.0	Verify IST launch readiness	(4)			х	i	,		:	
38.0	Payload hav closeout	(4)			х	1				
39.0	Launch operations	(4)			<u>x</u>	î,	·			
	1									
<del></del>					,					
						i	·	ŕ		
·	ORRITEP I/F HOT REQUIRED		<u> </u>	3 DIRPOT FUNCTI	ORSITER ON SIMULATED	l	<u></u>	<u> </u>		
	© OPPITED I/F KNOVIETCH REDUTERN		(	TYPRICE (T	OPPITER I/F					

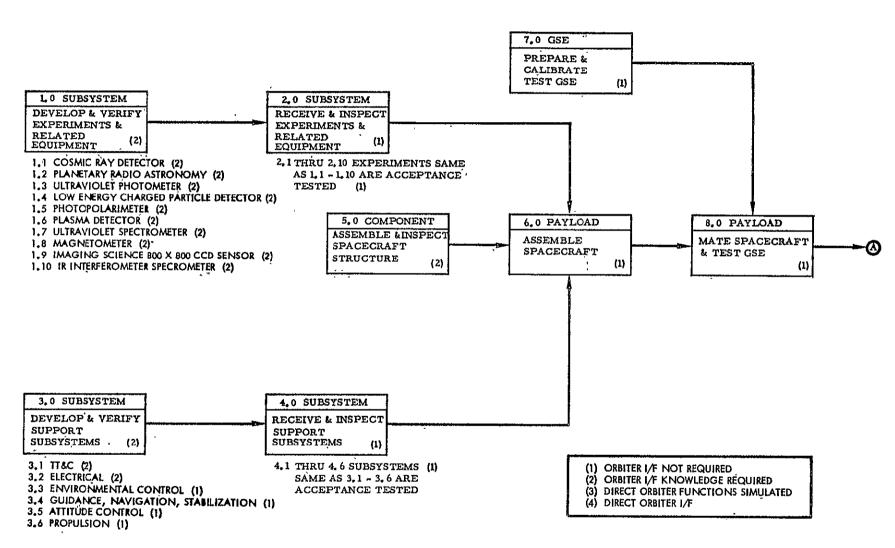
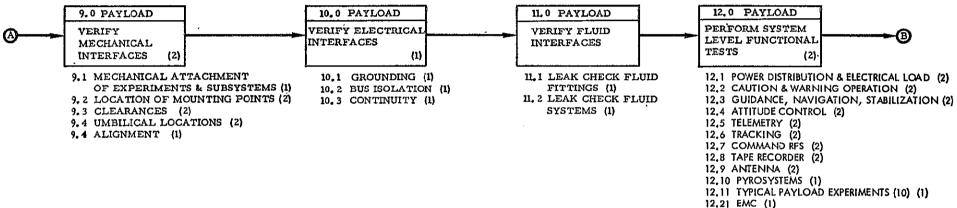


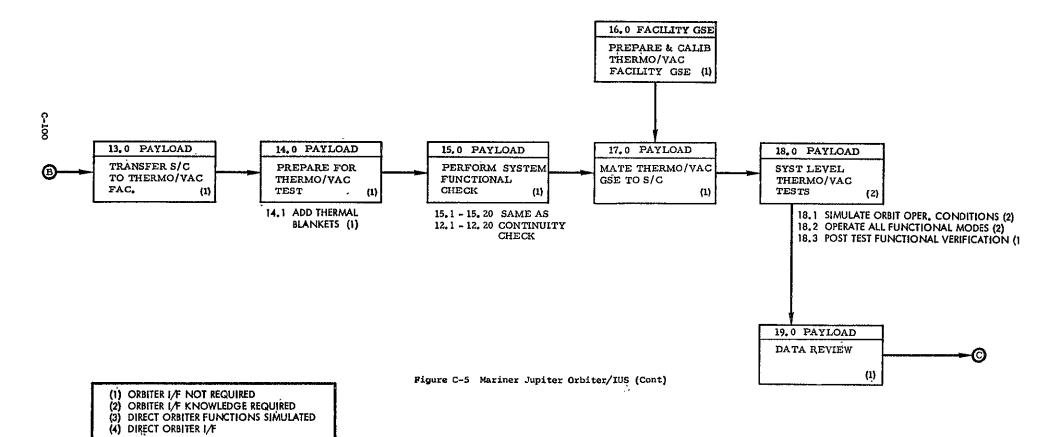
Figure C-5 Mariner Jupiter Orbiter/IUS

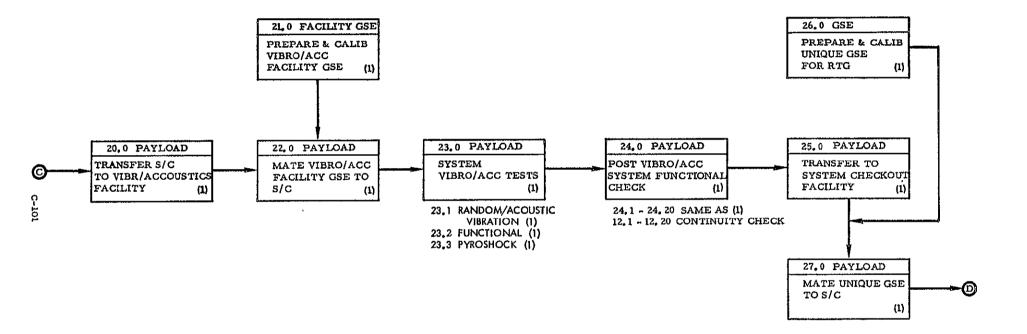




- (1) ORBITER I/F NOT REQUIRED
- (2) ORBITER I/F KNOWLEDGE REQUIRED
  (3) DIRECT ORBITER FUNCTIONS SIMULATED
- (4) DIRECT ORBITER I/F

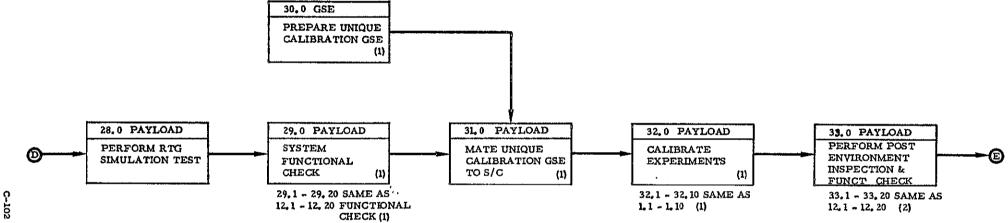
Figure C-5 Mariner Jupiter Orbiter/IUS (Cont)





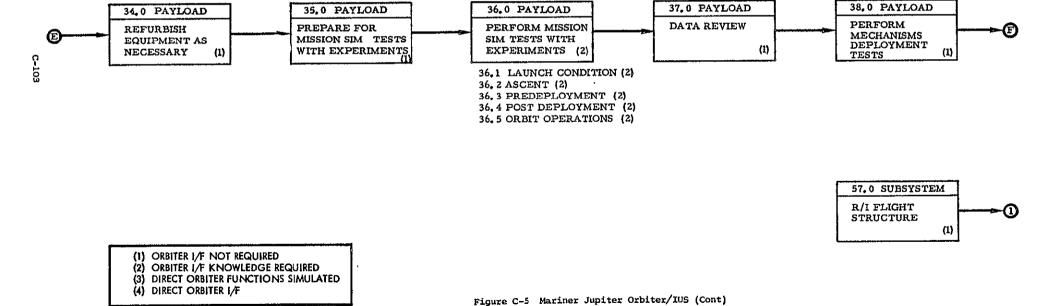
- (1) ORBITER I/F NOT REQUIRED
  (2) ORBITER I/F KNOWLEDGE REQUIRED
- (3) DIRECT ORBITER FUNCTIONS SIMULATED
- (4) DIRECT ORBITER 1/F

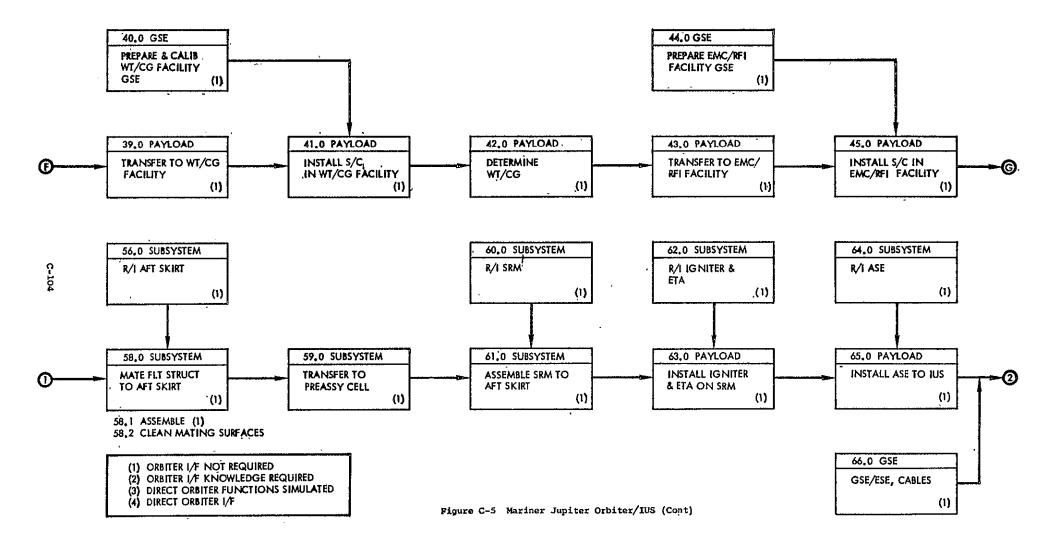
Figure C-5 Mariner Jupiter Orbiter/IUS (Cont)

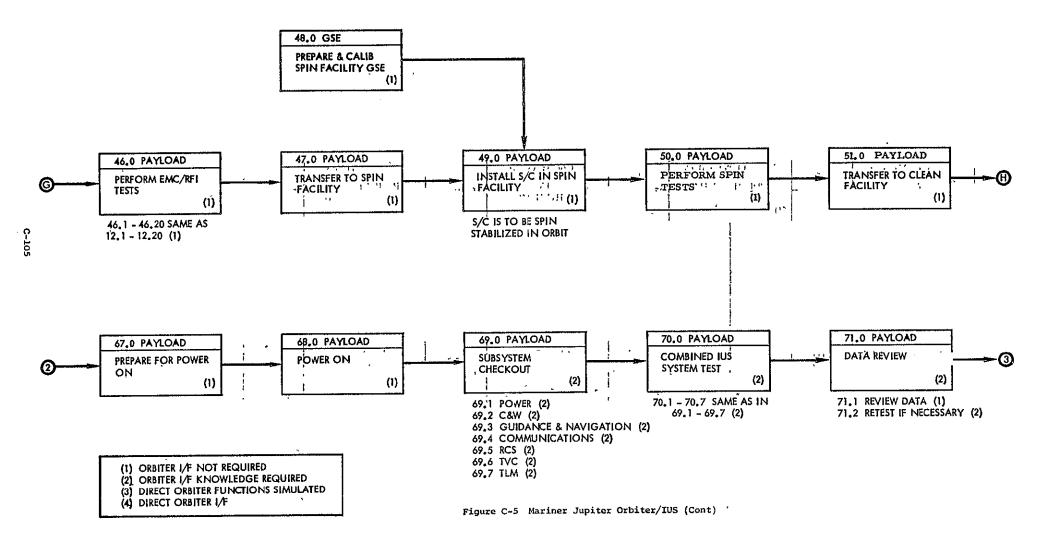


- (1) ORBITER I/F NOT REQUIRED
  (2) ORBITER I/F KNOWLEDGE REQUIRED
  (3) DIRECT ORBITER FUNCTIONS SIMULATED
  (4) DIRECT ORBITER I/F

Figure C-5 Mariner Jupiter Orbiter/IUS (Cont)







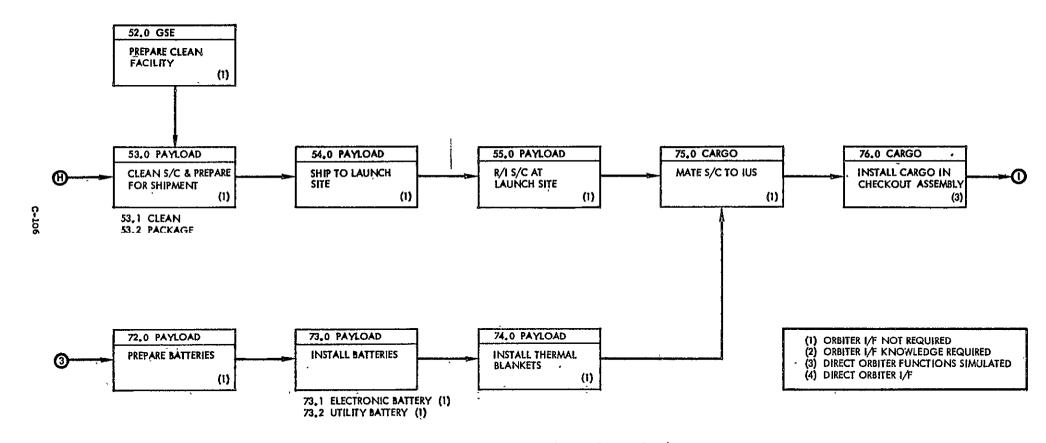
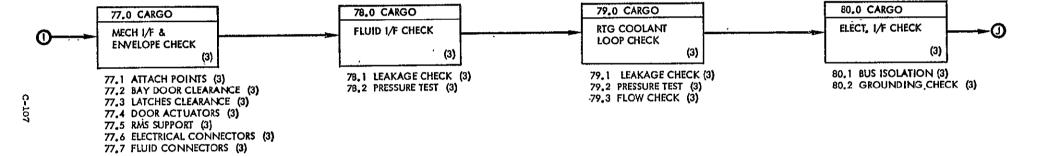
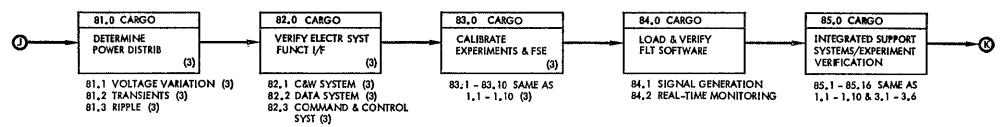


Figure C-5 Mariner Jupiter Orbiter/IUS (Cont)



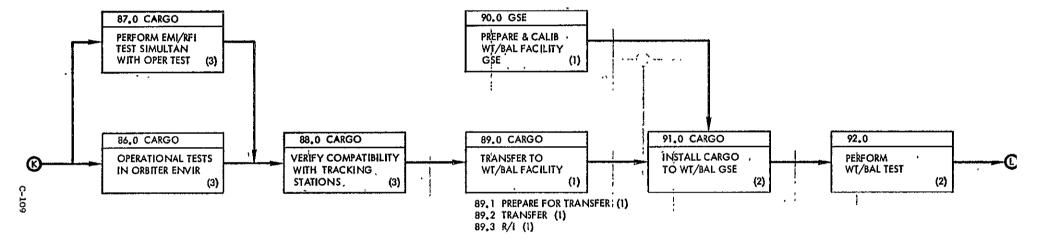
- (1) ORBITER I/F NOT REQUIRED
  (2) ORBITER I/F KNOWLEDGE REQUIRED
  (3) DIRECT ORBITER FUNCTIONS SIMULAT
  (4) DIRECT ORBITER I/F
- DIRECT ORBITER FUNCTIONS SIMULATED

Figure C-5 Mariner Jupiter Orbiter/IUS (Cont)



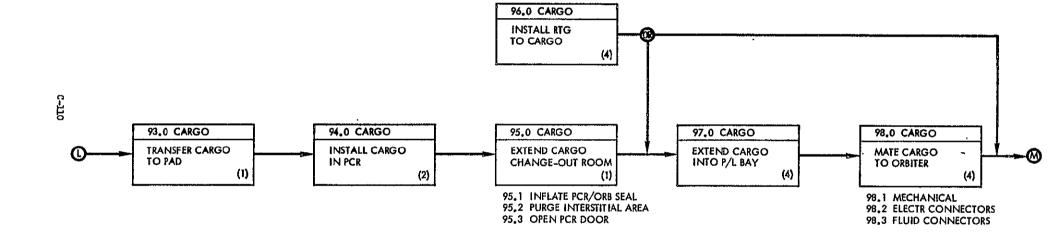
- (1) ORBITER I/F NOT REQUIRED
  (2) ORBITER I/F KNOWLEDGE REQUIRED
  (3) DIRECT ORBITER FUNCTIONS SIMULATED
  (4) DIRECT ORBITER I/F

Figure C-5 Mariner Jupiter Orbiter/IUS (Cont)



- (1) ORBITER I/F NOT REQUIRED
  (2) ORBITER I/F KNOWLEDGE REQUIRED
  (3) DIRECT ORBITER FUNCTIONS SIMULATED
  (4) DIRECT ORBITER I/F

Figure C-5 Mariner Jupiter Orbiter/IUS (Cont)



- (1) ORBITER I/F NOT REQUIRED
  (2) ORBITER I/F KNOWLEDGE REQUIRED
  (3) DIRECT ORBITER FUNCTIONS SIMULATED
  (4) DIRECT ORBITER I/F

Pigure C-5 Mariner Jupiter Orbiter/IUS (Cont

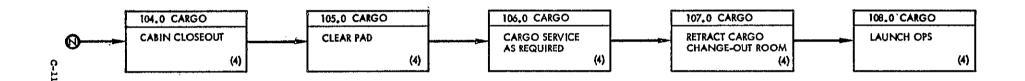


100.3 INSTALL & VERIFY MECHANICAL

LATCHES & RETENTION DEVICES (4)

(1) ORBITER I/F NOT REQUIRED
(2) ORBITER I/F KNOWLEDGE REQUIRED
(3) DIRECT ORBITER FUNCTIONS SIMULATED
(4) DIRECT ORBITER I/F

Figure C-5 Mariner Jupiter Orbiter/IUS (Cont)



- (1) ORBITER I/F NOT REQUIRED
  (2) ORBITER I/F KNOWLEDGE REQUIRED
  (3) DIRECT ORBITER FUNCTIONS SIMULATED
  (4) DIRECT ORBITER I/F

Figure C-5 Mariner Jupiter Orbiter/IUS (Cont)



Table C.5 MJO/IUS

INTERPACE CHECKOUT MATRIX

			Irri	TRPACES		CHFKCOU	/TEST		REMARKS	
BLOCK NO.	description	* I/F	BASELINE LOCATION	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITF ORIENTED	G98 SIM.	IVE	SPECIAL PACILITY		
1.0	Develop & verify experiments & related equipment	(a)	User	х					Various sources	
1.1	Cosmic ray detector	(2)		x						
1.2	Planetary radio astronomy	(2)		х						
1.3	Ultraviolet photometer	(5)		` x						
1.4	Low energy charged particle detector	(2) .		х						
1.5	Photopolarimeter	(5)		х			* '			
1.6	Plasma detector	(2)		х						
1.7	Ultraviolet spectrometer	(5)		х						
1.8	Magnetometer	(2)		x						
1.9	Imaging science 800x800 CCD sensor	(2)		х		*				
1.10	IR interferometer spectrometer	(5)		<u> </u>						
5.0	Receive & inspect experiments & related equip.	(1)		· ×						
2.1	2.1 through 2.10 experiments same as 1.1 through 1.10 are acceptance tested	(1)		х						
3.0	Develop & verify support subsystems	(5)		х .						
3.1	TT & C	(2)		х		t				
3.2	Electrical	(5)		×	,			<u> </u>	<u> </u>	
3.3	Environmental control	. (1)		х.				<u> </u>	<u> </u>	
3.4	Guidance, navigation & stabilization .	(1)	<u> </u>	. х	ORSITER	 	<u></u>	1	<u> </u>	
	* (1) ORBITEP I/F NOT REQUIRED  OPETTEP I/F KNOWLEDGE REQUIRED		.(	FUNCTI	ONT CIMULATED					

Table C.5 (Cent) MJO/IUS INTERPACE CHECKOUT MATRIX

<del></del>			i Inti	HPACES	· ` ` ` `	CHFKCOU EQUIPM	r/test ent	<u> </u>	
вьоск но.	PRSCRIPTION	* I/P KNOWLEDGE	Baseline Location	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITF ORIENTED	G9R SIM.	IVE	SPPCIAL FACILITY	RMARKO
3.5	ttitude control	(1)	'User	х					Various sources
3.6	Provul*ion	(1)		х					<u>'</u>
4.0	Receive & inspect support subsystems	(1)		х					
3.1	4.1 through 4.6 subsystems same as 3.1 through 3.6 are acceptance tested	(1)		x					
5.0	Assemble & inspect spacecraft structure	(2)		Х					١
6.0	Assemble spacecraft	(1)		х	х				Either site; trade study
7.0	Prepare & calibrate test GSE	(1)		х					
8.0	Mate spacecraft & test GSE	(1)		х	x				
9.0	Verify machanical interfaces	(2)		x	х .				
9.1	Mechanical attachment of experim. & subsystems	(1)		х	x				
9.2	Location of mounting points	(2)		x	×	<u> </u>			
9.3	Clearances	(5)		х	x				
9.4	Umbilical locations	(5)		x	x				
9.5	Alisment -	<u>(1)</u>		X	x			<u> </u>	
10.0	Verify electrical interfaces	(1)		x	х		·	ļ	
10.1	Grounding	(1)		х	x				
10.2	Bus isolation	(1)		х	х			1	
10,3	Continuity  * (1) ORRITEP I/F NOT REQUIRED	(1)	*	X DIRPOR	X ORSITER N° CIMULATED		<u> </u>	1	<u> </u>
	OPBITEF I/F KNOTLENGE REQUIFED			_	OPPITER I/F				



Table C.5 (Cont) MLO/IUS

INTERFACE CHECKOUT MATRIX

			n:h:	REACES		CHFKCOU! EQUIPM	r/test int			
BLOCK NO.	precription	* I/F	Baseline Location	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITF ORIENTED	GSR SIM.	IVE	SPFCIAL PACILITY	REMARKS	
11.0	Verify fluid interfaces	(1)	User	Х	х				Either site: trade study	
11.1	Leakeheck fluid fittings	(1)		<u> </u>	х			<u> </u>		
11.2		(1)		X	Х					
12.0	Perform system level functional tests	(5)		х	х .	x	x	ļ		
12.1	Power distribution & electrical load	(5)		x	x	х	x			
12.2	Caution & varning overation	(5)		х	х	х	х			
12.3	Guidance, navigation & stabilization	(5)		х	х .	х	x	<u> </u>		
12.4	Attitude control	(5)		x	х	x	x	ļ		
12.5	Telemetry	(5)		х	, х	х	х	1		
12.6	Tracking	(5)	-	χ	х	х	х	<u> </u>		
12.7	Command RFS	(5)		X	х	Х	x			
12.8	Tane recorder	(8)		х	X	X	х			
12.9	Antenna	(2)		х	x	x	Х			
12.10	Pyrosystems	(1)		х	x	x	х	<u> </u>		
12.11	Typical payload experiments (10 total)	(1)		х	x	x	x	<u> </u>		
12.21 ·	емс	(1)		х	x	х	x	<u> </u>	<b>*</b>	
13.0	Transfer spacecraft to thermo/vac facility.	(1)		x ·	Х			x		
14.0	Prepare for thermo/vac tests	(1)		x	x			x		
14.1	Add thermal blankets	(1)	,	X DIRPORT	X ORSITER			х	<u> </u>	
	* (1) ORBITED I/F HOT REQUIRED  OPELTED I/F KIGALENCE REQUIFED		(	רבי דניוניזו -	OPPITEP I/F					

## Table C.5 (Cent) MJO/IUS INTERFACE CHECKOUT MATRIX

			ırt.	FRFACES		CHPKCOU	r/test ent		
BLOCK NO.	Pricription	* 1/F	Baseline Location	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITF ORIENTED	GSR SIM.	IVE	SPECIAL PACILITY	REMARKS
15.0	Perform system continuity check	(1)	User	x	x				,
15.1	15.1 through 15.20 same as 12.1 through 12.20 functional check	(1)		×	x			<u>.</u>	
16.0	Prepare & calibrate thermo/vac facility GSE	(1)		×	х .			х	
17.0	white thermo/was GSE to spacecraft	(1)		х	x			х	
18.0	Exstem level thermo/wac tests	(2)		x	x	<u>x</u> ,	х		
18.1	Simulate orbit operational conditions	(2)		х	. x	х	'x		
18.2	Operate all functional modes	(5)	,	. х	x	χ	х		
18.3	Post test functional verification	(1)		x	х	x	χ	;	
19.0	Onto review	(1)		x	х				Alternate site
30.0	Cransfer S/C to vibr/seconstic facility	(1)		х	x		,	x	
21.0	Prepare & calibrate vibro/acc. facility GSE	(1)		Х	x	,	,	х	
22.0	Mate wibro/acc facility GSE to spacecraft	(1)	,	х	х			х	
23.0	System vibro/accoustic tests	(1)	,	x	х			- x	
23.1	Random/accoustic vibration tests	(1)		x	х			χ	,
23.2	Functional check	(1)		x	х			х	,
_23,3	Pyroshock	(1)		х	×			x	•
	Post vibro/acc system continuity check 24.1 through 24.2 Osame as 12.1 through 12.20 functional check	(1) . (1)	<del>     </del>	X	х			Х	
	* (1) ORBITED I/E HOL BESTILLED			y function	ORSITER N° CIMULATED OPPITER I/F			<u> </u>	



Table C.5 (Cont) MIO/IUB

INTERFACE CHECKOUT MATRIX

			Tivit	INVACES	·	CHPKCOUT EQUIPM	r/Test NT		
BLOCK NO.	description	ENONTEDGE * I\k	Baseline Location	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITE ORIENTED	gsr sim.	IVE	SPECIAL PACILITY	REMARKS
25.0	Transfer to system checkout facility	(1)	User	X	×				
26.0	Prepare & calibrate unique GSE for RTG	(1)		x	x			<b>}</b>	Unique GSE
27.0	Nate unique GSE to spacecraft	(1)		X	<u>  x  </u>	X		· · ·	
28.0	Perform RTG simulation test	(1)		x	- x	X			9
29.0	Systems functional check	(1)		x	x	х	x	ļ	
29,1	29.1 through 29.20 same as 12.1 through 12.20 functional check	(1)		х	<u>  x  </u>	X	х		
30.0	Prepare unique calibration GSE	(1)		х	х				Unique GSE
31.0 -	Hate unique calibration GSE to spacecraft	(I)		х	x		<b></b>	ļ	
32.0	Calibrate experiments	(1)		<u> </u>	X	<del> </del>			
32.1	32.1 through 32.10 same as 1.1 through 1.10.	(1)		x	x		<u> </u>	ļ	
33-0	Perform post environmental insp. & funct. check	(5)		x	X	X	x		
33.1	33.1 through 33.2) same as 12.1 through 12.20	(5)		×	×	<u> </u>	x	ļ	
34.0	Refurbish equirment as necessary	(1)		x	x		<del> </del>	<del> </del>	,
35.0	Prepare for mission sim. tests with experiments	(2)	<u> </u>	x	<u> </u>		<del> </del>	<del> </del>	
36.0	Perform mission sim. tests with experiments	(5)		<u> </u>	Х	x	х	<del> </del>	Either site: trade study
36.1	Launch conditions	(5)		x	, x	х	х	<u> </u>	
36.2	Ascent	(5)		<u>x</u>	Х	x	x	_	
36.3	Predeployment	(2)		<u> </u>	x	х	x	<del> </del>	
36.4	Post deployment	(2)	<u> </u>	X	X P ORSITER	х	х		V
	* (1) ORBITEP I/F NOT REQUIRED		(	3) FUNCT:	CONC SIMULATED				
	OPBITER I/F KNOTHERDE REQUIRED		(	DIRECT	r offiter I/F				

Table C.5 (Cent) MJO/IUS INTERFACE CHECKOUT MATRIX

· · · · · · · · · · · · · · · · · · ·		1	Inti	ERPACIES		CHPKCOU	t/Test Ent		
NO. BIOCK	PESCRIPTION	* 1/F	Baseline Location	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITF ORIENTED	G975 SIM.	IVE	SPECIAL PACILITY	RIMAFECI
36.5	Orbit operations	(5)	Mer	х	х	Х.	<u> </u>		Either site: trade study
37.0	Data review	(1)		Y	x				Alternate site
38.0	Perform mechanisms deployment tests	(1)		х	x	х			
39.0	Transfer to WT/CG facility	(1)		x	x				
40.0	Prepare & calibrate WT/CG facility GSE	(1)		· x	х			x	
41.0	Install spacecraft in WT/CG facility	(1)		х	х		•	×	
42.0	Determine WT/CO	(1)		x	х			x	
43.0	Transfer to EMC/RFI Acility	(1)		x	x		,		
44.0	Prepare EMC/RFT facility GSE	(1)		x	x				•
45.0	Install spacecraft in EMS/HFI facility	(1)		х	х				
46.0	Perform EMC/RFI tests	(1)		x	x				
46.1	46.1 through 46.20 same as 12.1 through 12.20	(1)		x	x	1			
47.0	Transfer to spin facility	(1)		x	х				
48.0	Prepare & calibrate spin facility GSE	(1)		X	x			х	
49.0	Install spacecraft in spin facility, (apacecraft to be apin stabilized in orbit)	(1)		x	х			x	
50.0	Perform spin tests	(1)		x	, х			х	
51.0	Transfer to clean facility	(1)		х	х				
52.0	Prenare clean facility	(2)		x	х			хх	
53.0	Clean spacecraft & prepare for shipment	(1)	4	x	x		,		,
	* (1) ORRITEP T/F NOT REQUIRED		C	FUNCTIO	URBITER No CIMULATED				
	OPBITER I/F KNOVLEDGE REQUIERD		C	DIRECT	OPPITER I/F				

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Table C.5 (Cont) MIO/IUS

INTERPACE CHECKOUT MATRIX

			INT	ERPACES	I	CHYKCOU	/Test		
BI,OCK NO.	DESCRIPTION	* 1/P RNOWLETGE	BASELINE LOCATION	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITF ORIENTED	GSA SIM.	IAE	SPECIAL FACILITY	REMATECS
53. <u>1</u>	Dlean	(1)	User	x	х		·		
53.2	Package	(1)		х	х				
54.0	Ship to Launch site	(1)	*	х	х			•	
55.0	R/I spacecraft at launch site	(1)	Launch site		х				
56.0	R/I aft skirt	(7)			Х				·
57.0	R/I IUS flight structure	(1)			х			ļ	
58.0	White flight structure to aft skirt	(I)	· ·		х	- <u>-</u>			Option: IUS could be shipped to user site ready to mate with S/C & transfered as a unit to LS
58.1	Assemble	(1)			х			<u> </u>	S/C & transfered as a unit to 18
58.2	Clean mating surfaces	(1)			х			<u> </u>	<u> </u>
59.0	Transfer to preassembly cell	(1)		<u>,</u>	· x				
60.0	R/T solid rocket motor (SRM)	(ı)			х				
61.0	Assemble SRM to aft skirt	(1)			х				14 - 14 - 14 - 14 - 14 - 14 - 14 - 14 -
62.0	R/I igniter & explosive train assembly (FTA)	(1)			х .			<u> </u>	·
63.0	Install igniter & ETA on SRM	(1)			х				
64.0	Receive & inspect ASE	(1)			x				. '
65.0	Install ASE to IUS	(1)	·	` .	х			ļ	
66.0	GSE/ ESE _ cables	u,		,	х				
67.0	Prepare for power on	(1)	<u> </u>		х			<u> </u>	
68.0	Power on  * (1) ORRITEP I/F NOT REQUIRED	(1)	* 7	3) DIRFOR	X ORBITER				
	* (1) ORRITED I/F NOT REQUIRED  OPSITED I/F KNOWLEDGE REQUIRED		(	FUNCTI	ON" CIMULATED OPPITER 1/F				

Table C.5 (Cont) MJO/108

Table C.5 (Cont). HTO/IUS

INTERFACE CHECKOUT MATRIX

			Tet	REACES	ľ	CHEKCOU	t/test Ent		·
BLOCK NO.	description	MHOWLEDGE * I/P	BASELINE LOCATION	OPTION 1 USER ORIENTED	OPTION 2 LAUNCH SITE ORTENTED	G95 Sim.	IVE	SPECIAL FACILITY	Rippateus
76.0	Install cargo in checkout assembly	(3)	Launch site		х	х	ж		Could be IVE or assembly structure
77.0	Mechanical interface and envelope check	(3)			х		х		
77.1	Attach points	(3)			X		<u> </u>	<u> </u>	
T7.2	Bay door clearance	(3)			х		x	ļ	
<i>T7.</i> 3	Latches clearance	(3)	<u> </u>		<del>  x                                   </del>	<del></del>	x	<u> </u>	
77.4	Door setuators	(3)			X		x		
77.5	RMS support	(3)			x		x	ļ	· · · · · · · · · · · · · · · · · · ·
77.6	Electrical connectors	(3)		<u> </u>	×		×	<u> </u>	
77-7	Fluid connectors	(3)			×	·	х		
78.0	Fluid interface check	(3)			х	х	x	<b> </b>	
78.1	Leakage check	(3)	<u> </u>	<u> </u>	_ x _	X	X	ļ	
78.2	Pressure test	(3), ı		<u> </u>	x	X	x	ļ	IVE cooling system will
79.0	RTG coolant loop check	(3)* `	<u> </u>	, ,	X		x	<u> </u>	accommodate RNG
79.1	Jeakage check	(3)	<u> </u>		<u> </u>		ж .	<u> </u>	
79.2	Pressure test	(3)					x	<u> </u>	,
79.3	Flow check	(3)		<u> </u>	x		х	<del>                                     </del>	· · •
80.0	Electrical interface check	(3)	<del>                                     </del>	ļ	x	·	ļ	<u> </u>	
80,1	Bus isolation	(3)	<u>  ·                                     </u>	ļ				<del> </del>	
80.2	# (1) ORRITEP T/F NOT REQUIRED	(3)	1 * .	3) DIRPOR	X ORSITER		<u> </u>	<u> </u>	<u> </u>
	OPBITER I/F KNOWLENGE REQUIPED				ON" CIMULATED				

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Table C.5 (CHO4) MJO/IUS INTERPACE CHECKOUT MATRIX

DESCRIPTION  Exmine power distribution  age variation	(3) (3) * 1/P	MASELIN LOCATI	ON ORIENTED	OPTION 2 LAUNCH SITF ORIENTED	GSP SIM.	IVE	SPECIAL PACILITY	rimarks	
age variation		Launch si	te '	9 8			<b></b>	RIMATECS	
naients '	(3)	1		<u> </u>	хх	x		Either one; trade study	
· · · · · · · · · · · · · · · · · · ·				х	х	. х			
	(3)			x	Х	·x			
) Dle	(3)			х	Х	х			
fy electrical system functional interface	(3)			x	х	х			
system	(3)			х	X	x			
system	(3)			х	x	x			
and & control system	(3)	<u> </u>		x	x	x	<u> </u>		
brate experiments & FSE	(3)			×	хх	x		Unique GSB	
through 83.10 same as 1.1 through 1.10	(3)			×	х	х	<u> </u>	4	
& verify flight software	(3)			x	x	х		Either one: trade atuly	
al generation	(3)			. x	х	х	<u> </u>		
time monitoring	(3)			х	х	х			
	(3)			x	x	x	<u> </u>		
	(3)	<u> </u>		x	x	х			
	(3)			x	х	х			
	(3)			x	Х	х	ļļ	,	
ify compatibility-with tracking stations	(3)		<u> </u>	х	X	<u>x'</u>	<b></b>	Unique equipment/facility	
	(1)	†	72X 111 (915 40)	X	<del></del>	<u> </u>			
ORBITEP I/F NOT REQUIRED									
	system  a system  braid & control system  braid & control system  braid & control system  braid & control system  braid & control system  braid & control system  braid & control system  braid & control system  braid & system as 1.1 through 1.10  d & verify flight software  constant system/experiment verific.  through 85.16 same as 1.1 through1.10  and 3.1 through 3.6  rational tests in orbiter environment  form EMI/RFI tests simultaneously with  operational tests  ify compatibility with tracking stations  construct to WT/BAL facility  1 ORBITEP I/F NOT REQUIRED	system (3)  a system (3)  a system (3)  brand & control system (3)  brate experiments & FSE (3)  character experiment & FSE (3)  character experiments & FSE (3)  character experiments & FSE (3)  character experiments & FSE (3)  character experiments & FSE (3)  character experiments & FSE (3)  character experiments & FSE (3)  character experiments & FSE (3)  character experiments & FSE (3)  character experiments & FSE (3)  character experiments & FSE (4)  character experiments & FSE (4)  character experiments & FSE (4)  character experiments & FSE (4)  character experiments & FSE (4)  character experiments & FSE (4)  cha	system (3)  a system (3)  a system (3)  a system (3)  a system (3)  a system (3)  a system (3)  a system (3)  a system (3)  a through 83.10 same as 1.1 through 1.10 (3)  a through 83.10 same as 1.1 through 1.10 (3)  a time monitoring (3)  a system (3)  a through 85.16 same as 1.1 through 1.10  ad 3.1 through 3.6 (3)  a system (3)  a sys	system (3)  a system (3)  a system (3)  and & control system (3)  brate experiments & FSE (3)  through 83.10 same as 1.1 through 1.10 (3)  a & verify flight software (3)  at weight software (3)  through 85.16 same as 1.1 through 1.10  at 3.1 through 85.16 same as 1.1 through 1.10  ad 3.1 through 3.6  cational tests in orbiter environment (3)  corn EMI/RT tests simultaneously with operational tests (3)  ify competibility with tracking stations (3)  anser to WT/BAL facility (1)  DIRECTION	system (3) X  system (3) X  sand & control system (3) X  brate experiments & FSE (3) X  through 83.10 same as 1.1 through 1.10 (3) X  the verify flight software (3) X  sal generation (3) X  time monitoring (3) X  trime monitoring (3) X  system (4) X  through 85.16 same as 1.1 through 1.10 (3) X  cartional tests in orbiter environment (3) X  rational tests in orbiter environment (3) X  corner EMI/RFT tests simultaneously with operational tests in orbiter environment (3) X  corner to WT/BAL facility (1) X  anafer to WT/BAL facility (1) DIRFOR DESITER FUNCTION' FIMULATED	System   (3)	System   (3)	System   (3)	

				TARFACES		CHPKCOU EQUIPM	r/Test Ent		
BLOCK NO.	PESCRIPTION	# I/F	BASELIN LOCATI		OPTION 2 LAUNCH SITF ORIENTED	GSB SIM.	IVE	SPECIAL PACILITY	REMARKS
89.1	Prepare for transfer	(1)	Launch s	te	x				
89.2	Transfer	(1)			x		<u> </u>		
89.3	Receiving inspection	(1)			x				
90.0	Prepare & calibrate WT/BAL facility GSE	(1)			х		<u> </u>		
91-0	Install cargo to WT/BAL GSE	(1)			X			<b> </b>	
92.0	Perform WT/BAL test	(5)			х				
93.0	Transfer cargo to pad	(1)			х		ļ <u>.</u>	<b></b>	
94.0	Install cargo in FCR	(2)			x		ļ	х	Specilized handling equipment
95.0	Extend cargo changeout room	(4)			¥	,		X	
95.1	Inflate FCR/orbiter seal	(4)			х		<u> </u>	X	
95.2	Furge interstitial area	(4)			×		<u> </u>	×	
95.3	Open FCR door	(1)			х			X	
96.0	Install RTG to cargo	(4)	<u> </u>		<u> </u>		<u> </u>	<u> </u>	Specialized handling equipment
97.0	Extend cargo into payload bay	<u>(4)</u>	<u> </u>		х		<del> </del>		
98.0	Mate cargo to orhiter	(4)	1		, x	<b></b>	<u> </u>	<u> </u>	
98.1	Mechanical	(4)			х	<u> </u>	<u> </u>	<del> </del>	
98.2	Electrical connectors	. (4)	1		x		<u> </u>	<b></b>	
98.3	Fluid connectors	(4)	1.  -		x	· '	<del> </del>	<del>                                     </del>	,
99.0	Cargo/orbiter interface verification	(4)	•	(3) DIRFC	Y ORSITER -	<u> </u>	1	<u> </u>	
	* (1) ORBITER I/F NOT REQUIRED  ORBITER I/F KNOTLEDGF REQUIRED			O FURT	ION- FIMULATE T OPPITER I/F	1			

Table C.5 (Cont) MJO/IUS INTERFACE CHECKOUT MATRIX

			Triti	iryacres		CHPKCOU EQUIPM	/Test	·	
BLOCK NO.	description	* 1/F	maseline Location	OPTION 1 USIN ORIENTED	OPTION 2 LAUNCH SITF ORIENTED	Gen Gen	IVE	SPRCIAL PACILITY	BPMARKS
99.1	Plugs out electrical continuity	(4)	Launch site	,	х				
99,2	Sonnect & verifyumbilicals and fluid lines	(4)			х				
99•3	Install and verify mechanical latches and retention devices	(4)			x	and the state of t			
100.0	Cargo power on	(4)		WW	х				
101.0	Cargo Launch Readiness Verification (LRV)	(h)			Х				
102.0	Disconnect payload ground handling mechanisms (RGHM) from cargo	(4)			x				
102.1	Secure RHM in RCR	(h)			хх				
103.0	Closeout bay doors	(4)			х		,		+40.444.
103.1	Retract Rendezvous radar (R/R) antennas	(4)			х				,
103,2	Retract RMS	(4)			x	····			
104.0	Cabin closeout	(4)			х				
105.0	Clear md	(4)			×				
106.0	Cargo services (as required)	(4)			. х				
107-0	Retract cargo changeout room	<u>'(4)</u>			· x				
108.0	Inunch operations	(h)			X		,		
		<u> </u>							5 '
						······································	*****		
								-	
	* (1) ORBITEP I/F NOT REQUIRED		(	DIRFOR	URSITER ON CIMULATED			<u> </u>	
	(2) OFFITTER I/F KNOWLEDGE REDUIERD		7		OPPITER I/F				